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GRAPHANT: A FORTRAN PROGRAM FOR THE SOLUTION
AND GRAPHIC DISPLAY OF GAIN AND PATTERNS FOR WIRE
AND LINEAR ANTENNAS IN THE PRESENCE OF LOSSY
GROUND

BY

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ABSTRACT:

An interactive computer graphics antenna gain pattern computation and display program for real-world antenna systems is presented. The use of the program as a teaching tool at the Naval Postgraduate School is discussed. Methods for applying the program for the synthesis and design of complex antenna systems are indicated. Research applications include techniques for rapid inspection of gain equations of newly developed antennas. A ship motion model is developed for studying the effects of electrical geometry variations caused by ship motion in heavy seas on maritime antenna systems and a dynamic presentation of pattern variations is made.

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INITIAL DISTRIBUTION LIST

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A. INTRODUCTION

Anyone who has attempted to correlate the actual performance of HF and VHF wire antennas operating in a real-world environment to the "highly sanitary," theoretical radiation patterns and gain which proliferate in text books and handbooks will immediately recognize the need for a simple method of predicting antenna performance in the presence of the earth. This report describes the development and use of a computer tool, which enables anyone with a working knowledge of Fortran and a free-space radiation pattern and mutual impedance formulation to analyze and design antenna systems of arbitrary orientation above a specified lossy plane earth.

It serves as both a teaching aid and a design tool. In instructional use, it provides quick interaction via a graphics display of antenna parameters and radiation patterns plots. In seconds, a student can observe the performance of several popular HF/VHF antennas in any plane earth configuration he chooses. Equivalent digital computer/plotter turn-around time is in excess of 1 hour and manual calculation time on the order of days. The program firmly convinces the student that the antenna system is composed of the antenna plus its environment.

For the communications system designer, rapid evaluation of antenna systems enables him to choose an antenna type and orientation which will enhance the performance of the total system rather than arbitrarily guessing which antenna package might be suitable. When new antenna types are developed, their radiation pattern and impedance equations can readily be added to the basic calculation package and the full potential of the antenna may be painlessly determined, not just for the usual, mystical free space environment, but for the surroundings in which the radiator will be used. Using an HF Ionospheric Propagation Prediction program which can return optimum radiation angles for a specified path and time, a systems designer can synthesize an optimum antenna pattern for the particular situation under investigation. The optimum may be manually entered and each design iteration compared with the optimum. The pattern Save and Recall options are used to arrive at the best type antenna and orientation available to him.

Previous work on antenna patterns in the presence of ground is widely scattered in the literature. Specific antenna types are referenced in the appendices. The initial incentive for this investigation was to increase the usefulness of an antenna radiation pattern subroutine for HF antennas, currently in use as part of a HF Ionospheric Propagation Prediction program written by ESSA¹. The gain and input resistance equations from this report with some modifications and corrections were used. Equations programmed are included in Appendix C.

The program as presently configured assumes a current distribution on the antenna. The consequences of this are small errors in terminal impedance with a corresponding discrepancy in gain value. Radiation patterns are affected very little by the differences between assumed and actual currents. To calculate exact current distributions would be prohibitive in both time and programming effort, considering the limited worth of the more exact gain figures which would result.

1 Ref: ITS 78 Report.

B. BRIEF DESCRIPTION OF PROGRAM

The program consists of two basic parts:

1. The solution of antenna pattern equations for radiators of arbitrary orientation above a flat earth of specified ground constants (conductivity and permitivity). Gain values are calculated from terminal impedance expressions containing self (free space) and mutual (coupling between the antenna and its image) effects.

2. The graphics display portion which displays program input parameters specifying antenna type, size, orientation, ground constants and special features such as recall and storage of patterns. This part of the program also generates power intensity plots vs. azimuth and elevation angles and displays these radiation patterns on the graphics screen. Gain values can be displayed in conjunction with the patterns.

Subroutines for special functions which are usually found in wire antenna patterns and impedance formulae and numerical quadrature calculations are included for the convenience of persons wishing to apply their own specific antenna to the program. The user who wishes to do this must provide pattern equations for his antenna for arbitrary orientation. The effect of ground reflections and the selection of observation angles is provided by the program itself.

Special features available to the user are:

1. Plotting patterns on a log scale vs. linear.
2. Storage and recall of patterns for comparison purposes.
3. Ability to generate a desired pattern shape (via light pen) which is stored and recalled for comparison.
4. Simple ship-ocean model for dynamic simulation of shipboard antenna systems.

When the program is used at USNPGS, the XDS 9300 digital computer core limitations restrict the calculation and viewing of one pair of cuts in the 3 dimension geometry (i.e. one azimuth rotation at one specified elevation angle and one zenith to horizon elevation cut.) Typical time for a pattern calculation is 15 seconds, with a simple dipole requiring 9 seconds and a vertical monopole with ground screen up to 2 minutes. Seven common antennas are currently programmed:

1. Arbitrary Tilted Dipole
2. Vertical Monopole
3. Vertical Monopole with Ground Screen
4. Inverted L
5. Sloping Long Wire

6. Rhombic

7. Vertical Half-Rhombic

More complex programs for arrays such as Yagis, Log Periodic Dipoles and Monopoles, and curtains will require fairly long calculation times due to the extensive mutual impedance calculations.

C. DYNAMIC ANALYSIS OF SHIPBOARD ANTENNAS

The ease of obtaining the effect of the earth on antenna performance prompted the investigation of the programs potential to display the effect of typical ship motion of shipboard HF antenna radiation. The equations for an arbitrary tilted dipole, sloping long wire and vertical monopole were already in the form to allow variable tilt angle. By programming a ship-ocean model that reorientates the antenna with ship motion (roll and pitch), it is possible to show slow but dynamic pattern changes with sea surface as a function of sea state and ship direction for a chosen type of vessel. This simplified model rocks the antenna in two planes as the ship responds in roll and pitch to ocean waves but still assumes a plane ocean reflecting surface. For medium and heavy seas, the results indicate an appreciable re-lobing effect and show that the variation in signal at a particular observation angle may be as high as 20 db.

This is an additional factor which should be considered when assigning locations for antennas in new ship designs. Present efforts at evaluation of these antenna locations do not include sea state perturbations.

The next stage of investigation of ship motion effects will include a variable geometry for the sea surface to replace the plane shape. At low HF the effect might be approximated by a partially random "fuzzy" surface while for UHF the distances are large in term of radio wavelengths and the model could be more nearly that of a rolling surface contour.

The final results of these extensions will be of benefit to the antenna locator, as previously explained, as well as to communications managers. Depending upon the sensitivity of the total communications link to antenna lobe structure, communications procedures for a given frequency may be improved by insight obtained from the investigation of sea state effects.

D. RECOMMENDATIONS

The radiation equations for most of the antennas do not include the arbitrary geometry factors needed for ship motion effects study and should be expanded to include them. Gain calculations depend upon input resistance which in the case of the sloping long wire and others do not include mutual effects. Where possible and where warranted, these effects should be included by deriving coupling terms for input impedance equations. (This will not alter the shape of the radiation patterns and affects only the magnitude of the fields and hence the gain),

Array antenna equations exist in the literature and should be carefully verified and adapted for inclusion in the plotting program.

When this program is used for matching patterns produced by HF propagation prediction programs, a convenient data interfacing technique (such as tape) should be developed for use between the graphics system and the larger general purpose digital machine used in the prediction calculations.

APPENDIX A

DETAILED PROGRAM DESCRIPTION

Appendix A

Section III contains a description of program operation. The program is divided into processors, program subsections that perform the major computational tasks. Processor operation and interaction are described.

1. PROGRAM OPERATION

The program displays a data and option command input format at the graphics terminal (see figure A-1). The program operator enters applicable parameters values for antenna geometry, environment, and option commands using text editing techniques. A blank graphics block is then displayed at the CRT. Trial patterns may be manually entered in this block using graphics editing techniques. Manually entered patterns will be displayed with all subsequently computed patterns allowing the operator to compare computed patterns with trial patterns on the CRT. Exercising the reinitialization option will erase the trial pattern.

The program computes horizontal and vertical gain patterns and displays them at the graphics terminal. The horizontal pattern is computed with zenith constant at the input value for θ' and azimuth varied from 1 to 360 degrees by one degree increments. The vertical pattern is computed for azimuth constant at the input value of ϕ' and zenith varied from 1 to 90 degrees by 1 degree increments. Linear and log displays are available. If a log display is not ordered with the log pattern option command, linear patterns will be displayed.

Patterns are saved by exercising the save pattern option. Pattern vector data is stored in the digital machine in a save array when save is ordered. Exercising the recall option will cause patterns saved in the save array to be displayed.

Use of save and recall options allows simultaneous display of saved and current patterns for comparison purposes.

A dynamic simulation of a shipboard antenna mounted on a rolling pitching ship in a stop-action type of presentation is programmed. Entering sea state and direction in the data format causes the simulation to operate. Sea motion is resolved into ship motion and ship motion into antenna parameter variation. Patterns are computed and displayed at 10 degree intervals of wave period. Sea state 0 must be entered to by-pass the dynamic simulation if it is not desired.

Appendix F is operating instructions for use of the program implemented at the Computer Graphics Laboratory, U. S. Naval Postgraduate School, Monterey, California. Figure A-16 is a schematic of the graphics computer system at this facility.

2. PROCESSOR DESCRIPTION

A processor flow chart is presented in figure A-2. Processor operation and interaction is described below.

A. The Parameter Format Processor initializes the display graphics and text data blocks and displays the text format for parameter and program options commands input.

B. The Parameter and Options Input Processor is used to enter problem parameters and program option commands using the format provided by the previous processor. The parameter and options input processor requires entry of all parameters each time the program is initialized. All other utilizations of this processor require changing only individual parameters as desired. If the reinitialization option is selected by the operator, the parameter format processor is branched to from the parameter and options input processor.

C. The Pattern Manual Entry Processor displays a blank graphics data block. By manually editing this data block, the operator may draw a pattern that will be displayed with all subsequently computed patterns. Erasing this manually entered pattern must be done by reinitializing in the parameter and options command processor. If no manual pattern is desired, this processor may be terminated without entry being made.

D. The Environmental Constants Processor computes the values of problem constants that are functions of antenna parameters and environmental conditions and not dependent upon observation angles.

E. The Input Resistance Processor computes a value for input resistance of the antenna entered in the parameter and options input processor. If the equations in the gain processor assume a nominal value of input resistance, a value of 1.0 is assigned to input resistance.

F. The Observation Angle Constants Processor computes values of problem constants that are functions of observation angles for those observation angles for which the antenna gain is to be computed.

G. The Gain Processor computes the gain of the antenna selected in the parameter and option command input processor at the selected zenith angle all integer values of azimuth angle from 1-360 degrees, and the selected azimuth angle and all values of zenith angle from 1-90 degrees. These two gain vectors are the horizontal and vertical gain patterns.

H. The Normalize and Max Gain Processor selects the maximum value of gain from both horizontal and vertical linear patterns and normalizes both patterns with respect to this maximum value. This operation is required to scale patterns for graphics display. The absolute value of maximum gain is computed and its \log_{10} taken. This value is displayed in the text data format.

I. The Log Gain Processor operates if the operator manually selects the log gain option in the parameter and options command input processor. The horizontal and vertical linear patterns are converted to logarithmic patterns with a 30 db range of $(10\log_{10}\text{max gain})$ to $(10\log_{10}\text{max gain})$. These patterns are renormalized by the log gain processor.

J. The Pattern Display Processor is a two part processor which displays the horizontal and vertical patterns at the graphics terminal.

K. The Pattern Save Processor is a two part processor which operates if the horizontal save and vertical save option are selected by the operator in the parameter and option command input processor. They may be independently selected. This processor transfers the pattern currently displayed by the display processor to storage in the digital machine in a save array. Entering a pattern in a save array destroys the pattern previously saved so care must be exercised to bypass this processor if saving the pattern for several compute cycles is desired.

L. The Display Saved Patterns Processor operates when the recall option has been selected by the operator. The processor recalls the patterns saved in the save array and displays them at the graphics terminal. Operation of this pattern does not destroy data in the save array.

M. Dynamic Processor. This processor computes and displays a simulation of shipboard whip, dipole and sloping longwire antenna patterns. Entry of an integer larger than 0 in ISEA will cause this processor to operate. The processor computes sinusoidal ocean waves with magnitude dependent upon sea state. Ship roll and pitch which are functions of ship type, sea state and relative direction of the seas are computed. Parameter variations caused by ship motion are computed and the normal compute loop entered with the modified values of antenna parameters. The patterns are computed and displayed and the gain at the θ' and ϕ' inputted in the parameter input processor is displayed under SIGL in the text data format. The ocean model is re-entered. The ship roll and pitch cycles are divided into 36 discrete steps and a pattern computed and displayed for each step. The display will be a stop-action type display of antenna pattern vs. time. Entry of 0000 under ISEA will cause this processor to be bypassed.

3. PROCESSOR FUNCTIONAL DESCRIPTION

Figures A-3 thru A-15 are functional flow diagrams of processors. Equations for the ocean model, gain, and input resistance used in the gain and input resistance processors are included in Appendix E. The sources for gain and input resistance equations are ESSA Technical Report ESSA-ERL-110-ITS 78, A.F. Barghausen, J. W. Finney, L. L. Proctor, L. D. Schultz, May 1969 and ESSA Technical Report ESSA-ERL-104-ITS 74, M.T. Ma, L.C. Walters, April 1969. A listing of the Fortran Program used to implement the program is Appendix D.

4. EXTENSION OF THE PROGRAM

The program may be extended to compute patterns for other types of antennas. Adding antennas may be accomplished by inserting an input resistance branch in the input resistance processor and a gain branch in the gain processor. If additional parameters are required, the parameter format must be changed to accept them. Multi-element antennas such as Yagi or Log Periodic will have a mutual impedance matrix; the terms of this matrix may be evaluated using the mutual impedance equations in the dipole

branch. Specific changes required to add antennas to the USNPGS implementation are as follows:

1. Statements 134 and 135 may be changed to new parameter names.
2. After statement 162, add DECODE statements for new parameters.
3. New constants statements, if any, should be inserted between statements 198 and 218.
4. After statement 229 in the input resistance processor, insert
IF(ANTN.EQ.9) GO TO 1900.
5. After statement 771 in the input resistance processor, add 1900
INPUT RESISTANCE BRANCH STATEMENTS
.
.
.
GO TO 2000
6. In the gain processor after statement 272, insert IF(ANTN.EQ.9)
GO TO 900.
7. In the gain processor after statement 771, insert 900 ANTENNA
GAIN STATEMENTS
.
.
.
GO TO 42

The dynamic simulation is available for whip, sloping longwire and vertical whip antennas. Simulation of other antennas aboard ship may be made by rewriting gain equations to allow arbitrary orientation of the antenna. Orientation variations are available in the ocean model and the dynamic simulation can then be made.

FIGURE A-1

ANTN

LENG

HGHT

PHIP

THEP

FREQ

EPSL

SGMA

PHI

THET

PARM

ISTH

ISTV

IRCL

HGTT

ALPH

GAIN

ISEA

ICRS

SIGL

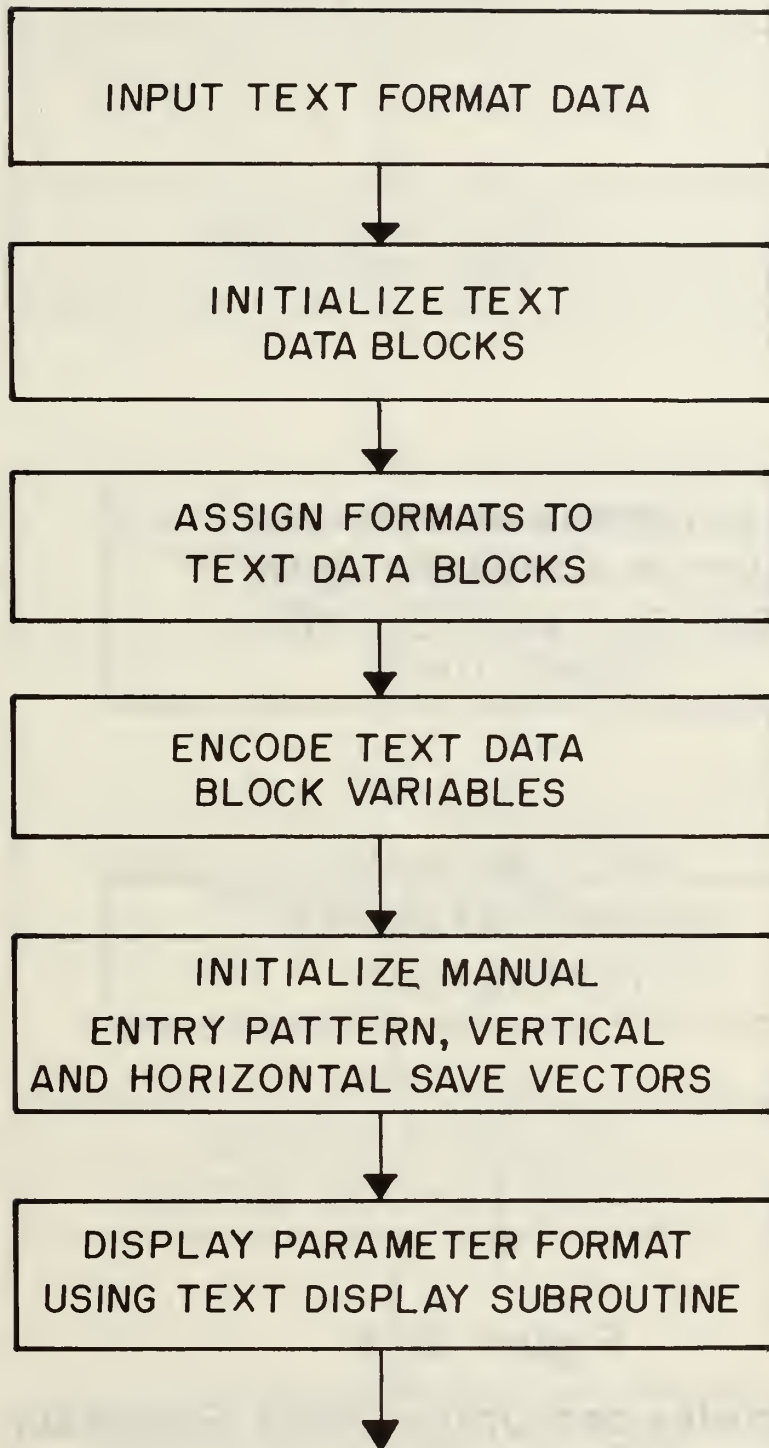


Figure A.3
Parameta Format Processor

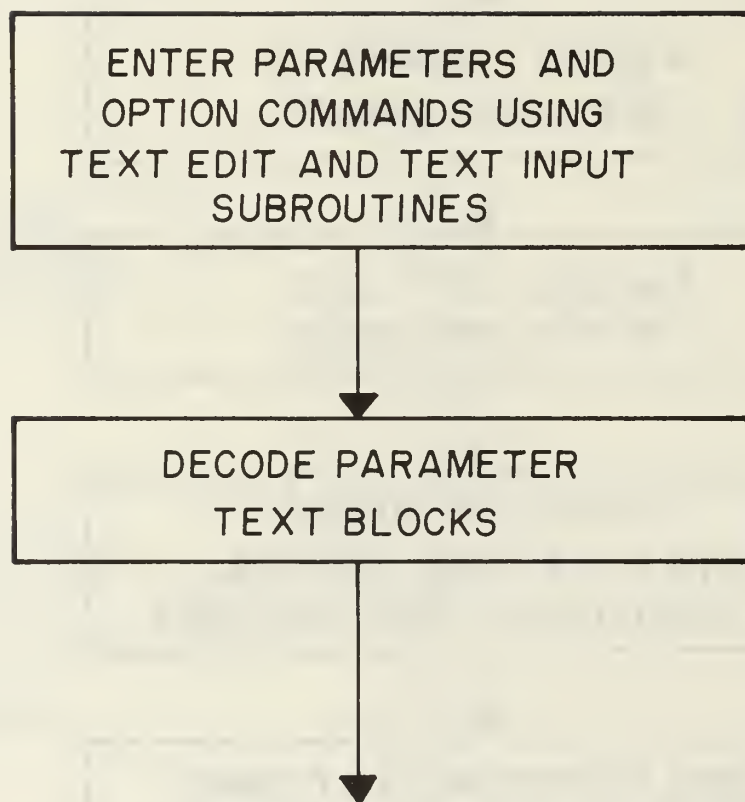


Figure A. 4
Parameter and Option Input Processor

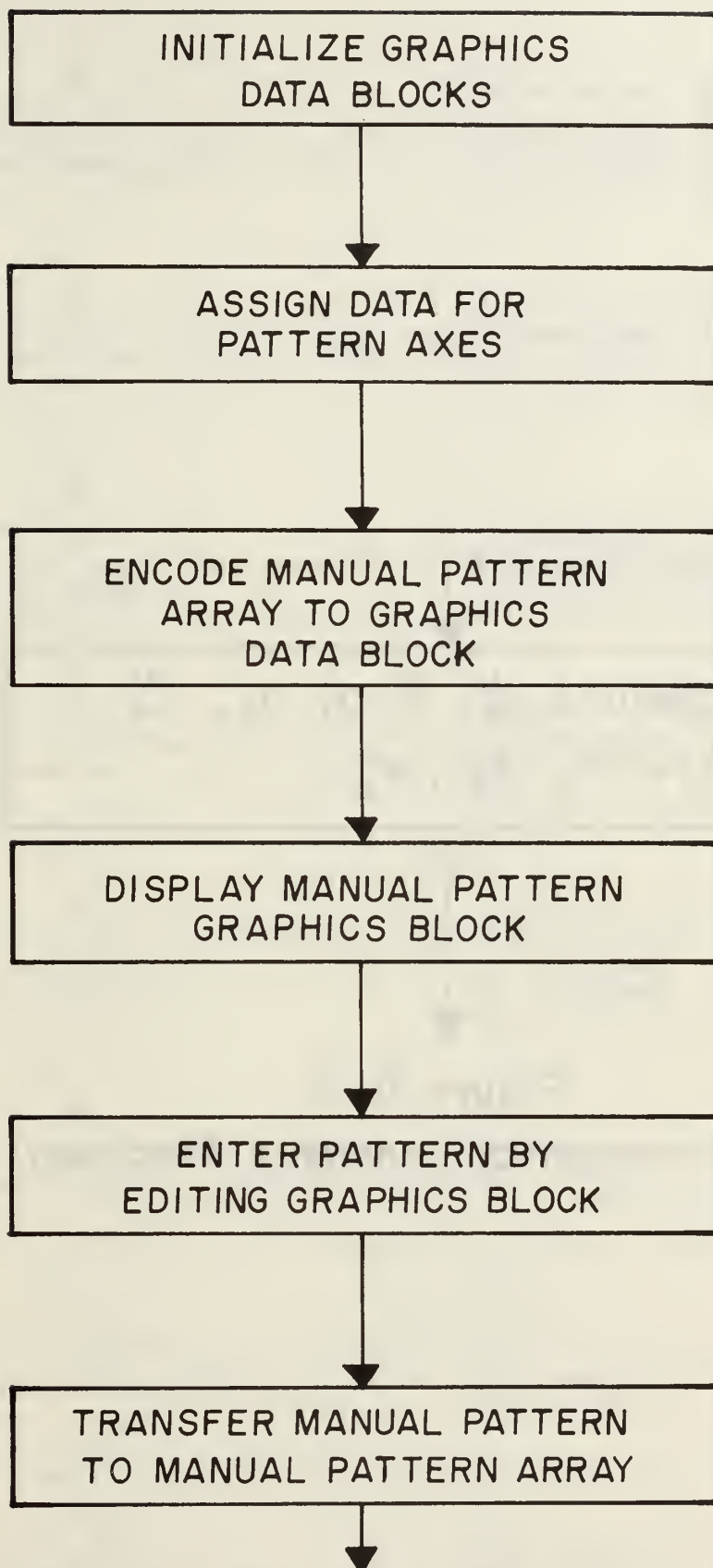


Figure A. 5
Pattern Manual Entry Processor

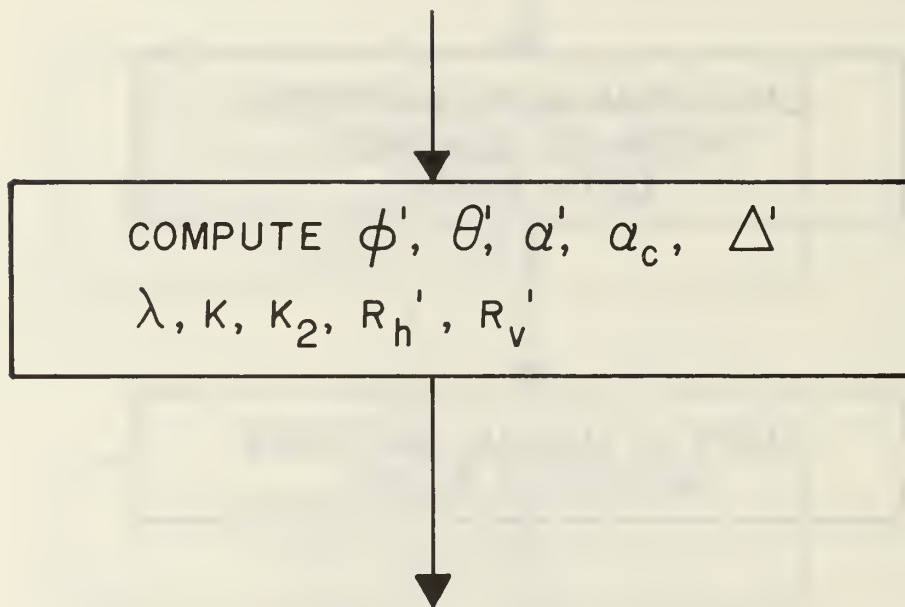


Figure A. 6
Environmental Constants Processor

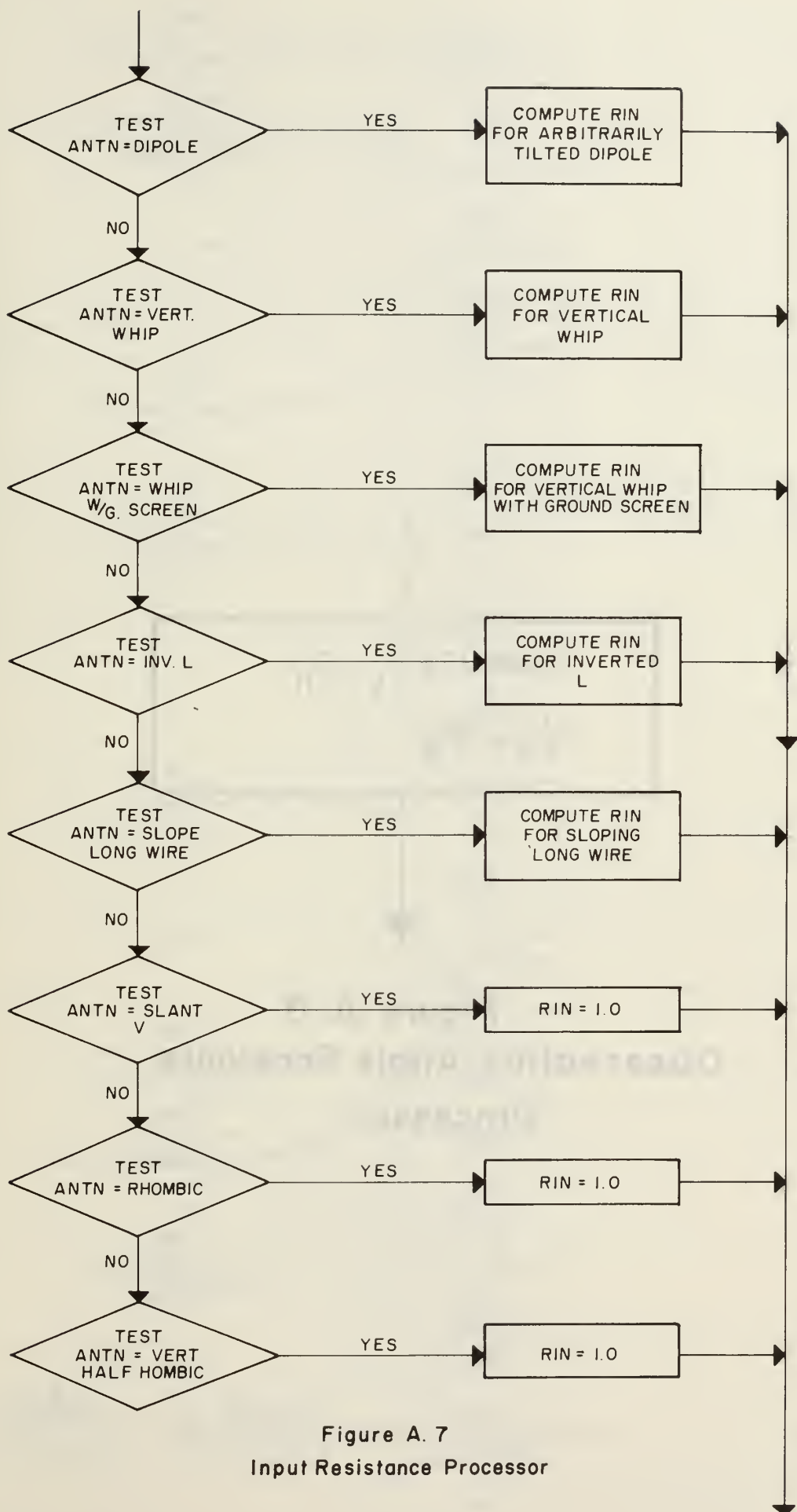


Figure A. 7
Input Resistance Processor

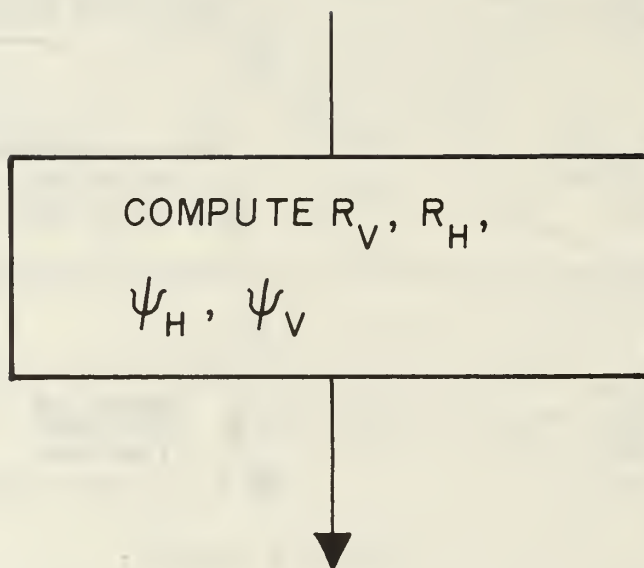


Figure A. 8
Observation Angle Constants
Processor

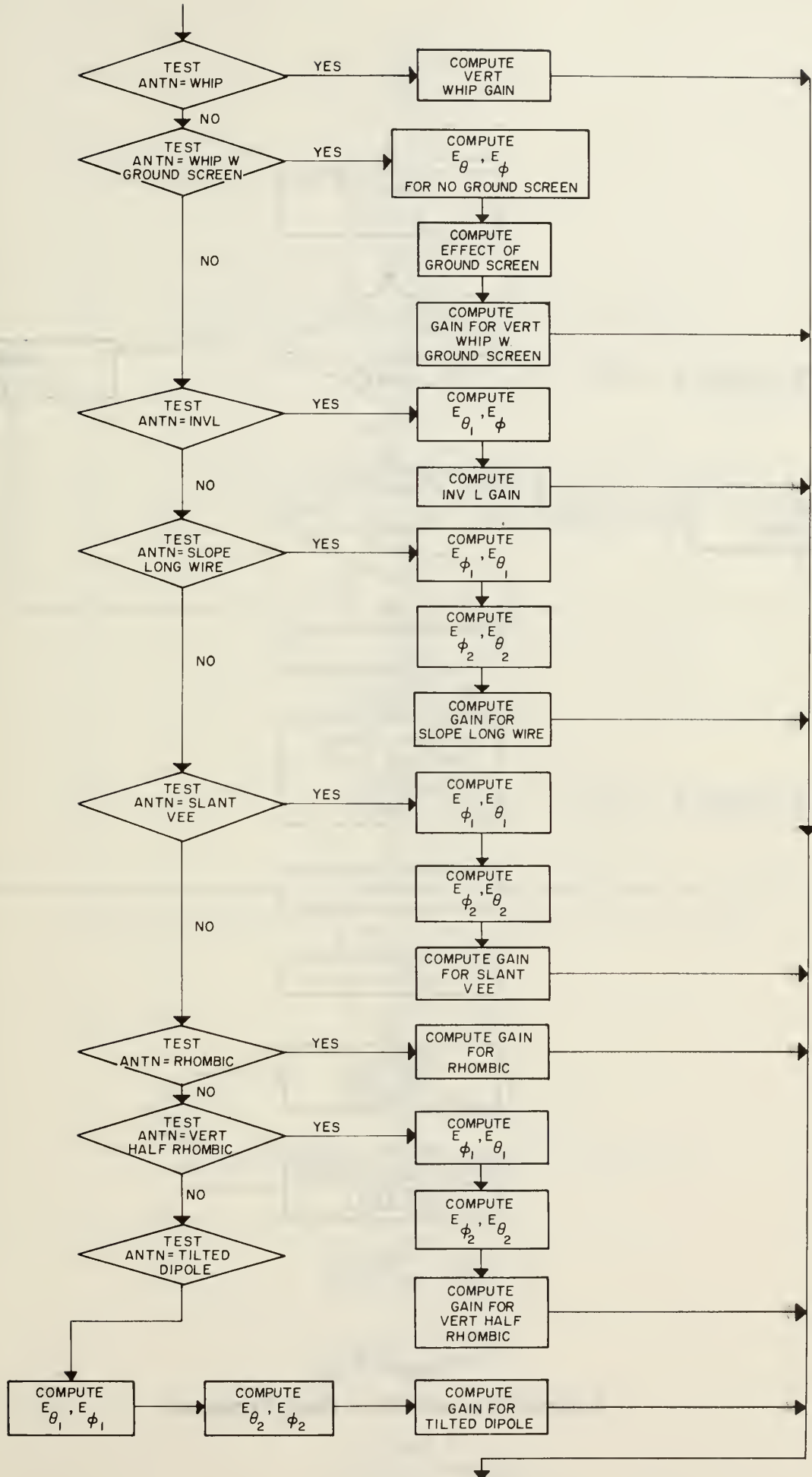


Figure A.9
Gain Processor

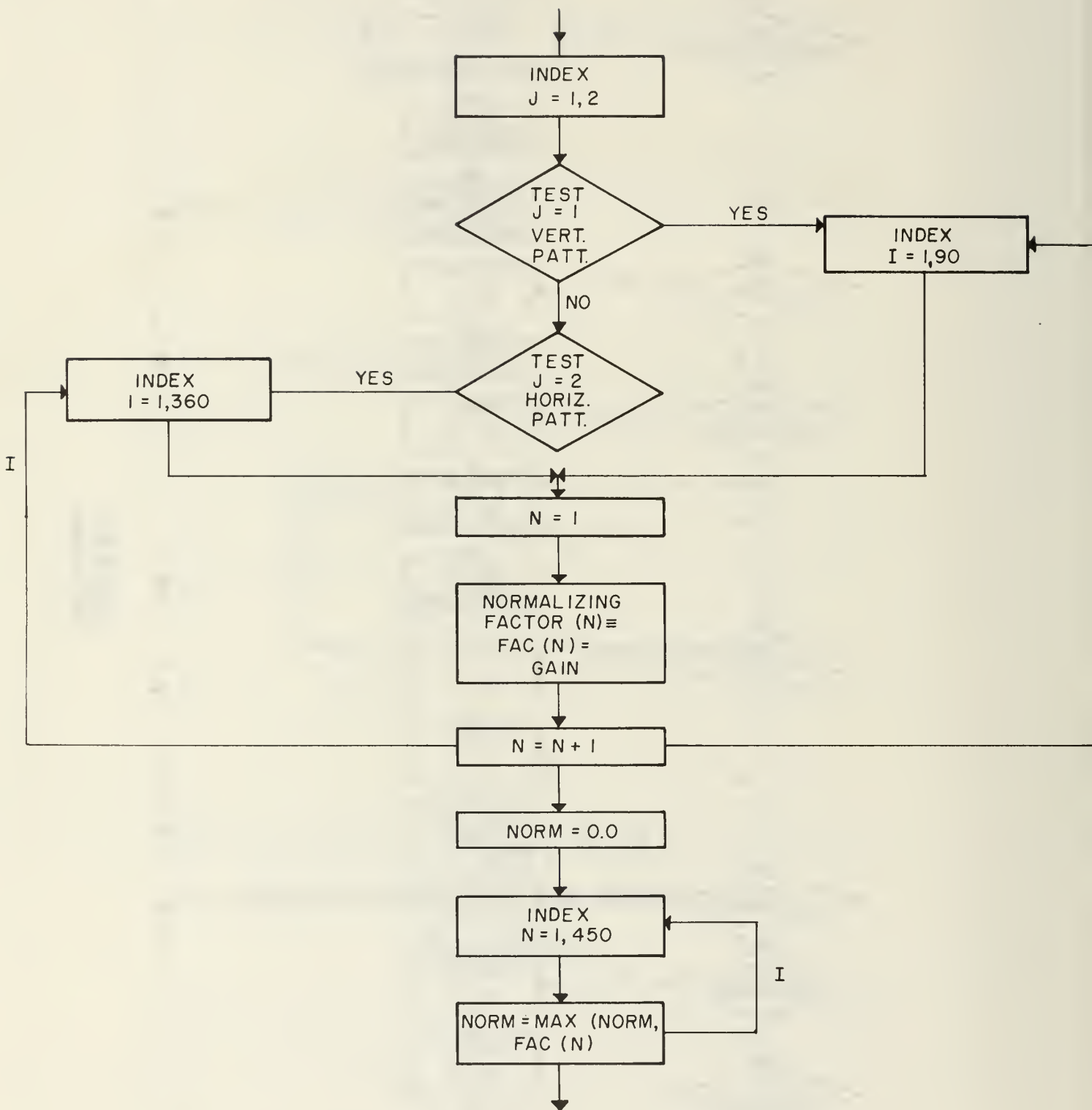


Figure A. 10
Normalize and Max Gain Processor

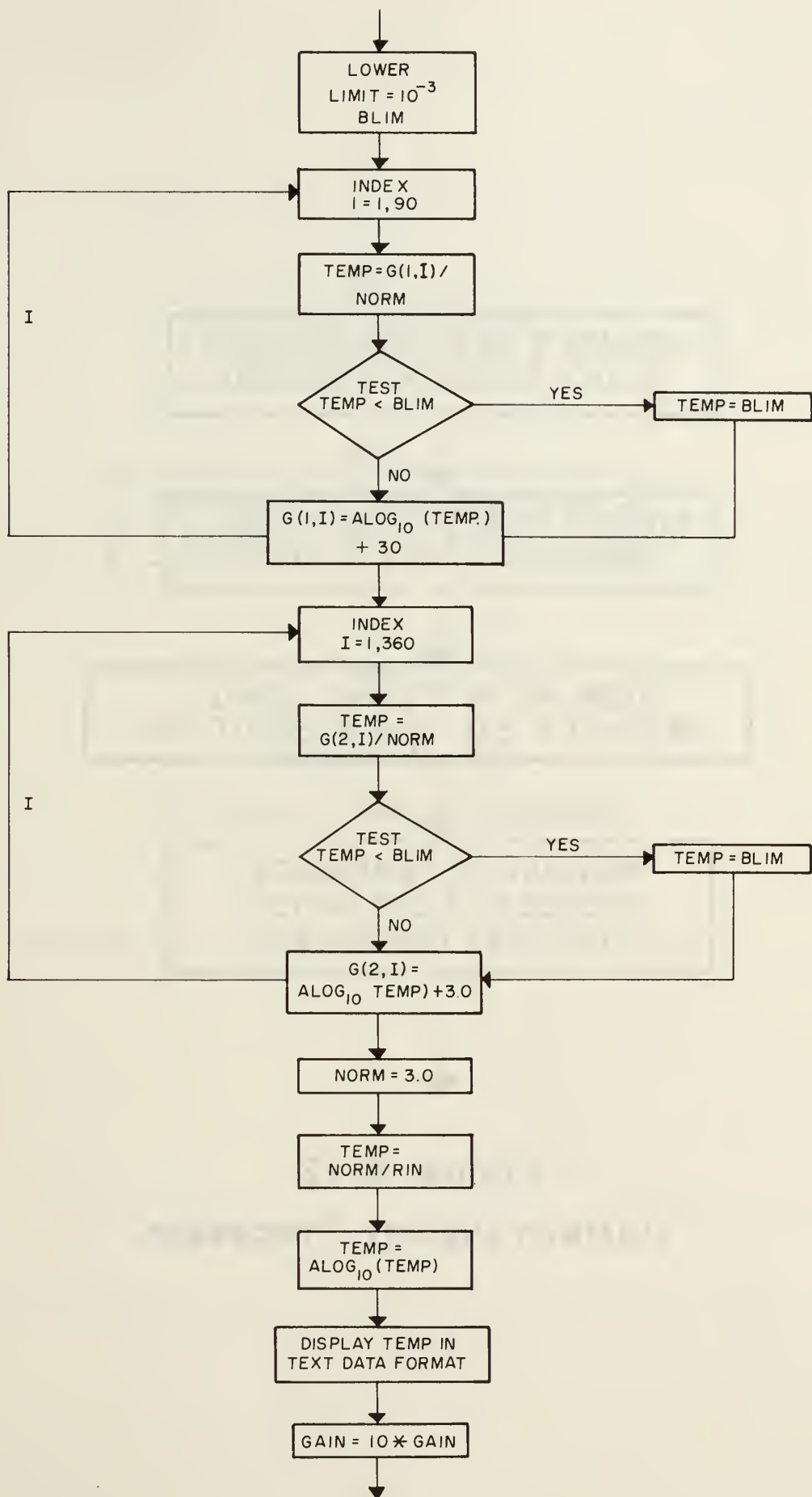


Figure A. II
Low Gain Processor

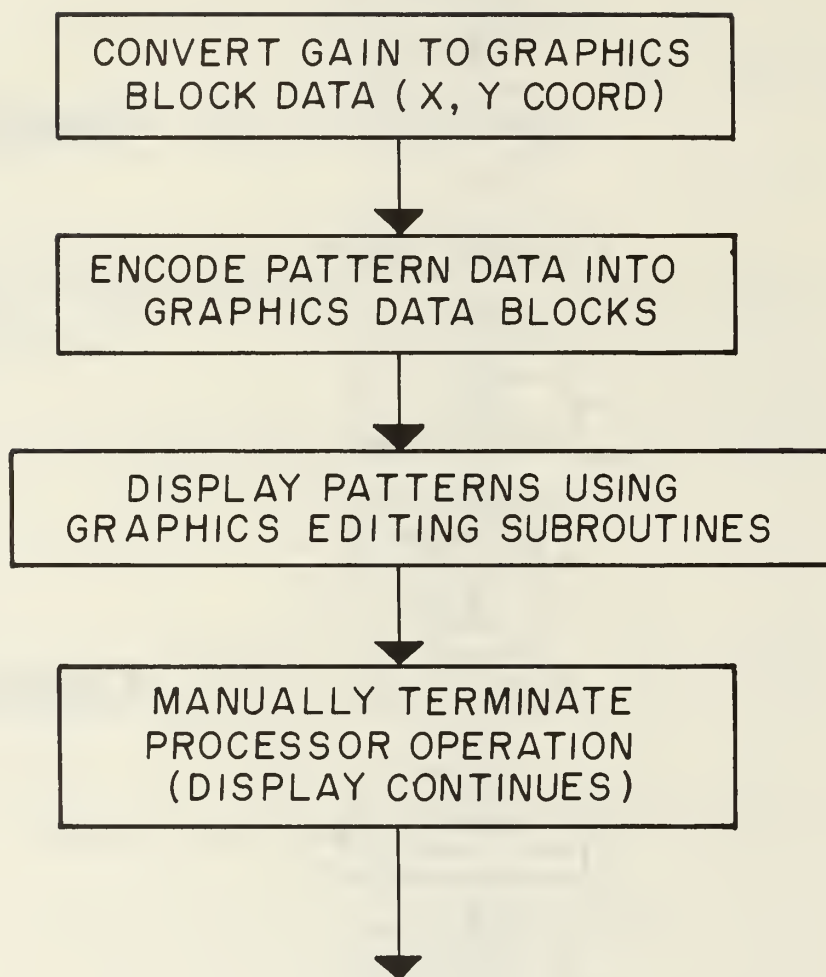


Figure A. 12
Pattern Display Processor

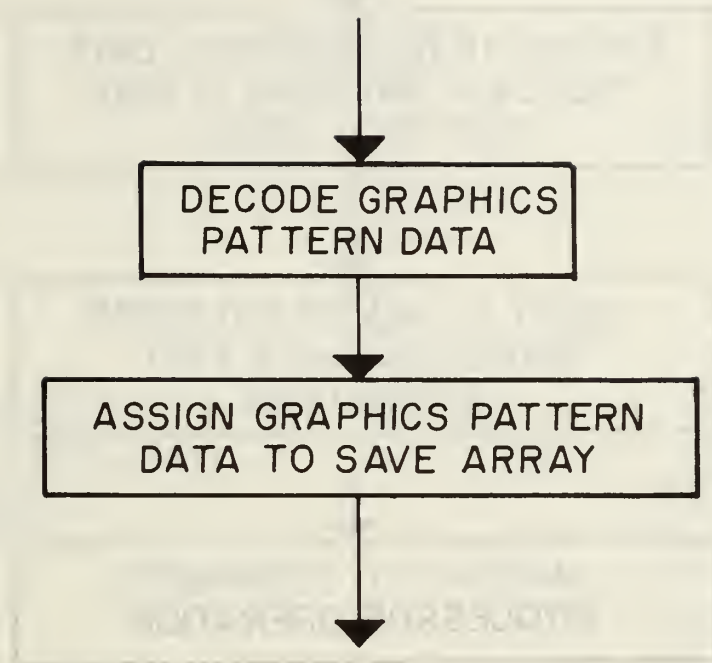


Figure A. 13
Pattern Save Processor

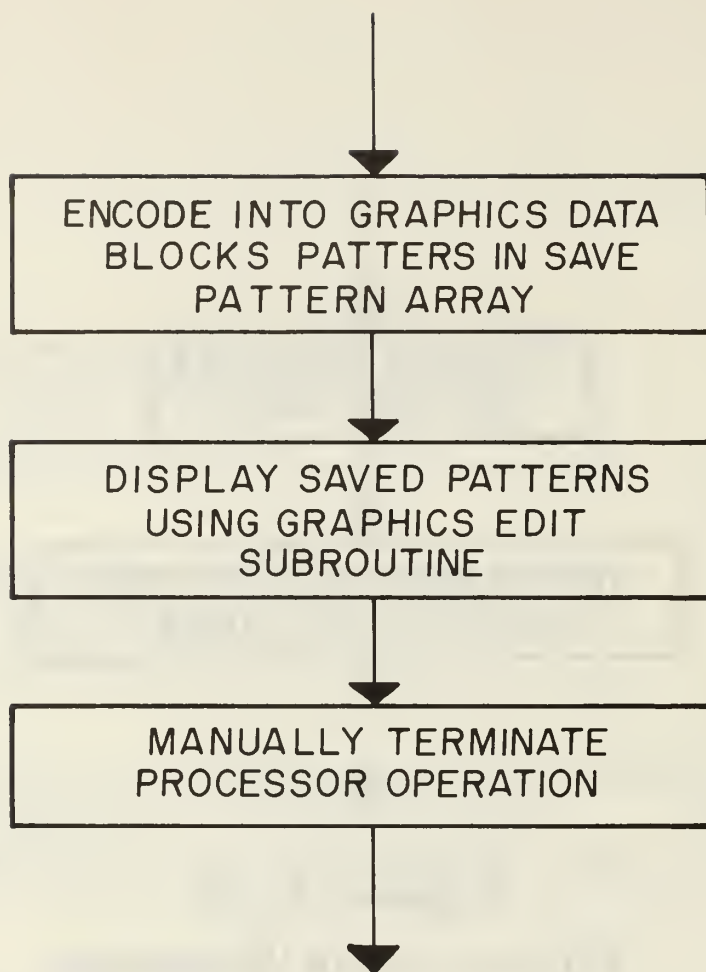


Figure A. 14
Saved Pattern Display Processor

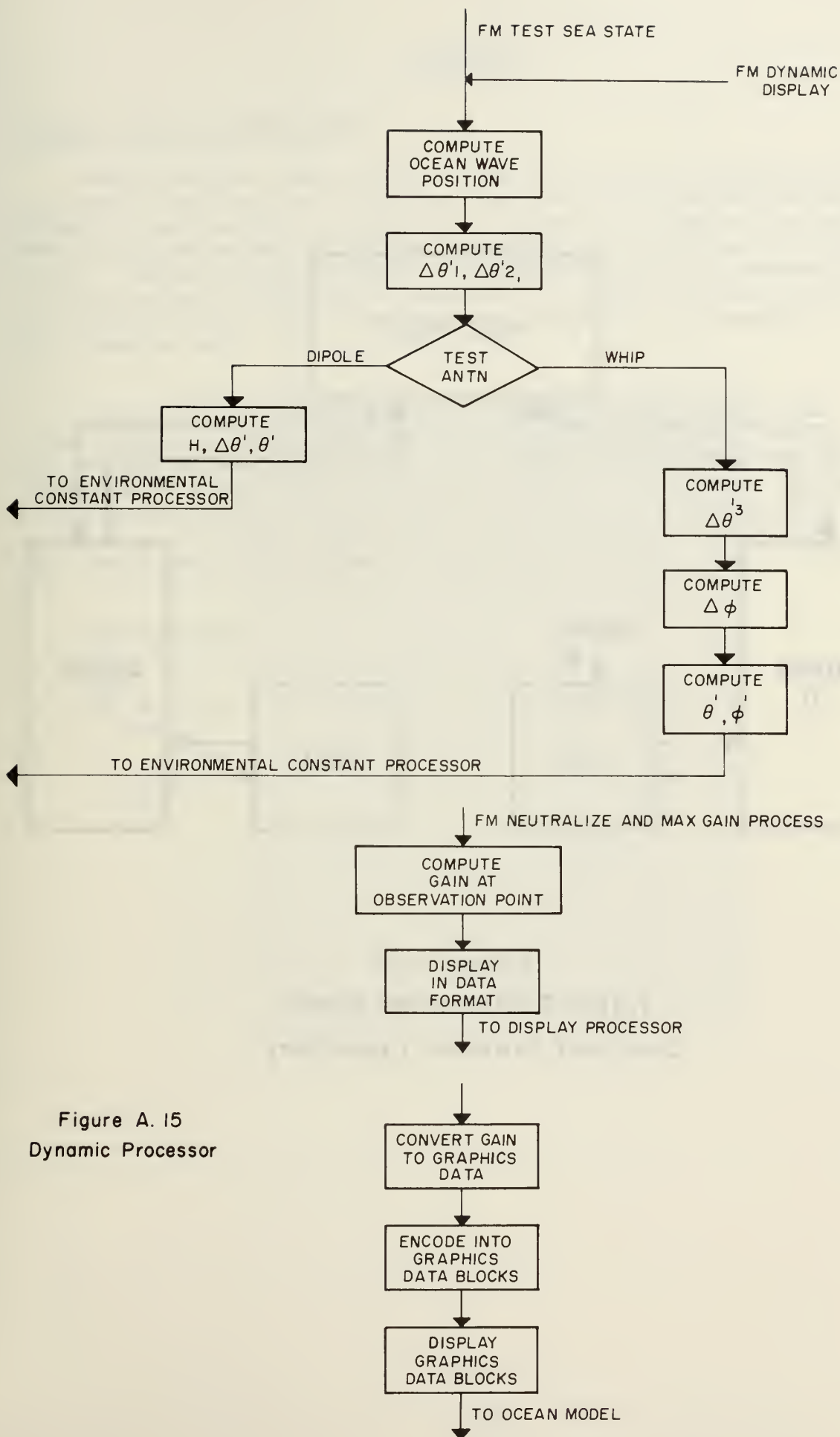


Figure A. 15
Dynamic Processor

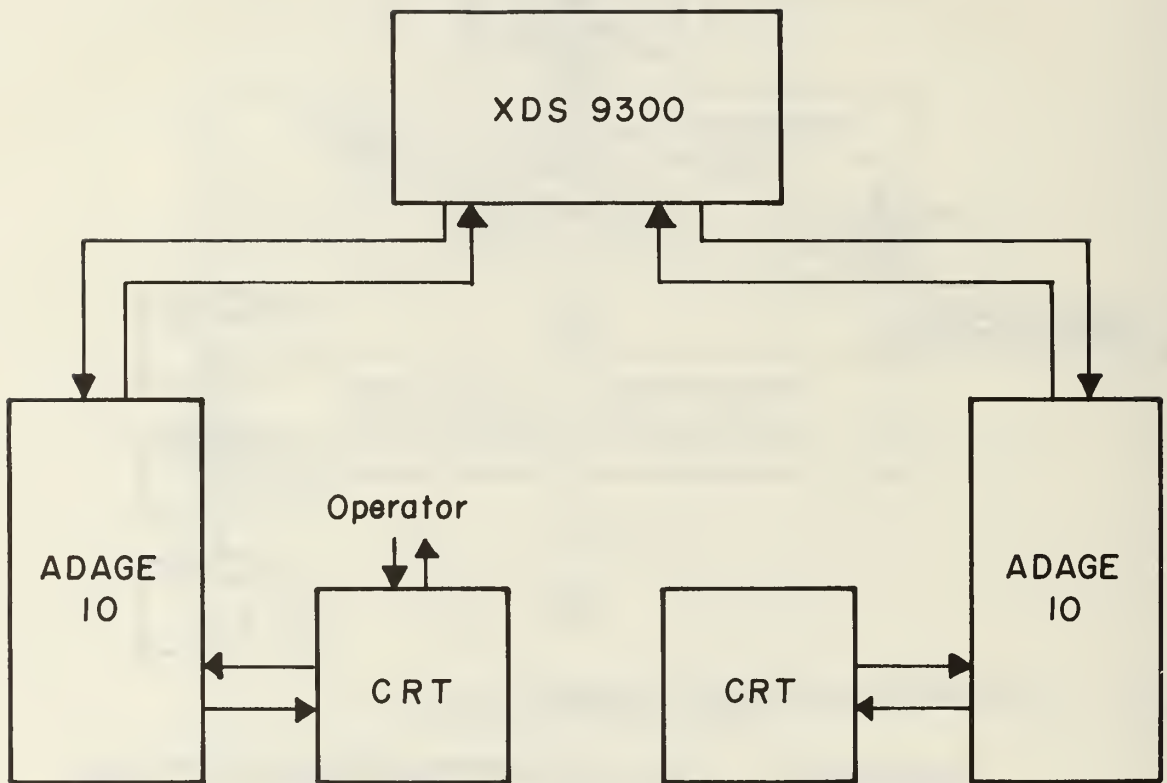


Figure A16
Naval Postgraduate School
Computer Graphics Laboratory

APPENDIX B

EXAMPLE PATTERN COMPUTATIONS

Example pattern calculations for the seven antennas programmed are presented in this section. Patterns were computed for typical parameter values for each antenna. Computation of effects of parameter and environment variations as well as the use of program control options are demonstrated. The text input required to compute each pattern is presented with a CRT photograph of the pattern computed. The USNPGS user may use the text input in conjunction with the user instructions of Appendix F to learn program use.

Figure B-21 is a film strip of the 36 images that comprize the dynamic simulation of a shipboard vertical whip antenna in a state 5 sea from 045 degrees relative to ship's bow. Figure B-22 is a dynamic simulation of a horizontal dipole in the same sea conditions. The images of the dynamic simulation are computed at 10 degree intervals of the ship's roll and pitch period. Figures B-21 and B-22 should be scanned down columns and from bottom of left column to top of right columns.

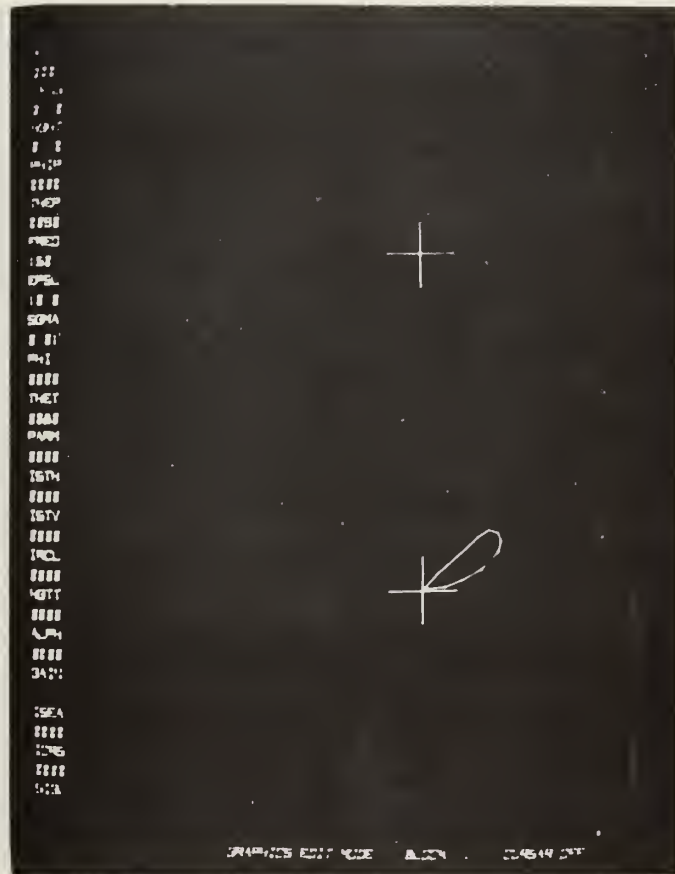
VERTICAL WHIP

2.5m
 $f = 30.0 \text{ mhz}$
 $\epsilon_r = 80$
 $\delta = 5.00$
 $\theta = 045$
 $\phi = 060$

DIPOLE

1.0m
 150.0 mhz
 $\epsilon_r = 80$
 $\delta = 5.0$
 $\theta = 75^\circ$
 $\phi = 0$
 $h = 6\text{m}$

ANTEN
 0001
 LENG
 01.0
 HGHT
 01.0
 PHIP
 0000
 THEP
 0090
 FREQ
 150.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 LSTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: Manually entered $\lambda/2$

Vertical dipole pattern; height $\lambda/2$

ANTEN
 0001
 LENG
 01.0
 HGHT
 01.0
 PHIP
 0000
 THEP
 0090
 FREQ
 150.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: $\lambda/2$ dipole; $\lambda/2$ height; good ground; $\phi' = 0$, $\theta' = 90$,
 $f = 150$ mhz, $\ell = 1.0$ $h = 1.0$; overlay manually entered pattern;
 Observation angles $\phi = 0$, $\Theta = 80$

FIGURE B.3

ANTN
0001
LENG
01.0
HGHT
01.0
PHIP
0000
THEP
0090
FREQ
150.
EPSL
10.0
SGMA
0.01
PHI
0000
THET
0080
PARM
0001
ISTH
0000
ISTV
0000
IRCL
0000
HGTT
0000
ALPH
0000
GAIN
ISEA
0000
ICRS
0000
SIGL

Comments: Erase manual trial pattern
no pattern computed

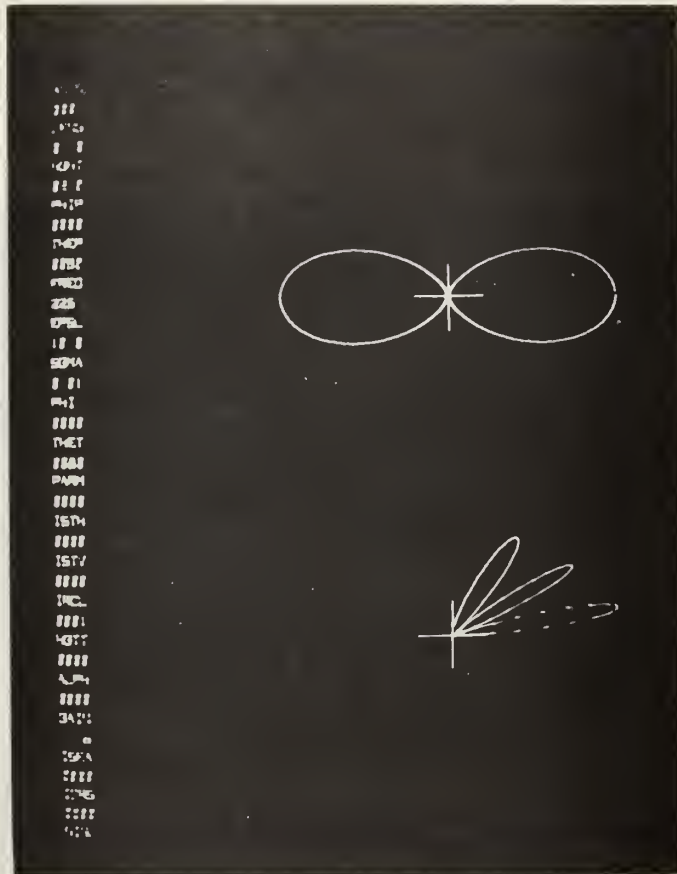
ANTN
0001
LENG
01.0
HGHT
02.0
PHIP
0000
THEP
0090
FREQ
150.
EPSL
10.0
SGMA
0.01
PHI
0000
THET
0080
PARM
0000
ISTH
0001
ISTV
0001
IRCL
0000
HGTT
0000
ALPH
0000
GAIN
ISEA
0000
ICRA
0000
SIGL



Comments: $\lambda/2$ dipole; λ height; good ground; $\phi'=0$, $\theta'=90$,
f= 150 mhz, $l= 1.0$, $h=2.0$; Observation angles $\phi=0$,
 $\theta=80$; save patterns

FIGURE B.5

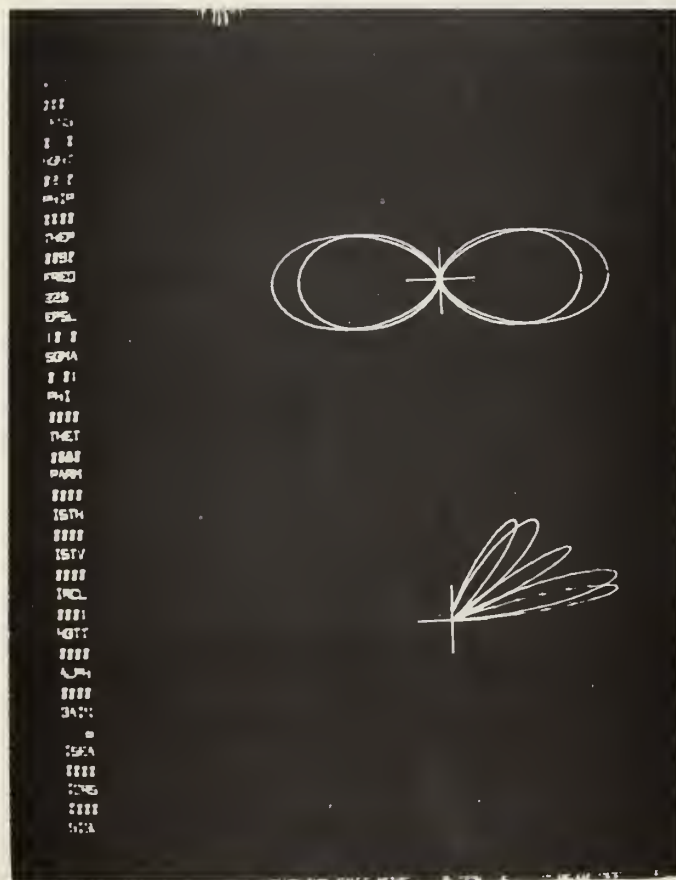
ANTN
 0001
 LENG
 01.0
 HGHT
 02.0
 PHIP
 0000
 THEP
 0090
 FREQ
 225.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0001
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: $2 \lambda/3$ dipole; $4 \lambda/3$ height, good ground, $\phi'=0$, $\theta'=90$,
 $f=225$ mhz, $l=1.0$, $h=2.0$; observation angles, $\phi=0$, $\theta=80$.
 The effects of changing frequency are shown here.

FIGURE B.6

ANTEN
 0001
 LENG
 01.0
 HGHT
 02.0
 PHIP
 0000
 THEP
 0090
 freq
 225.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0001
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



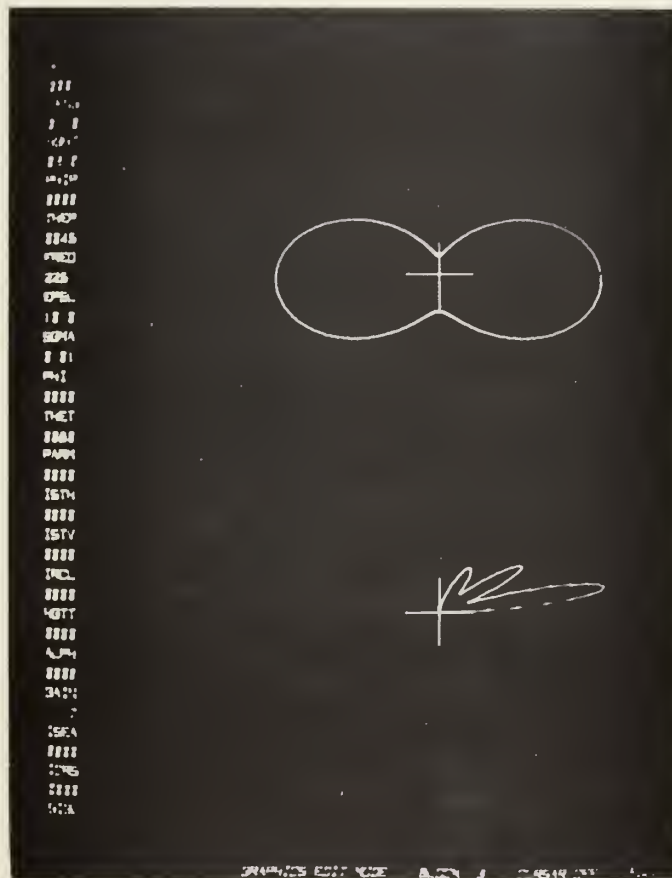
Comments: Recall pattern 4 and overlay on pattern 5. The use of save and recall options are shown in this example. The options are used to compare the λ dipole (inside and 2 lobe pattern) with $4/3\lambda$ dipole (outside and 3 lobe pattern).

ANTN
 0001
 LENG
 01.0
 HGHT
 02.0
 PHIP
 0000
 THEP
 0090
 FREQ
 225.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0002
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



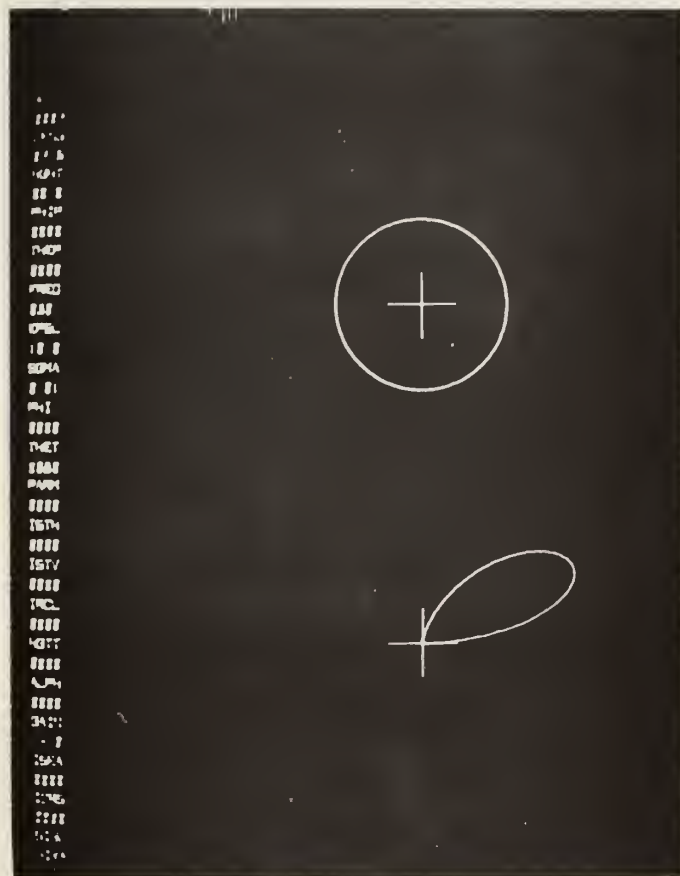
Comments: Two thirds wave length dipole; Four thirds wave length height; good ground, $\phi'=0, \Theta'=90$, $f=225$ mhz, $\ell=1.0$, $h=20$; observation angles $\phi=0$ $\Theta=80$; Log patterns 30 db scale. Log pattern option is used to study side lobe structure.

ANTEN
 0001
 LENG
 01.0
 HGHT
 02.0
 PHIP
 0000
 THEP
 0045
 FREQ
 225.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



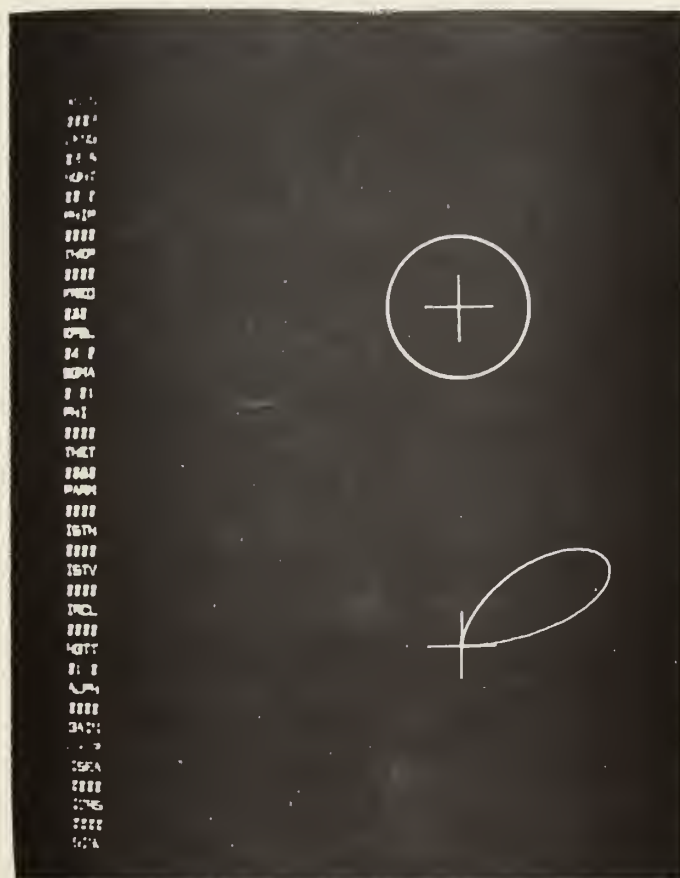
Comments: Two thirds wavelength dipole; four thirds wave length height,
 good ground, $\phi'=0$, $\Theta'=45^\circ$ (tiltangle), $f=225$ mhz, $\ell=1.0$
 $h=2.0$, observation angles $\phi=0$ $\Theta=80$; Tilted dipole. The
 effect of tilt on dipole radiation patterns is demonstrated here.

ANTN
 0002
 LENG
 02.5
 HGHT
 00.0
 PHIP
 0000
 THEP
 0000
 FREQ
 030.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



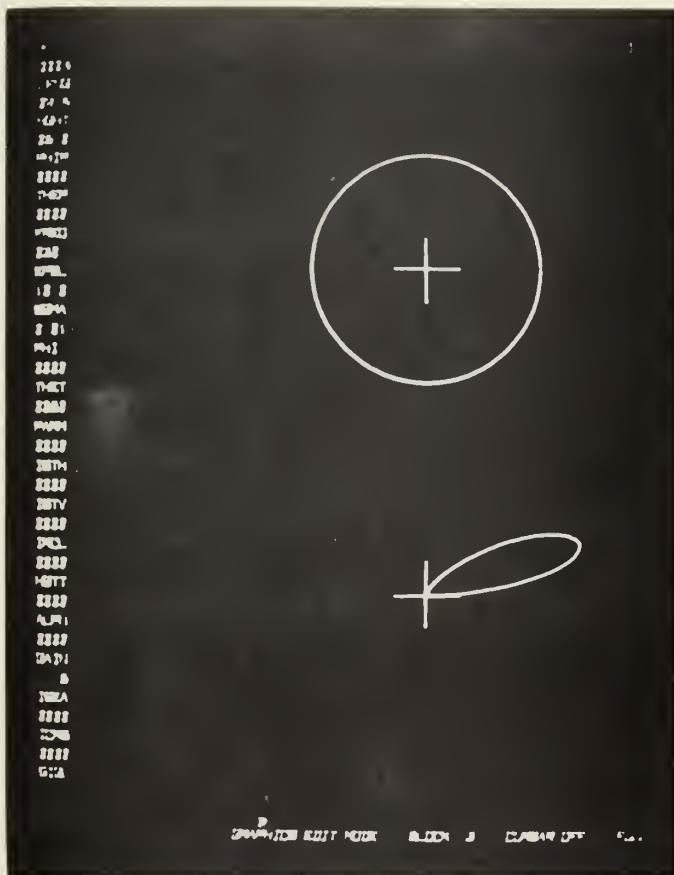
Comments: Quarter wavelength whip; good ground, $f=30$ mhz, $\lambda=25$;
 observation angles $\phi=0$, $\theta=80$

ANTEN
 0001
 LENG
 02.5
 HGHT
 00.0
 PHIP
 0000
 THEP
 0000
 FREQ
 030.
 EPSL
 04.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 01.0
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: Quarter wavelength whip; poor ground, $f=30$ mhz, $l=2.5$, observation angles $\phi=0$, $\theta=80$. The effects of changes in reflecting ground are shown in this example. The ground change from good ground to poor ground causes a decrease in gain of 2db and a slight increase in θ of max radiation.

ANTN
0003
LENG
02.5
HGHT
05.0
PHIP
0000
THEP
0000
FREQ
030.
EPSL
10.0
SGMA
0.01
PHI
0000
THET
0080
PARM
0000
ISTH
0000
ISTV
0000
IRCL
0000
HGTT
0000
ALPH
0000
GAIN
ISEA
0000
ICRS
0000
SIGL



Comments: $\lambda/4$ vertical whip with
half wave length ground radial screen.

FIGURE B.12

ANTN

0003

LENG

02.5

HGHT

10.0

PHIP

0000

THEP

0000

FREQ

030.

EPSL

10.0

SGMA

0.01

PHI

0000

THET

0080

PARM

0000

ISTH

0000

ISTV

0000

IRCL

0000

HGTT

0000

ALPH

0000

GAIN

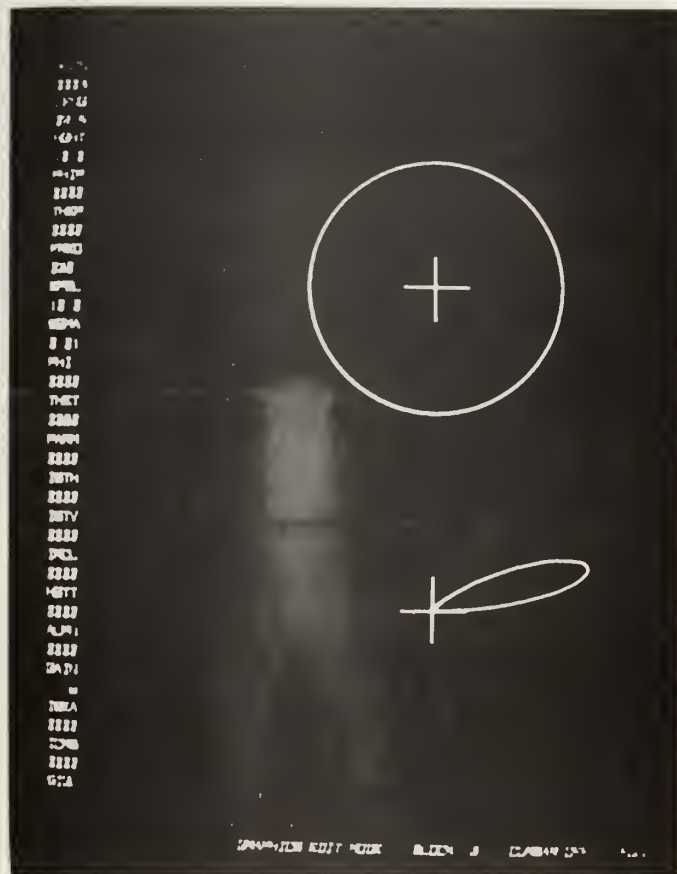
ISEA

0000

ICRS

0000

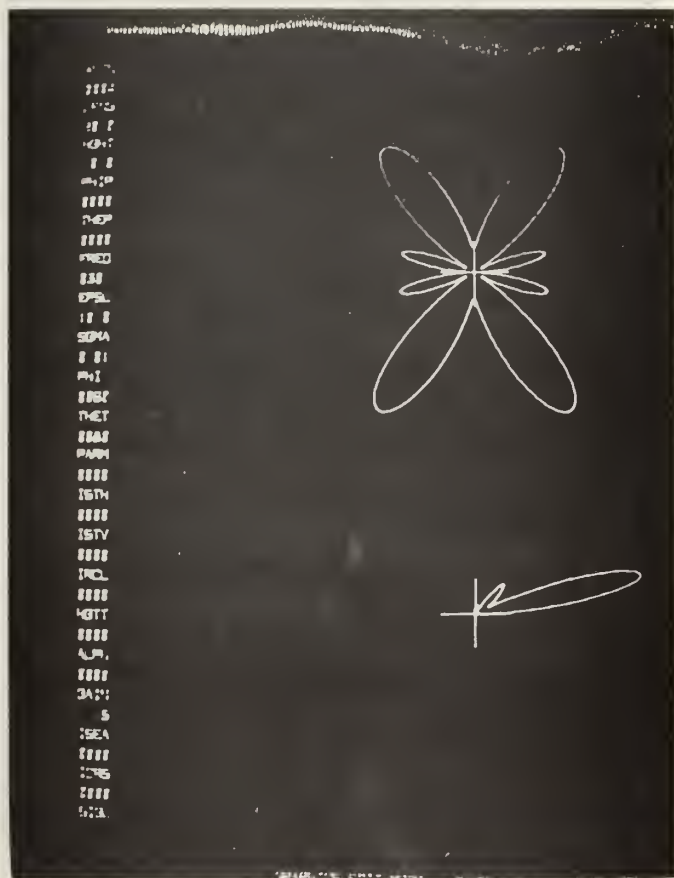
SIGL



Comments: $\lambda/4$ vertical whip with wave length ground screen. Increasing screen narrows and depresses vertical pattern.

FIGURE B.13

ANTN
0004
LENG
20.0
HGHT
10.0
PHIP
0000
THEP
0000
FREQ
030.
EPSL
10.0
SGMA
0.01
PHI
0060
THET
0080
PARM
0000
ISTH
0000
ISTV
0000
IRCL
0000
HGTT
0000
ALPH
0000
GAIN
ISEA
0000
ICRS
SIGL



Comments: Inverted L, horizontal run two wavelengths, vertical run one wavelength; good ground; $h=10.0$, $\ell=20.0$, $f=30$ mhz; observation angles $\Phi=60$, $\Theta=80$.

FIGURE B.14

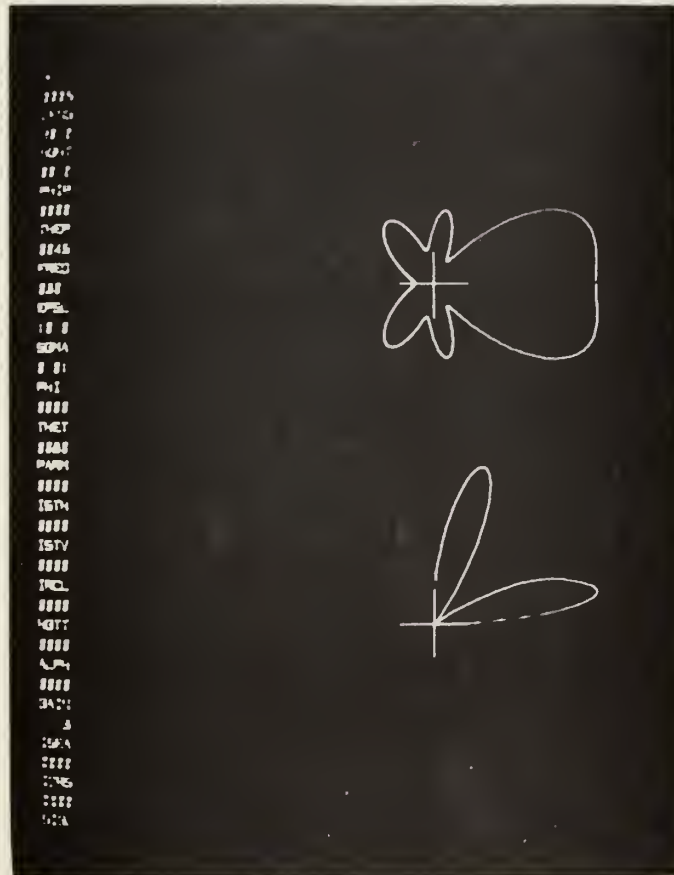
ANTN
0004
LENG
20.0
HGHT
05.0
PHIP
0000
THEP
0000
FREQ
030.
EPSL
10.0
SGMA
0.01
PHI
0090
THET
0080
PARM
0000
ISTH
0000
ISTV
0000
IRCL
0000
HGTT
0000
ALPH
0000
GAIN
ISEA
0000
ICRS
0000
SIGL



Comments: Inverted L, horizontal run two wavelengths, vertical run one-half wave length; good ground; $h=5.0$, $\ell=20.0$, $f=30$ mhz; observation angles $\phi=90$, $\theta=80$. The effects of change in vertical run length are shown here.

FIGURE B.15

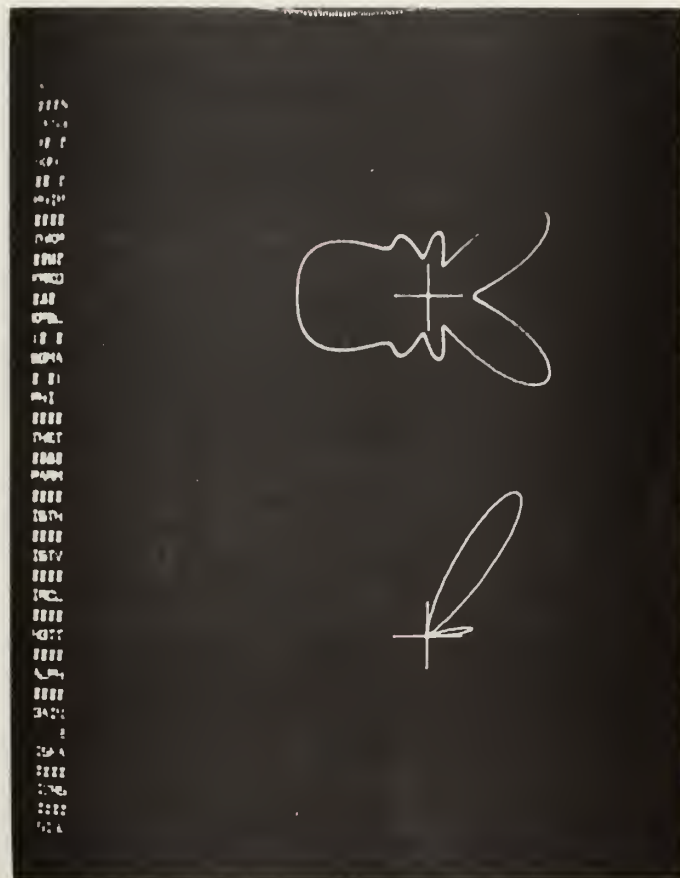
ANTN
0005
LENG
20.0
HGHT
00.0
PHIP
0000
THEP
0045
FREQ
030.
EPSL
10.0
SGMA
0.01
PHI
0000
THET
0080
PARM
0000
ISTH
0000
ISTV
0000
IRCL
0000
HGTT
0000
ALPH
0000
GAIN
ISEA
0000
ICRS
0000
SIGL



Comments: Sloping Long wire; two wavelengths; good ground; $\phi=0$,
 $\theta'=45^\circ$, $\ell=20$, $f=30$ mhz, observation angles $\phi=0$, $\theta=80$.

FIGURE B.16

ANTN
 0005
 LENG
 20.0
 HGHT
 00.0
 PHIP
 0000
 THEP
 0060
 FREQ
 030.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0000
 GAIN
 ISEA
 0000
 ICRS
 SIGL



Comments: Sloping Longwire; two wavelengths; good ground; $\phi'=0, \theta'=60$,
 $\ell=20.0$, $f=30$ mhz; observation angles $\phi=0, \theta=80$. This set of
 two examples demonstrates the effect of variation of tilt
 angle on radiation patterns.

FIGURE B.17

ANTN
 0007
 LENG
 30.0
 HGHT
 10.0
 PHIP
 0000
 THEP
 0000
 FREQ
 030.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0030
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL

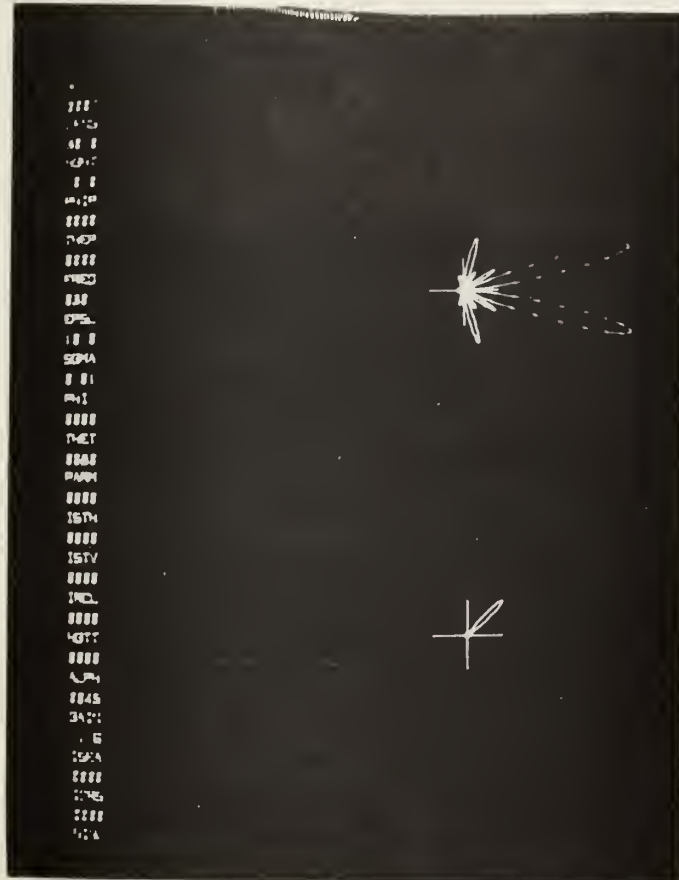


Comments: Horizontal rhombic; three wavelength sides; good ground;
 one wavelength height; $\ell=30.0$, $h=10.0$, $\alpha=30^\circ$, $f=30$ mhz;
 observation angles $\phi=0$, $\theta=80$.

FIGURE B.18

ANTEN
 0007
 LENG
 30.0
 HGHT
 10.0
 PHIP
 0000
 THEP
 0000
 FREQ
 030.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 0000
 ISTE
 0000
 ISTV
 0000
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0045
 GAIN

 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: Horizontal rhombic, three wavelength sides; good ground, one wavelength height; $\ell=30.0$, $h=10.0$ $\alpha=45^\circ$, $f=30$ mhz; observation angles $\Phi=0$, $\Theta=80$. These last two computations show clearly how the program may be used to synthesize antenna systems. A non-optimum α is compared to the optimum for a given h, ℓ etc. Since this antenna is fairly difficult to build, the use of the program to synthesize the optimum is well justified.

FIGURE B.19

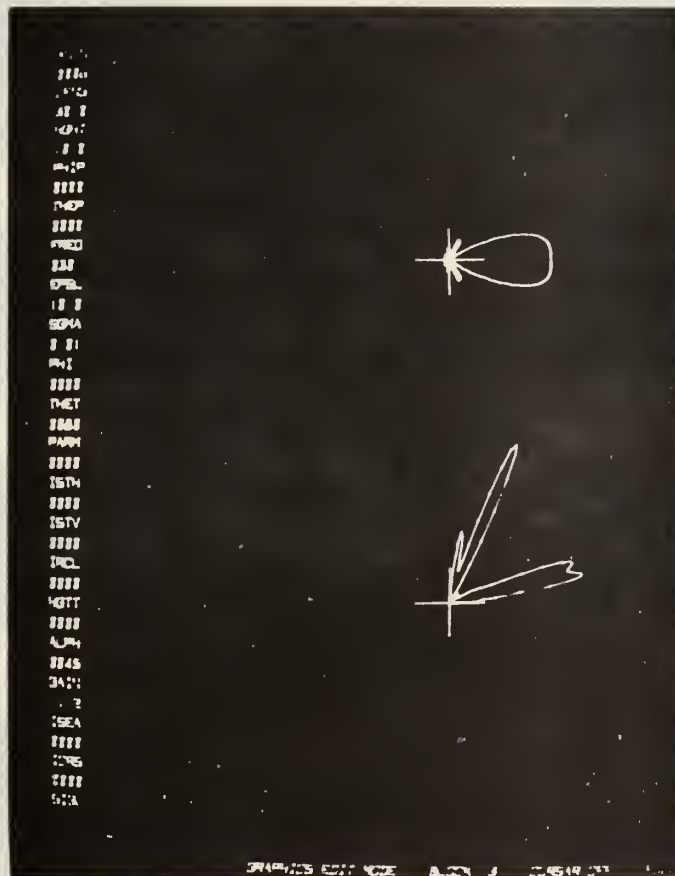
ANTN
 0008
 LENG
 30.0
 HGHT
 10.0
 PHIP
 0000
 THEP
 0000
 FREQ
 050.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0030
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: Vertical half rhombic, three wavelength sides, good ground,
 $\lambda=30.0$, $\alpha=30^\circ$, $f=30\text{mhz}$; observation angles $\phi=0$, $\theta=80$. This
 antenna has a major lobe at approximately 40° elevation and
 may be suited for propagation conditions requiring high
 elevation lobes.

FIGURE B.20

ANTEN
 0008
 LENG
 30.0
 HGHT
 10.0
 PHIP
 0000
 THEP
 0000
 FREQ
 030.
 EPSL
 10.0
 SGMA
 0.01
 PHI
 0000
 THET
 0080
 PARM
 0000
 ISTH
 0000
 ISTV
 0000
 IRCL
 0000
 HGTT
 0000
 ALPH
 0045
 GAIN
 ISEA
 0000
 ICRS
 0000
 SIGL



Comments: Vertical half rhombic, three wavelength aides, good ground,
 $\ell=30.0$, $\alpha=45^\circ$, $f=30$ mhz; observation angles $\phi=0$, $\theta=80$.
 Increasing α splits the energy into a high and low lobe and
 decreases gain from the previous case.

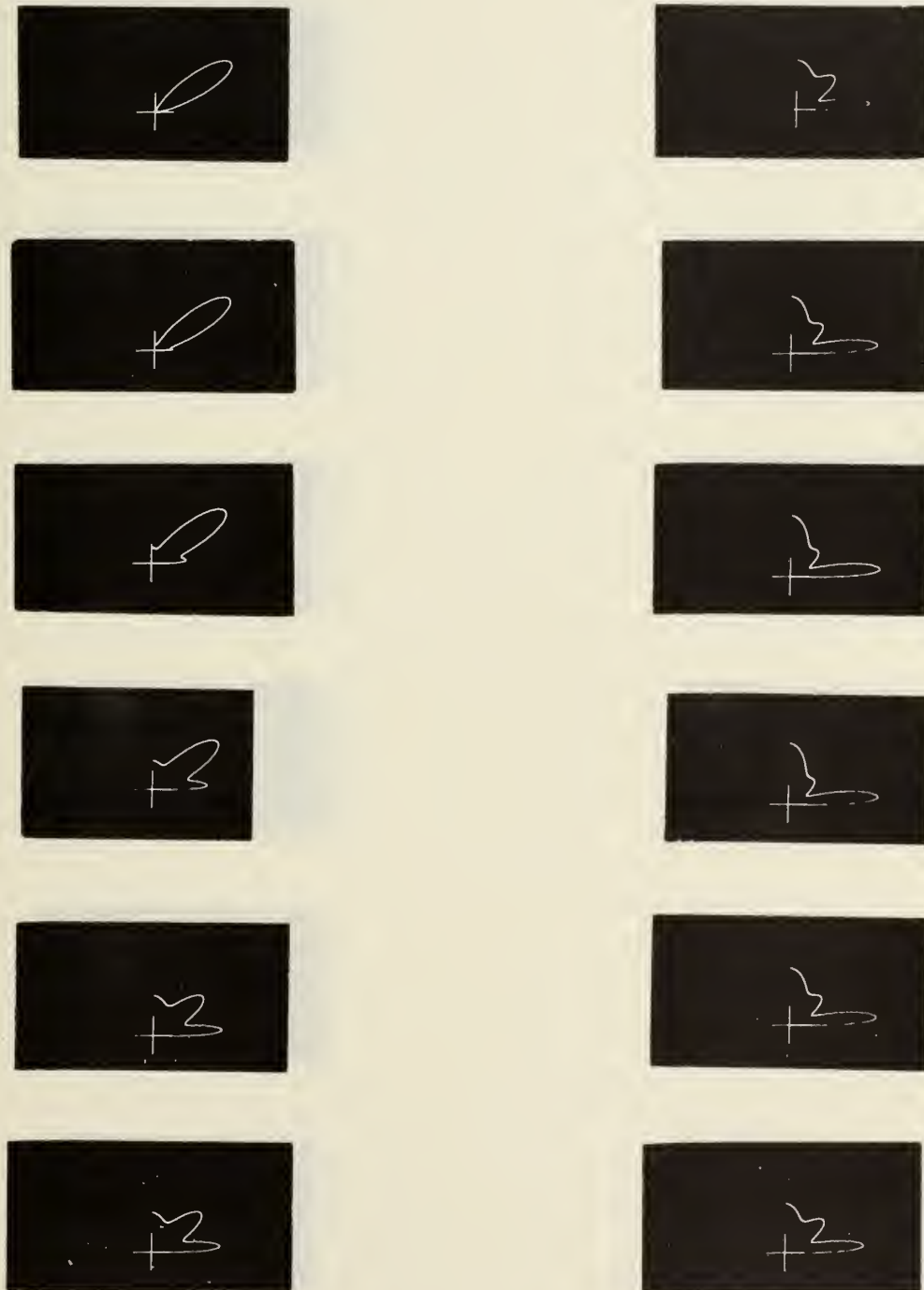


Figure B-21

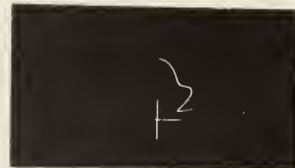
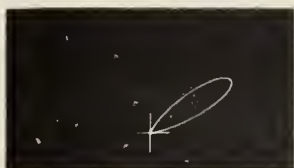
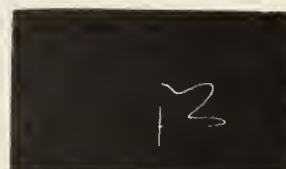
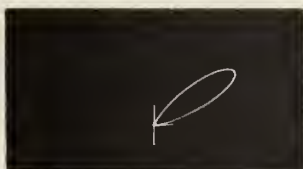
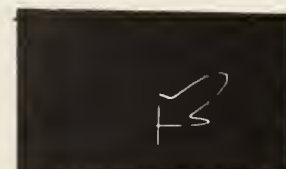
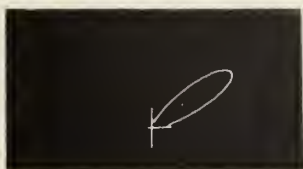
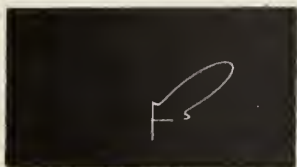
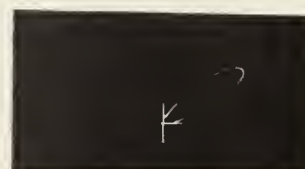
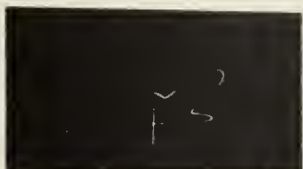
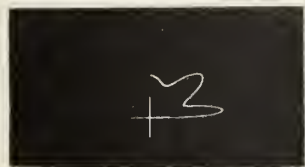


Figure B-21

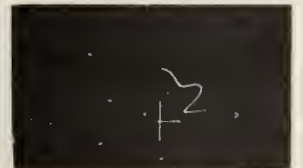
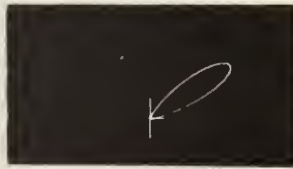
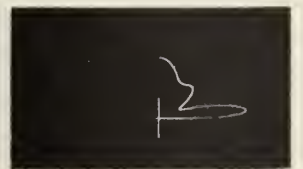
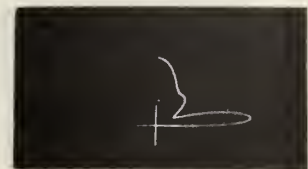
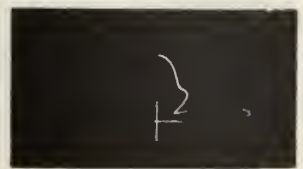
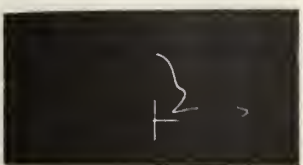
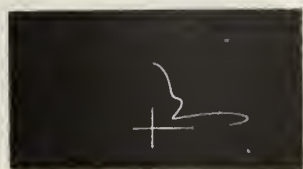


Figure B-21

Vertical Whip
Horizontal Pattern

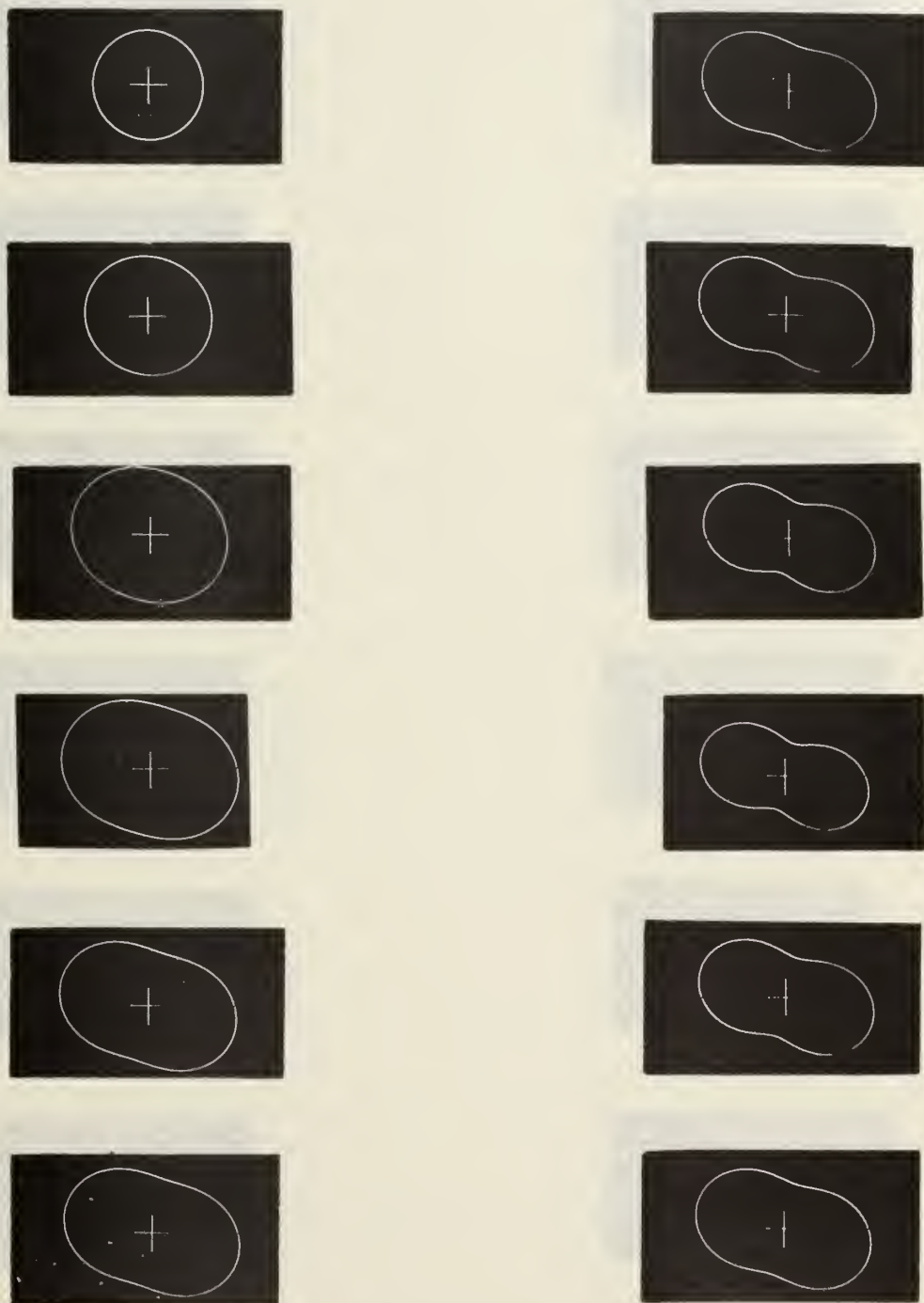


Figure B-21

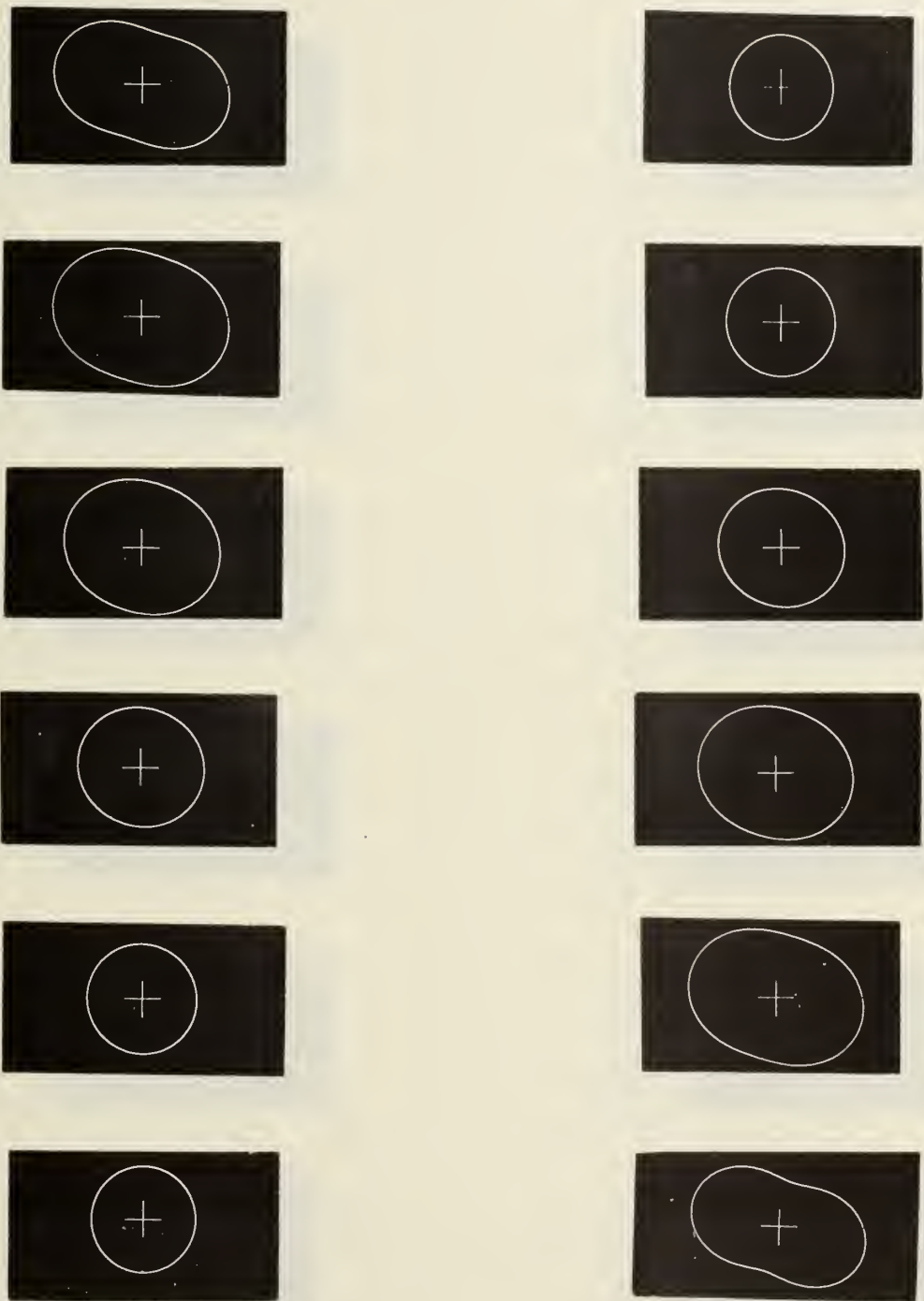


Figure B-21

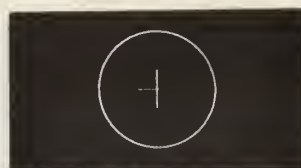
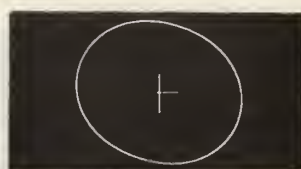
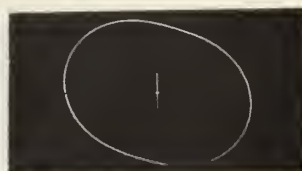
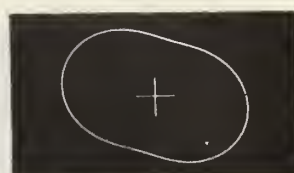
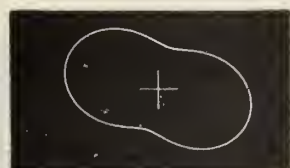
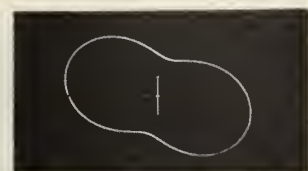
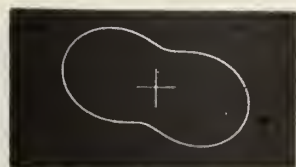
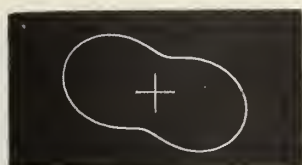


Figure B-21

Half Wave Dipole
Vertical Pattern

Small Ship
Sea State 5 from 045°R

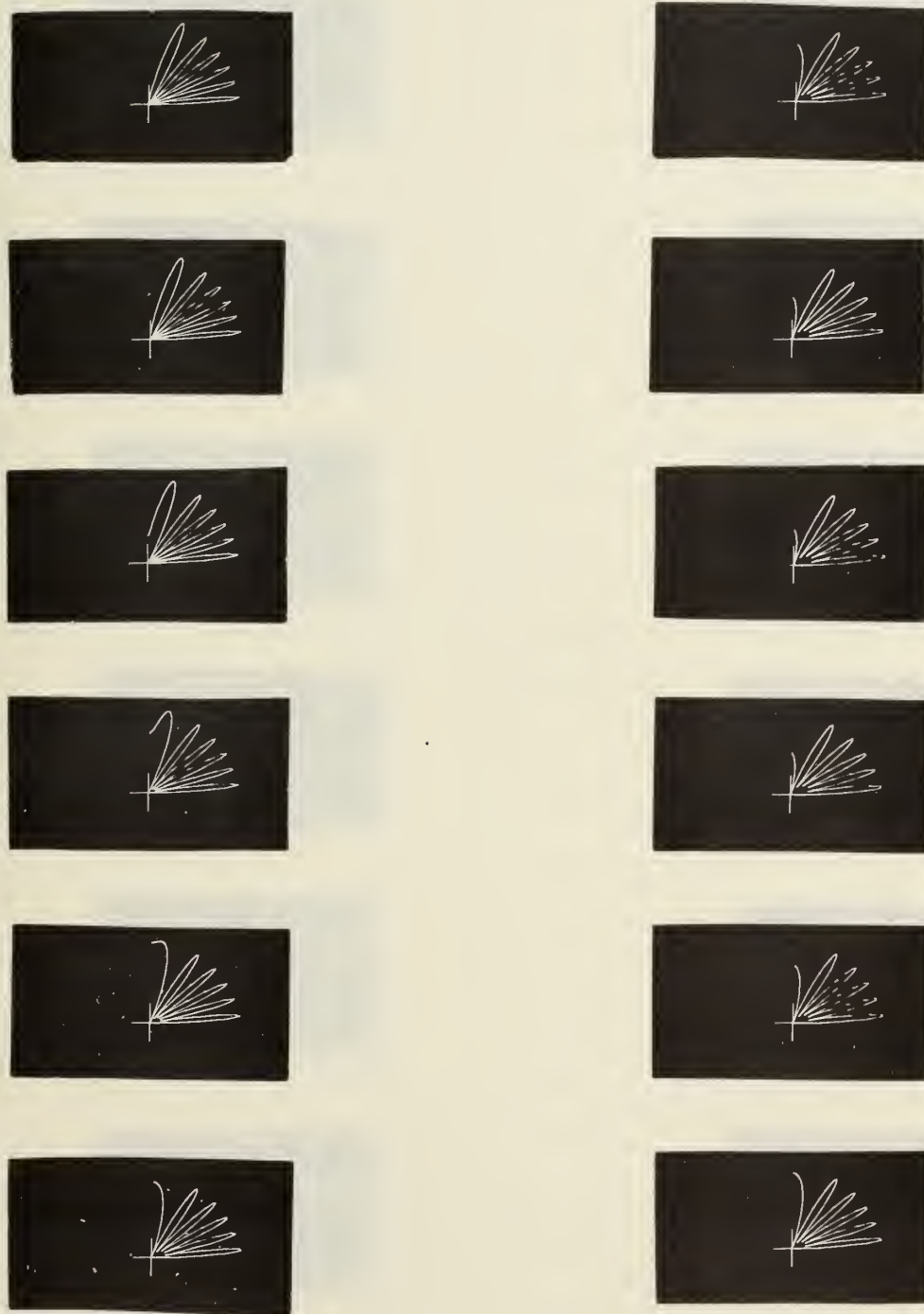


Figure B-22

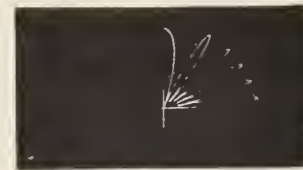
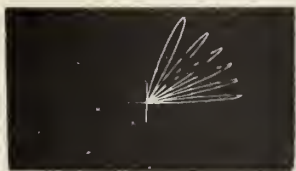
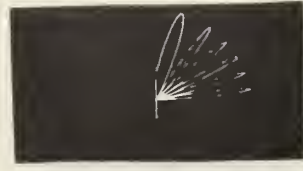
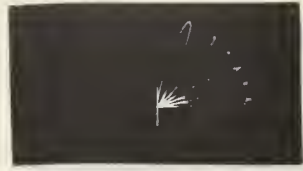


Figure B-22

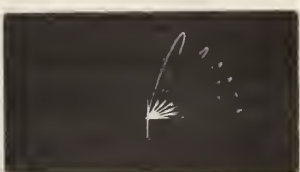
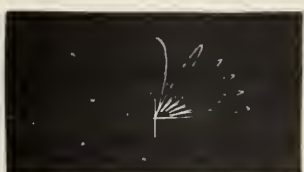
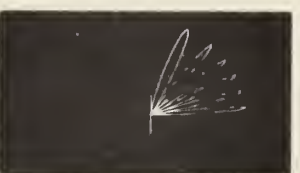
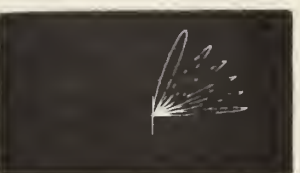
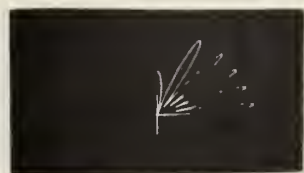
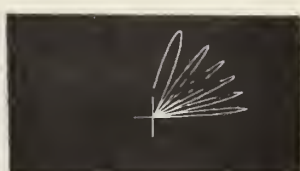
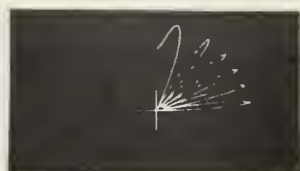
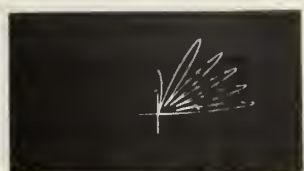
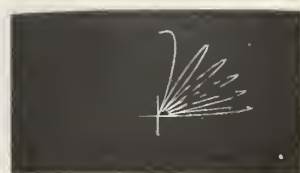
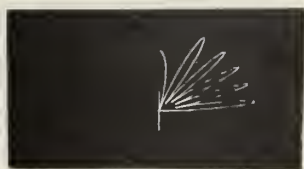


Figure B-22

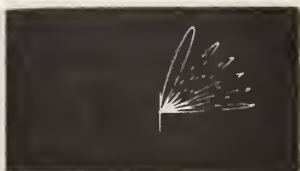
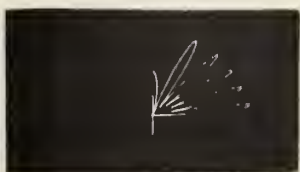
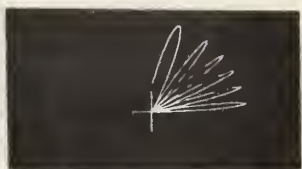
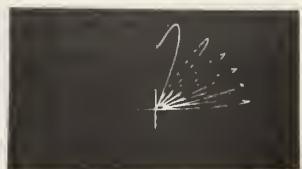
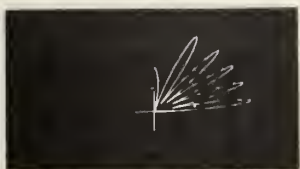
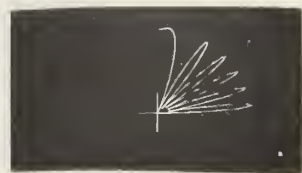
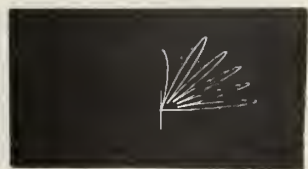
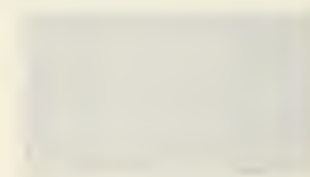
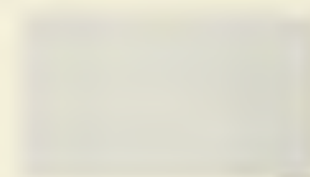
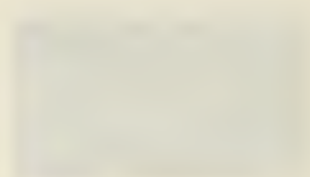
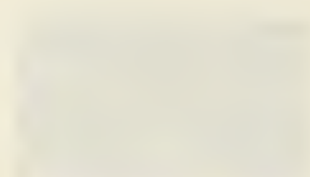


Figure B-22



Half Wave Dipole
Horizontal Pattern

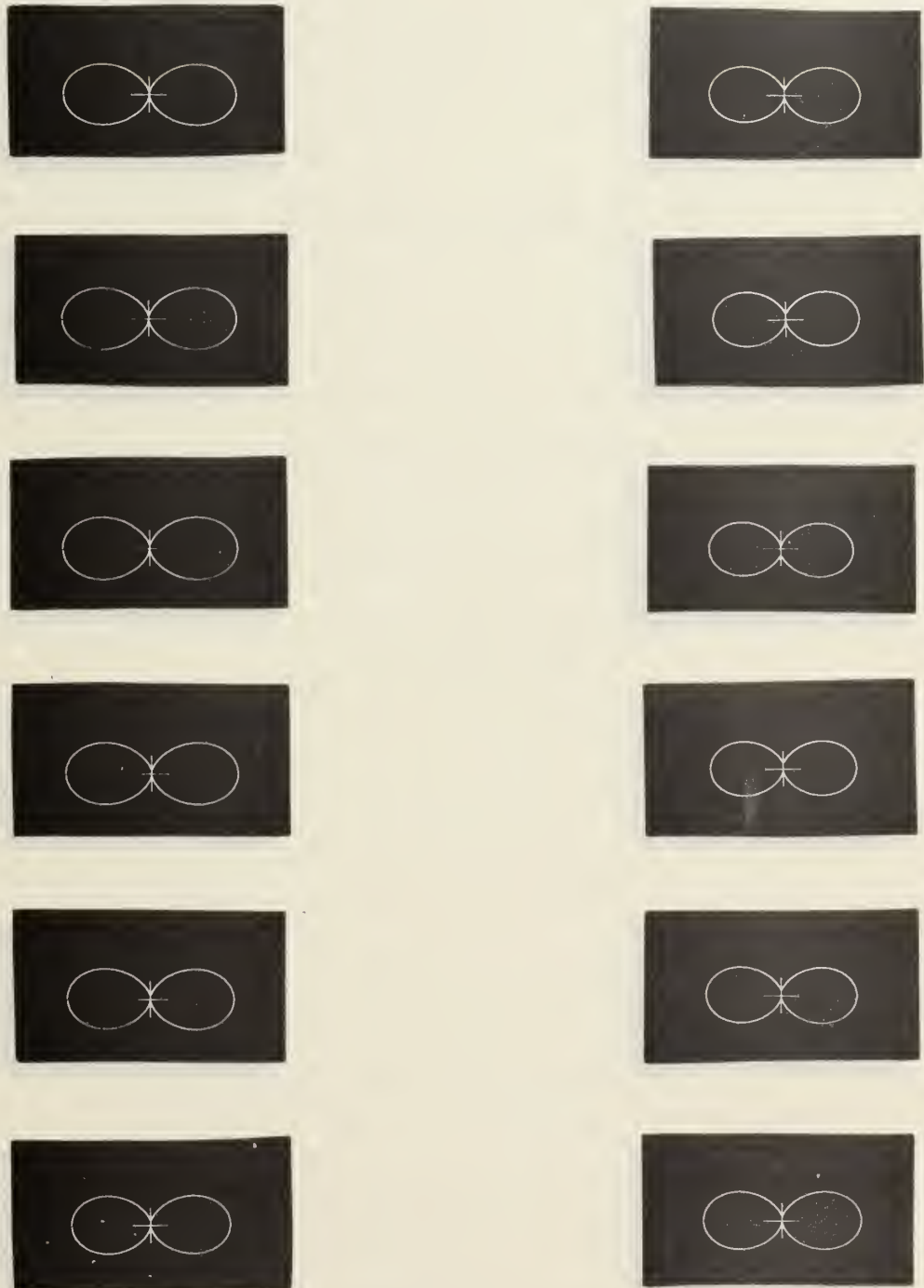


Figure B-22

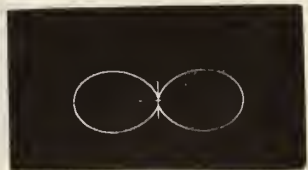
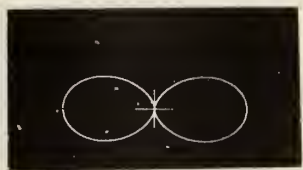
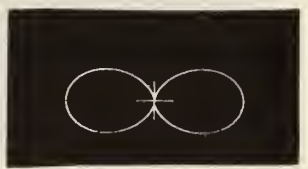
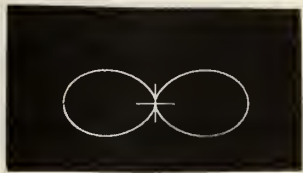
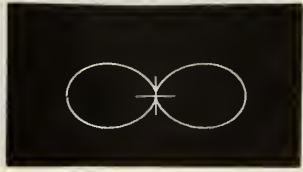


Figure B-22

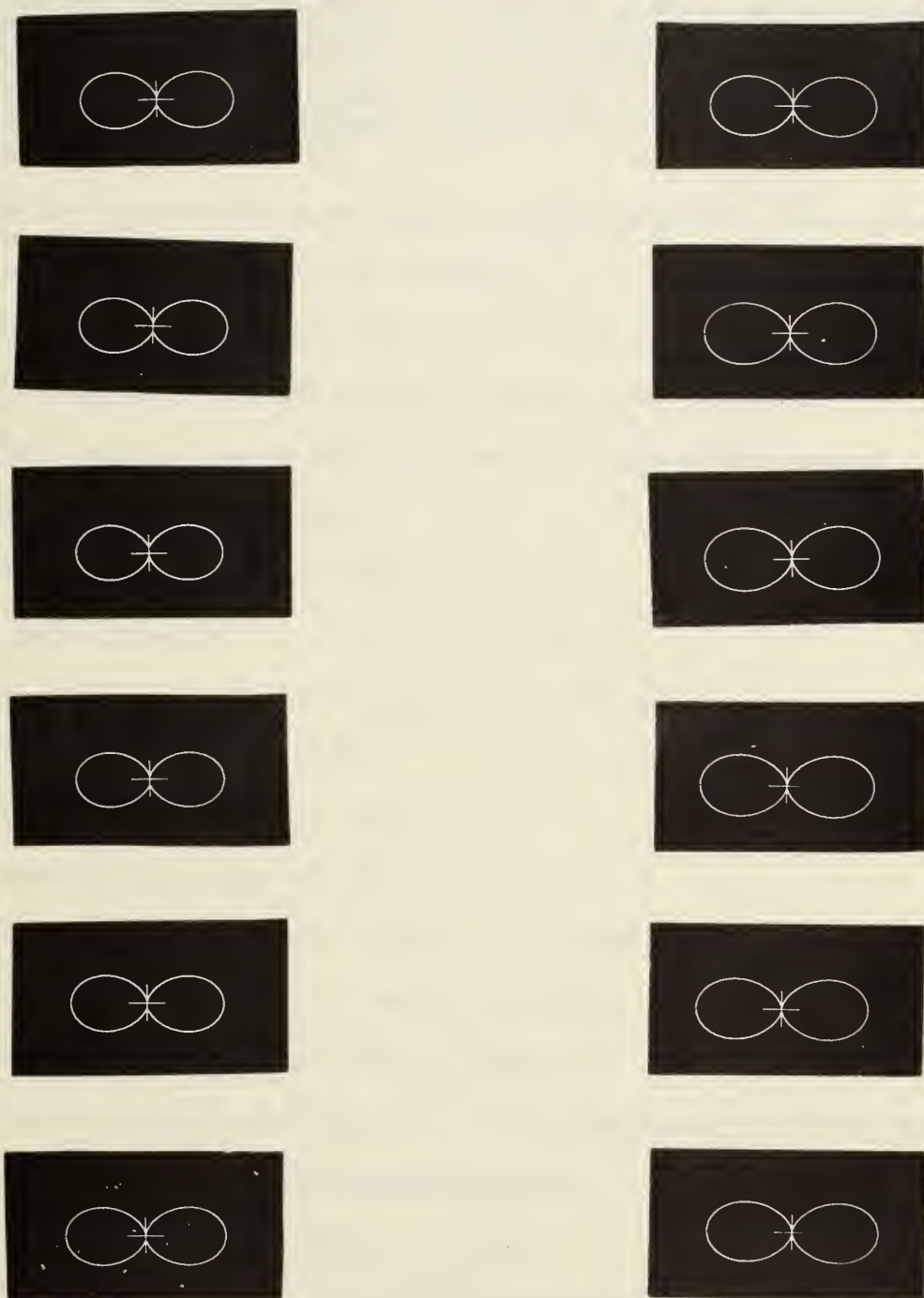


Figure B-22

APPENDIX C

ANTENNA GEOMETRY AND GAIN AND INPUT RESISTANCE EQUATIONS

This appendix details the antenna geometry of the programmed antennas and the equations used in the gain processor and input resistance processor. The sources of equations and geometry are ESSA Technical Report ESSA-ERL-110-ITS 78 and ESSA Technical Report ESSA-ERL-104-ITS 74. The spherical coordinate system used to describe antenna patterns is the IEEE standard and is shown in figure C-1. Antenna geometry for the antennas programmed is shown in figures C-2 thru C-9.

The definitions of the terms used in antenna equations are as follows:

ℓ = length of a unit radiator in meters

h = height of antenna feed point above ground plane

θ' = Tilt angle of the antenna axis measured from the zenith

Δ' = Tilt angle of the antenna axis measured from the horizontal

θ = Observation zenith angle

Δ = Observation elevation angle

α = Apex half angle

α_c = Complement of apex half angle

ϕ = Observation azimuth angle

ϵ_r = Dielectric constant of ground plane

σ = Conductivity of ground plane

R_h = Complex ground reflection factor for a horizontally polarized wave

CH = Magnitude of horizontal reflection factor

Ψ_h = Phase of horizontal reflection factor

R_v = Complex ground reflection factor for a vertically polarized wave

CV = Magnitude of vertical reflection factor

Ψ_v = Phase of vertical reflection factor

$R_{h,n}$ = Complex horizontal reflection factor evaluated for normal incidence

$R_{v,n}$ = Complex vertical reflection factor evaluated for normal incidence

f = Frequency in mhz

λ = wave length in meters

a = radius of ground screen

c = radius of ground screen wire

Equations for quantities that are common to all antennas programmed are as follows:

$$\lambda = \frac{3.0 \times 10^8}{f}$$

$$k = 2\pi/\lambda$$

$$k_2 \approx \underline{\underline{=}} k \left(\epsilon_r - j \frac{1.8 \times 10^4 \sigma}{f} \right)^{1/2}$$

$$R_v = \frac{\cos \theta - \frac{k}{k_2} \left[1 - \left(\frac{k}{k_2} \sin \theta \right)^2 \right]^{1/2}}{\cos \theta + \frac{k}{k_2} \left[1 - \left(\frac{k}{k_2} \sin \theta \right)^2 \right]^{1/2}}$$

$$R_h = \frac{\cos \theta - \frac{k_2}{k} \left[1 - \left(\frac{k}{k_2} \sin \theta \right)^2 \right]^{1/2}}{\cos \theta + \frac{k_2}{k} \left[1 - \left(\frac{k}{k_2} \sin \theta \right)^2 \right]^{1/2}}$$

$$R_v' = \frac{k_2 - k}{k_2 + k}$$

$$R_h' = \frac{k - k_2}{k + k_2}$$

$$S1 = \cos (\psi_h - 2Kh \sin \Delta)$$

$$S2 = \sin (\psi_h - 2Kh \sin \Delta)$$

$$S3 = \cos (\psi_v - 2Kh \sin \Delta)$$

$$S4 = \sin (\psi_v - 2Kh \sin \Delta)$$

C. 1 ARBITRARILY TILTED DIPOLE

The equations presented are for a thin, single element, center fed dipole arbitrarily oriented above a flat ground plane.

$$GI = \frac{\cos (1/2k\ell (\sin\Delta \sin\Delta' + \cos\Delta \cos\Delta' \sin\phi)) - \cos (1/2k\ell)}{1.0 - (\sin\Delta \sin\Delta' + \cos\Delta \cos\Delta' \sin\phi)^2}$$

$$DI = \frac{\cos (1/2k\ell (\cos\Delta \cos\Delta' \sin\phi - \sin\Delta \sin\Delta')) - \cos (1/2k\ell)}{1.0 - (\cos\Delta \cos\Delta' \sin\phi - \sin\Delta \sin\Delta')^2}$$

$$E_{\phi_1} = (\cos\Delta' \sin\phi \sin\Delta - \sin\Delta' \cos\Delta) \cdot GI \\ - (\cos\Delta' \sin\phi \sin\Delta + \sin\Delta' \cos\Delta) \cdot DI \cdot CV \cdot S3$$

$$E_{\phi_1} = \cos\Delta' \cos\phi (GI + DI \cdot CH \cdot S1)$$

$$E_{\theta_2} = (\cos\Delta' \sin\phi \sin\Delta + \sin\Delta' \cos\Delta) \cdot DI \cdot CV \cdot S4$$

$$E_{\phi_2} = \cos\Delta' \cos\phi \cdot DI \cdot CH \cdot S2$$

$$\text{Gain} = 120 \cdot (E_{\theta_1}^2 + E_{\theta_2}^2 + E_{\phi_1}^2 + E_{\phi_2}^2) / R_{in}$$

$$S = 2h$$

$$S_x = S \sin\theta' \cos\phi'$$

$$S_y = S \sin\theta' \sin\phi'$$

$$S_z = S \cos\theta'$$

$$\rho = \left[S_x^2 + (Y_o + S_y)^2 \right]^{1/2}$$

$$\left. \begin{aligned} Y_o &= 2 \cos\Delta' \frac{h}{\lambda} \\ Z_o &= 2 \sin\Delta' \frac{h}{\lambda} \end{aligned} \right\} \text{for mutual impedance}$$

$$\left. \begin{aligned} Y_o &= \sqrt{2} \times 10^{-3} \cdot \ell/\lambda \\ Z_o &= 0 \end{aligned} \right\} \text{for self impedance}$$

$$r = \left[\rho^2 + (Z_o + S_z)^2 \right]^{1/2}$$

$$r_1 = \left[\rho^2 + (Z_o + S_z + 1/2\ell)^2 \right]^{1/2}$$

$$r_2 = \left[\rho^2 + (Z_o + S_z - 1/2\ell)^2 \right]^{1/2}$$

$$SR = 1/r \sin (2\pi r)$$

$$SR1 = 1/r_1 \sin (2\pi r_1)$$

$$SR2 = 1/r_2 \sin (2\pi r_2)$$

$$FACR = 2 \cdot SR \cos (\pi \ell)$$

$$CR = 1/r \cos (2\pi r)$$

$$CR1 = 1/r_1 \cos (2\pi r_1)$$

$$CR2 = 1/r_2 \cos (2\pi r_2)$$

$$FACX = 2 \cdot CR \cdot \cos (\pi \ell)$$

$$Z_{ij} = (R_{ij} + j X_{ij})$$

$$R_{ij} = -30 \int_{-\ell/2}^{\ell/2} \left\{ \left[\frac{1}{\rho^2} \left(SR1 \cdot (S_z + Z_o + \frac{\ell}{2}) + SR2 (S_z + Z_o - \frac{\ell}{2}) \right. \right. \right. \\ \left. \left. \left. - FACR \cdot (S_z + Z_o) \right) \cdot (S_x^2 + Y_o S_y + S_y^2) \right] + S_z (FACR - SR1 - SR2) \right\} \cdot \\ \left[\frac{\sin 2\pi (\frac{\ell}{2} - |S|)}{S} \right] dS$$

$$X_{ij} = -30 \int_{-\ell/2}^{\ell/2} \left\{ \left[\frac{1}{\rho^2} \left(CR1 \cdot (S_z + Z_o + \frac{\ell}{2}) + CR2 \cdot (S_z + Z_o - \frac{\ell}{2}) - FACR (S_z + Z_o) \right) \cdot \right. \right. \\ \left. \left. (S_x^2 + Y_o S_y + S_y^2) \right] + S_z \cdot (FACX - CR1 - CR2) \right\} \left[\frac{\sin 2\pi (\frac{\ell}{2} - |S|)}{S} \right] dS$$

$$Z_{11} \equiv \text{self impedance}$$

$$Z_{21} \equiv \text{mutual impedance}$$

$$R_{in} = R_{11} + \text{Real} \left[Z_{21} (R_h' \cos \Delta' + j R_v' \sin \Delta') (\cos \Delta' - j \sin \Delta') \right]$$

C.2 VERTICAL MONOPOLE

Vertical whip antenna equations are for a base loaded vertical whip above a flat ground plane.

$$S3 = \cos (\Psi_v)$$

$$S4 = \sin (\Psi_v)$$

$$A = \cos (k\ell \sin \Delta) - \cos (k\ell)$$

$$B = \sin (k\ell \sin \Delta) - \sin \Delta \sin (k\ell)$$

$$\text{Gain} = \left\{ \frac{[A \cdot (1 + CV \cdot S3) + B \cdot CV \cdot S4]^2 + [A \cdot CV \cdot S4 + B \cdot (1 - CV \cdot S3)]^2}{R_{in} \cos^2 \Delta} \right\}$$

$$R_{in} = 15 \left\{ [2 + 2 \cos (2k\ell)] \cdot [\ln (2k\ell) + \gamma - Ci (2k\ell)] - \right. \\ \left. \cos (2k\ell) \cdot [\ln (4k\ell) + \gamma - Ci (4k\ell)] - \right. \\ \left. 2 \sin (2k\ell) \cdot \left[\frac{\pi}{2} + Si (2k\ell) \right] + \sin (2k\ell) \cdot \left[\frac{\pi}{2} + Si (4k\ell) \right] \right\}$$

$$\gamma = .577$$

$$Ci(x) = - \int_x^{\infty} \frac{\cos t}{t} dt$$

$$Si(x) = - \int_x^{\infty} \frac{\sin t}{t} dt$$

C.3 VERTICAL WHIP WITH GROUND SCREEN

The vertical whip with ground screen is a single monopole above a flat ground with a radial conductor ground system consisting of N equally spaced radial conductors. A value of 120 is used for N and 1 cw wire is assumed.

$$S3 = \cos \psi_v$$

$$S4 = \sin \psi_v$$

$$A = \cos(k\ell \sin \Delta) - \cos(k\ell)$$

$$B = \sin(k\ell \sin \Delta) - \sin \Delta \sin(k\ell)$$

$$\text{Gain} = \frac{[A \cdot (1 + CV \cdot S3) + B \cdot CV \cdot S4]^2 + [A \cdot CV \cdot S4 + B \cdot (1 - CV \cdot S3)]^2 [A_3^2 + B_3^2]}{R_{in}}$$

$$A_3 + jB_3 = \frac{1 - \eta \sin \theta \int_0^{ka} [e^{-j(x^2 + k^2 \ell^2)^{1/2}} - e^{-jx \cos(k\ell)}] J_1(x \sin \theta) dx}{120\pi \sin(k\ell) [\cos(k\ell \cos \theta) - \cos(k\ell)]}$$

$$\eta = \left[\frac{j\mu\omega}{\sigma + j\omega\epsilon} \right]^{1/2}$$

$$R_1 = R_{in} \text{ of vertical whip (C.2)}$$

$$\begin{aligned} \Delta Z_1 = & \frac{\eta}{4\pi \sin^2(k\ell)} \left\{ e^{j2k\ell} \text{Ei}[-j2k(r_o + \ell)] + e^{-j2k\ell} \text{Ei}[-j2k(r_o - \ell)] \right. \\ & + 2\cos^2(k\ell) \text{Ei}[-j2ka] + 4 \cos(k\ell) \text{Ei}[-jkr_1] \\ & \left. - 4\cos(k\ell) e^{-jk\ell} \text{Ei}[-jk(r_1 - \ell)] - 4 \cos(k\ell) e^{jk\ell} \text{Ei}[-jk(r_1 + \ell)] \right\} \\ \Delta Z_2 = & - \int_0^a \left[\frac{\eta_e \eta}{\eta + \eta_e} \right] \left[\frac{(e^{-jk(\rho^2 + \ell^2)^{1/2}} - e^{-jk\rho \cos(k\ell)})^2}{2\pi\rho \sin^2(k\ell)} \right] d\rho \end{aligned}$$

$$\text{Ei}(\pm jx) = \text{Ci}(x) \pm j \text{Si}(x)$$

$$r_o = (a^2 + \ell^2)^{1/2}$$

$$r_1 = a + (a^2 + \ell^2)^{1/2}$$

$$\eta = \left[\frac{j\omega\mu}{\sigma + j\omega\epsilon} \right]^{1/2}$$

$$\eta_e = \left(\frac{j 240 \pi^2 \rho}{N\lambda} \right) \ln \left(\frac{\rho}{Nc} \right)$$

$$R_{in} = R_1 + \text{Real}(\Delta Z_1 + \Delta Z_2)$$

C. 4 INVERTED L

The inverted L antenna equations are for a long wire antenna that is base loaded and arranged in an inverted L configuration.

$$A = \cos(k\ell) \cos(kh \sin \Delta) - \sin \Delta \sin(k\ell) \sin(kh \sin \Delta) \\ - \cos(k(h+\ell))$$

$$B = \sin \Delta \sin(k\ell) \cos(kh \sin \Delta) + \cos(k\ell) \sin(kh \sin \Delta) \\ - \sin \Delta \sin(k(h+\ell))$$

$$GI = \sin(k\ell \cos \Delta \sin \phi) - \cos \Delta \cos \phi \sin(k\ell)$$

$$GR = \cos(k\ell \cos \Delta \sin \phi) - \cos(k\ell)$$

$$|E_{\theta}|^2 = \left[\frac{\sin \phi \sin \Delta [GR(1.0 - CV \cdot S3) + GI \cdot CV \cdot S4]}{1.0 - \cos^2 \Delta \sin^2 \phi} \right. \\ \left. - \frac{A(1.0 + CV \cdot \cos \psi_v + B CV \cdot \sin \psi_v)}{\cos \Delta} \right]^2 \\ + \left[\frac{\sin \phi \sin \Delta (GI(1.0 - CV \cdot S3) - GR \cdot CV \cdot S4)}{1.0 - \cos^2 \Delta \sin^2 \phi} \right. \\ \left. - \frac{B(1.0 - CV \cdot \cos \psi_v) + A \cdot CV \cdot \sin \psi_v}{\cos \Delta} \right]^2$$

$$|E_{\phi}|^2 = \left[\frac{\cos \phi}{1 - \cos^2 \Delta \cos^2 \phi} \right] \cdot \left\{ [GR(1.0 + CH \cdot S1) - GI \cdot CH \cdot S2]^2 \right. \\ \left. + [GI(1.0 + CH \cdot S1) + GR \cdot CH \cdot S2]^2 \right\}$$

$$\text{Gain} = 30.0 \left[|E_{\phi}|^2 + |E_{\theta}|^2 \right]$$

$$Si(x) = \int_x^{\infty} \frac{\sin x}{x} dx$$

$$Rin = 60 \left[1.41 + \ln \left(\frac{2\ell}{\lambda} \right) + \frac{\sin(2k\ell)}{2k\ell} \right] \\ + 30.0 \left\{ -1/2 \cos(2kh) \left[(\ln(2kh) + 1.270 - Ci(4kh)) \right] \right. \\ \left. + (1.0 + \cos(2kh)) \left[\ln(2kh) + 0.577 - Ci(2kh) \right] \right. \\ \left. - \sin(2kh) \left[1/2 Si(4kh) - Si(4k\ell) \right] \right\}$$

C.5 SLOPING LONG WIRE

The equations for this antenna are for a base loaded longwire antenna arranged in a sloping configuration; the antenna zenith angle may be assigned values of 0 thru 90 degrees.

$$CIG = \frac{\cos [kl (\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos \phi)] - \cos (kl)}{1.0 - (\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos \phi)^2}$$

$$SIG = \frac{\sin [kl (\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos \phi)] - (\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos \phi) \sin (kl)}{1.0 - (\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos \phi)^2}$$

$$CIGP = \frac{\cos [kl (\cos \Delta \cos \Delta' \cos \phi - \sin \Delta \sin \Delta')] - \cos (kl)}{1.0 - (\cos \Delta \cos \Delta' \cos \phi - \sin \Delta \sin \Delta')^2}$$

$$SIGP = \left[\frac{\sin [kl (\cos \Delta \cos \Delta' \cos \phi - \sin \Delta \sin \Delta')]}{1.0 - (\cos \Delta \cos \Delta' \cos \phi - \sin \Delta \sin \Delta')^2} + (\sin \Delta \sin \Delta' - \cos \Delta \cos \Delta' \cos \phi) \sin (kl) \right]$$

$$E_{\phi_1} = -\cos \Delta' \sin \phi \left[CIG + CH \cdot (CIGP \cos \psi_h - SIGP \sin \psi_h) \right]$$

$$E_{\phi_2} = -\cos \Delta' \sin \phi \left[SIG + CH (CIGP \sin \psi_h - SIGP \cos \psi_h) \right]$$

$$E_{\theta_1} = CIG (\cos \Delta' \cos \phi \sin \Delta - \sin \Delta' \sin \Delta) + CV (\cos \Delta' \cos \phi \sin \Delta + \sin \Delta' \cos \Delta) \cdot [CIGP \cos \psi_v - SIGP \sin \psi_v]$$

$$E_{\theta_2} = SIG (\cos \Delta' \cos \phi \sin \Delta - \sin \Delta' \cos \Delta) - CV (\cos \Delta' \cos \phi \sin \Delta + \sin \Delta' \cos \Delta) \cdot [CIGP \sin \psi_v + SIGP \cos \psi_v]$$

$$\text{Gain} = 30 \left[E_{\phi_1}^2 + E_{\phi_2}^2 + E_{\theta_1}^2 + E_{\theta_2}^2 \right] / R_{in}$$

$$R_{in} = 30 \left\{ \frac{1}{2} \left[\ln (kl) + .577 + Ci (4kl) \right] \right.$$

$$+ 0.693 + \cos (kl) \left[\cos kl (\ln (kl) + .577 \right.$$

$$- 2.0 Ci (2kl) + Ci (4kl)) - \sin (kl) (Si (4kl)$$

$$- 2.0 Si (2kl) \left. \right\}$$

$$Ci(x) = - \int_x^\infty \frac{\cos t}{t} dt$$

$$Si(x) = - \int_x^\infty \frac{\sin t}{t} dt$$

C. 6 TERMINATED SLOPING VEE

The terminated sloping vee equations are for two sloping longwire antennas arranged in a vee configuration. The feed point is the apex of the vee. The elements are fed 180 degrees out of phase. The elements of the vee are terminated in 370 ohm non inductive resistors.

$$\cos \psi_1 = \sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos (\phi - \alpha)$$

$$\cos \psi_2 = \sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos (\phi + \alpha)$$

$$\cos \psi_3 = -\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos (\phi - \alpha)$$

$$\cos \psi_4 = -\sin \Delta \sin \Delta' + \cos \Delta \cos \Delta' \cos (\phi + \alpha)$$

$$\cos \psi_5 = \cos \Delta \sin \Delta' + \sin \Delta \cos \Delta' \cos (\phi - \alpha)$$

$$\cos \psi_6 = \cos \Delta \sin \Delta' + \sin \Delta \cos \Delta' \cos (\phi + \alpha)$$

$$\cos \psi_7 = -\cos \Delta \sin \Delta' + \sin \Delta \cos \Delta' \cos (\phi - \alpha)$$

$$\cos \psi_8 = -\cos \Delta \sin \Delta' + \sin \Delta \cos \Delta' \cos (\phi + \alpha)$$

$$U_i = k\ell (1.0 - \cos \psi_i) \quad i = 1, 2, 3, 4$$

$$A = \frac{\cos \psi_7 (\cos (U_1) - 1.0)}{U_1} - \frac{\cos \psi_8 (\cos (U_2) - 1)}{U_2}$$

$$+ CV \cdot \left\{ \frac{\cos \psi_6 \left[S3 (\cos (U_4) - 1.0) S4 + \sin (U_4) \right]}{U_4} \right.$$

$$\left. - \cos \psi_5 \frac{[\cos (U_3) - 1.0] S3 + \sin (U_3) S4}{3} \right\}$$

$$B = \frac{\cos \psi_8 \sin (U_2)}{U_2} - \frac{\cos \psi_7 \sin (U_1)}{U_1}$$

$$+ CV \left\{ \frac{\cos \psi_5 \left(\sin (U_3) S3 - (\cos (U_3) - 1.0) S4 \right)}{3} \right.$$

$$+ \cos \psi_6 \left(\frac{(\cos (U_4) - 1.0) S4 - \sin (U_4) S3}{U_3} \right)$$

$$\left. + \cos \psi_6 \left(\frac{(\cos (U_4) - 1.0) S4 - \sin (U_4) S3}{U_4} \right) \right\}$$

$$\begin{aligned}
C &= \frac{\sin(\phi + \alpha)(\cos(U_2) - 1.0)}{U_2} - \frac{\sin(\phi - \alpha)(\cos(U_1) - 1.0)}{U_1} \\
&+ CH \left\{ S1 \left[\frac{\sin(\phi + \alpha)(\cos(U_4) - 1.0)}{U_4} - \frac{\sin(\phi - \alpha)(\cos(U_3) - 1.0)}{U_3} \right] \right. \\
&\left. - S2 \left[\frac{\sin(\phi - \alpha)\sin(U_3)}{U_3} - \frac{\sin(\phi + \alpha)\sin(U_4)}{U_4} \right] \right\} \\
D &= \frac{\sin(\phi - \alpha)\sin(U_3)}{U_3} - \frac{\sin(\phi + \alpha)\sin(U_4)}{U_4} \\
&+ CH \left\{ \left[\frac{\sin(\phi - \alpha)\sin(U_3)}{U_3} - \frac{\sin(\phi + \alpha)\sin(U_4)}{U_4} \right] \cdot S1 \right. \\
&\left. + \left[\frac{\sin(\phi + \alpha)(\cos(U_4) - 1.0)}{U_4} - \frac{\sin(\phi - \alpha)(\cos(U_3) - 1.0)}{U_3} \right] S2 \right\} \\
Gain &= 0.05 \left[A^2 + B^2 + \cos^2 \Delta' (C^2 + D^2) \right]
\end{aligned}$$

C. 7 HORIZONTAL RHOMBIC

The horizontal rhombic antenna equations were developed under the assumption of uniform current distribution of the effective value of current. The antenna is loaded at the apex and terminated in dissipation lines at the opposite corner.

$$U_1 = 1.0 - \cos \Delta (\sin \alpha_c \cos \phi + \cos \alpha_c \sin \phi)$$

$$U_2 = 1.0 - \cos \Delta (\sin \alpha_c \cos \phi - \cos \alpha_c \sin \phi)$$

$$\text{Gain} = 2.16 \left[\frac{\cos \alpha_c \sin (1/2 k \ell U_1) \sin (1/2 k \ell U_2)}{U_1 U_2} \right]^2$$

$$\left[\left(\cos \phi - \sin \alpha \cos \right)^2 (CH^2 + 1.0 + 2.0 \cdot CH \cdot S1) + \right. \\ \left. \sin^2 \Delta \sin^2 \phi (CV^2 + 1.0 - 2.0 \cdot CV \cdot S3) \right]$$

C. 8 VERTICAL HALF RHOMBIC

The vertical half rhombic are for a base loaded longwire arranged in the vertical half rhombic configuration and terminated in a 400-500 ohm non-inductive resistor.

$$S1 = \sin (K (1.0 - \cos \psi_1))$$

$$C1 = \cos (K (1.0 - \cos \psi_1))$$

$$S2 = \sin (K (1.0 - \cos \psi_2))$$

$$C2 = \cos (K (1.0 - \cos \psi_2))$$

$$\cos \psi_1 = \cos \Delta \cos \Delta' \cos \phi - \sin \Delta \sin \Delta'$$

$$\cos \psi_2 = \cos \Delta \cos \Delta' \cos \phi + \sin \Delta \sin \Delta'$$

$$R1 = \frac{1 - C1}{1.0 - \cos \psi_1}$$

$$I1 = \frac{S1}{1.0 - \cos \psi_1}$$

$$R2 = \frac{C1 (1.0 - C2) + S1 \cdot S2}{1.0 - \cos \psi_2}$$

$$I2 = \frac{C1 \cdot S2 - S1 (1.0 - C2)}{1.0 - \cos \psi_2}$$

$$R3 = (1.0 - C1) \cos (2k\ell \sin \Delta' \sin \Delta) + S1 \sin (2k\ell \sin \Delta' \sin \Delta)$$

$$I3 = S1 \cos (2k\ell \sin \Delta' \sin \Delta) - (1.0 - C1) \sin \left(\frac{4\pi\ell}{\lambda} \sin \Delta' \sin \Delta \right)$$

$$F1 = \frac{I3 \cdot C1 - R3 \cdot S1}{1.0 - \cos \psi_1}$$

$$F2 = R3 \cdot C1 + I3 \cdot S1$$

$$F3 = \frac{1.0 - C2}{1.0 - \cos \psi_2}$$

$$F4 = \frac{S2}{1 - \cos \psi_2}$$

$$RB = R1 + R2 - CV [(F2 + F3) S3 - (F1 + F4) S4]$$

$$B1 = I1 + I2 - CV [(F2 + F3) S4 + (F1 + F4) S3]$$

$$RC = R2 - R1 + CV [(F2 - F3) S3 - (F1 - F4) S4]$$

$$CC = I2 - I1 + CV [(F2 - F3) S4 + (F1 - F4) S3]$$

$$RA = R1 + R2 + CH [(F2 + F3) S1 - (F1 + F4) S2]$$

$$A1 = I1 + I2 + CH [(F2 + F3) S2 + (F1 + F4) S1]$$

$$\begin{aligned} \text{Gain} = 0.1 & \left[(RB \cos \Delta' \cos \phi \sin \Delta + RC \sin \Delta' \cos \Delta)^2 \right. \\ & + (B1 \cos \Delta' \cos \phi \sin \Delta + CC \sin \Delta' \cos \Delta)^2 \\ & \left. + (RA \cos \Delta' \sin \phi)^2 + (A1 \cos \Delta' \sin \phi)^2 \right] \end{aligned}$$

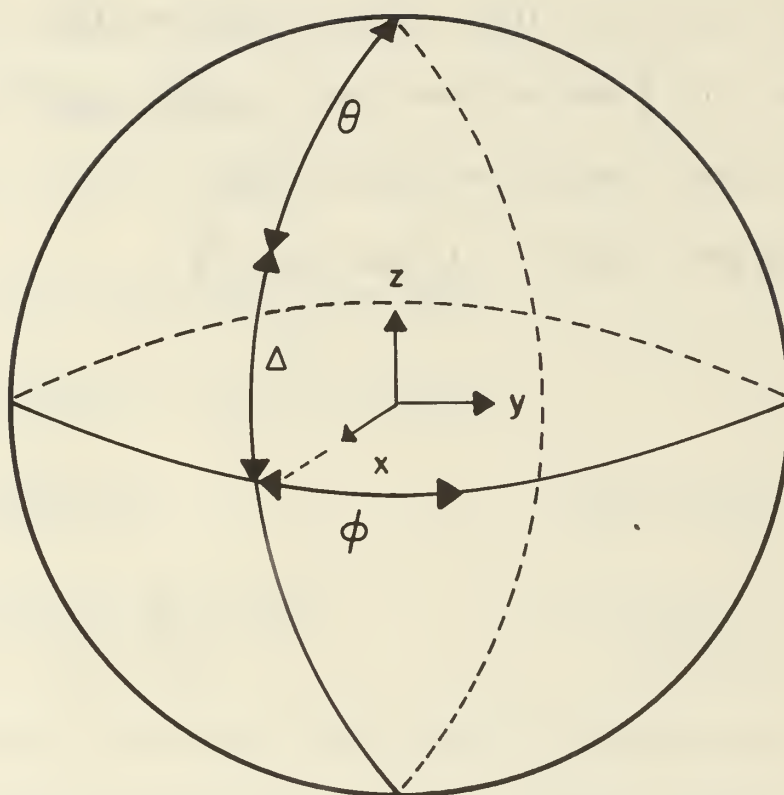
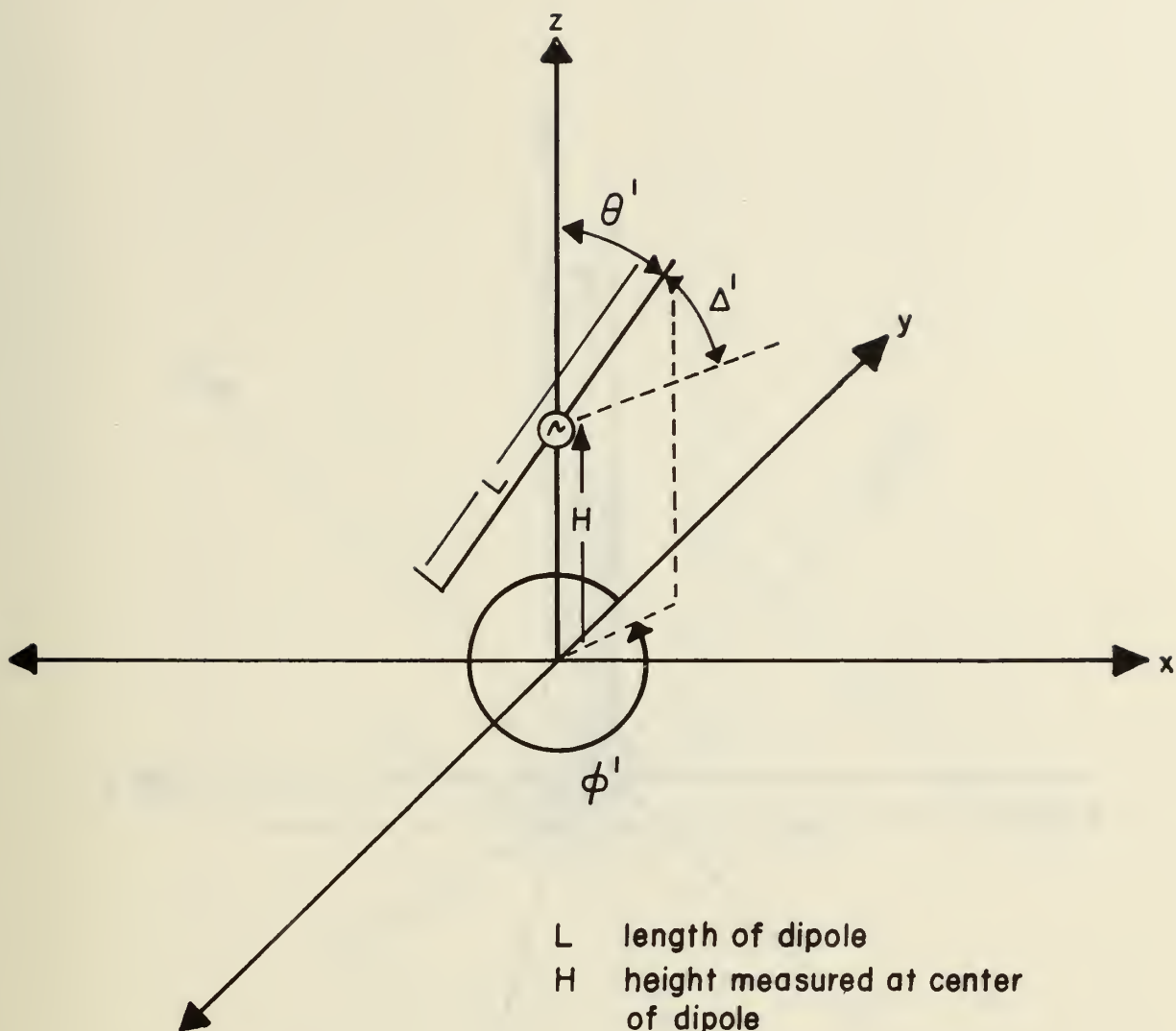


Figure C-1
Spherical Coordinate System



- L length of dipole
- H height measured at center of dipole
- θ' Tilt angle (measured in zenith)
- Δ' Tilt angle (measured in elevation above horizontal)
- ϕ' Train angle (measured in azimuth)

Figure C-2
Arbitrarily Tilted Dipole Geometry

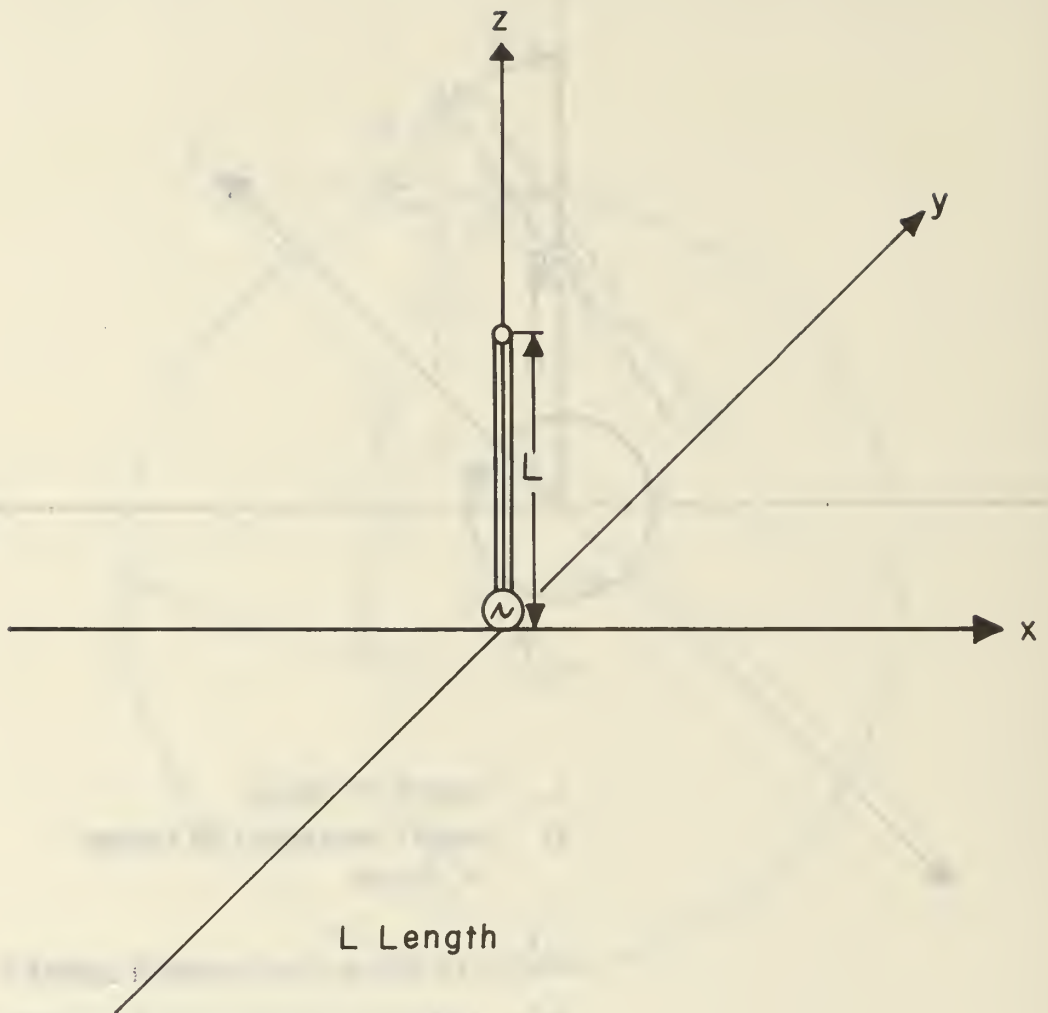
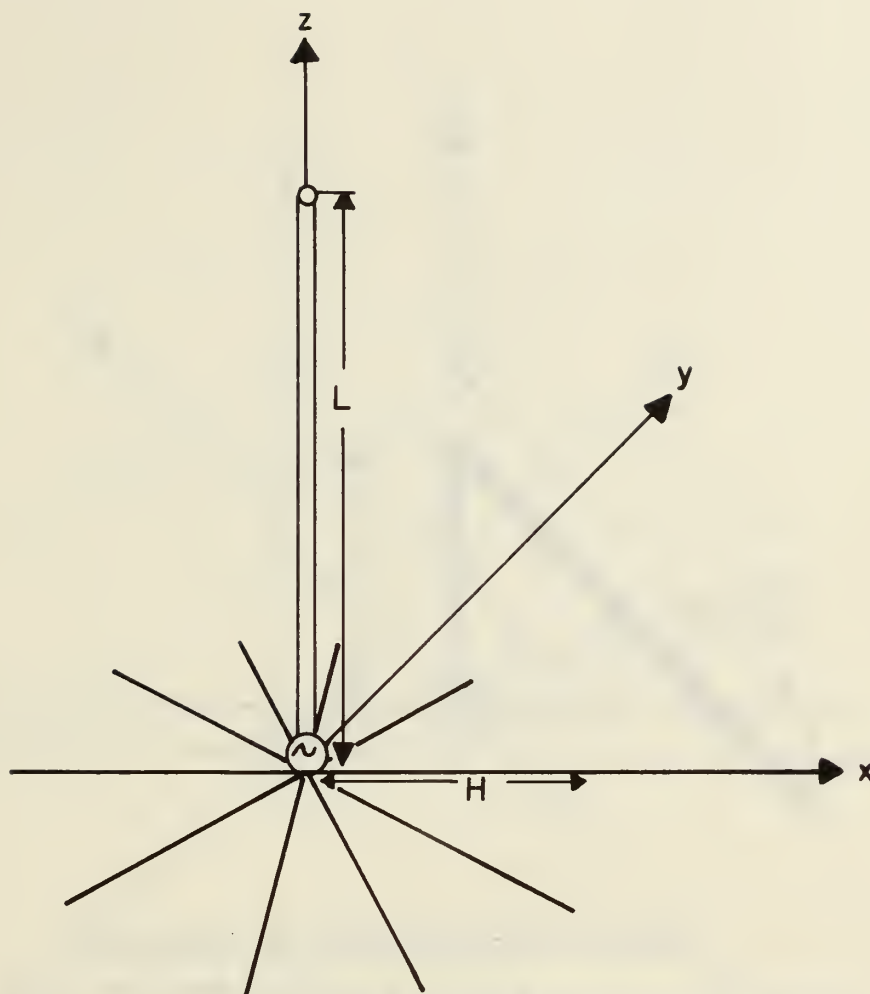


Figure C-3
Vertical Whip Geometry



- L length of whip
- H radius of ground screen radial elements

Figure C - 4
Vertical Whip with Ground Screen Geometry

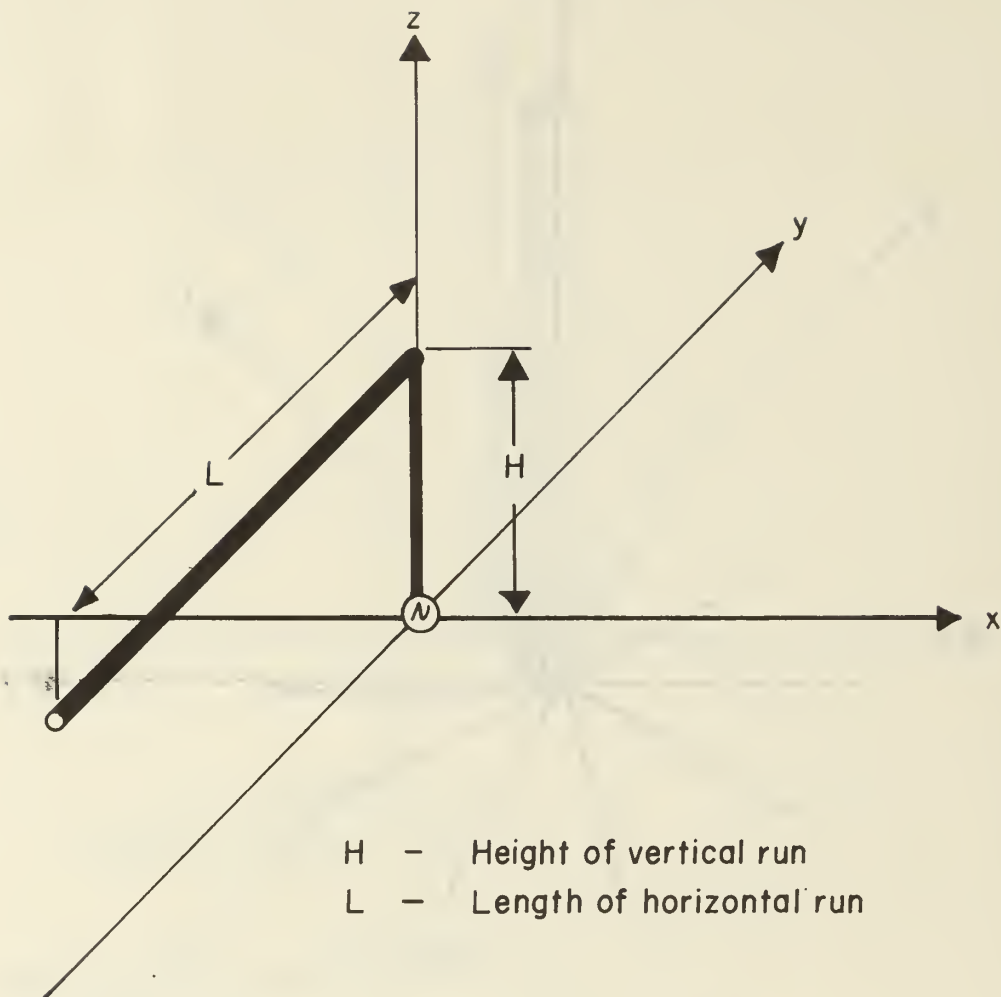


Figure C - 5

Inverted L Geometry

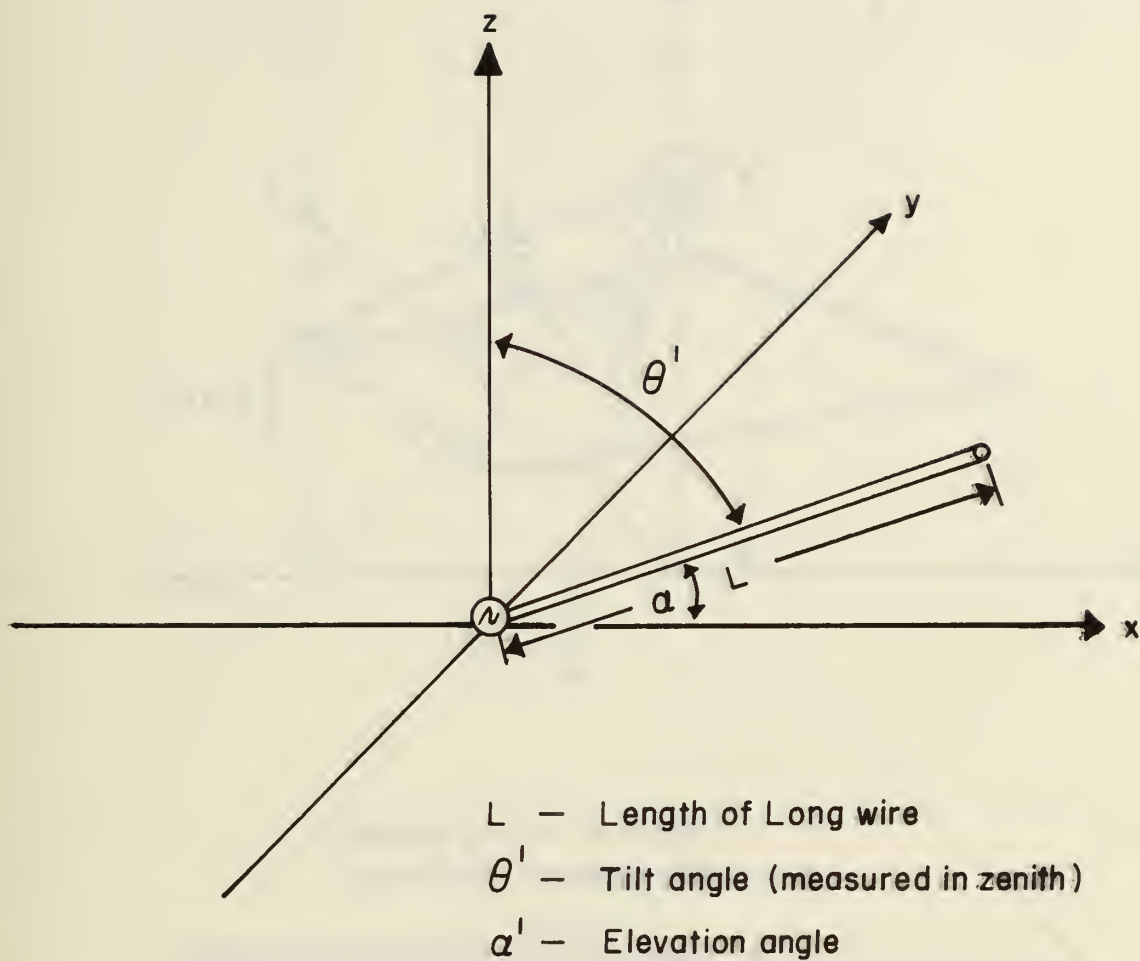
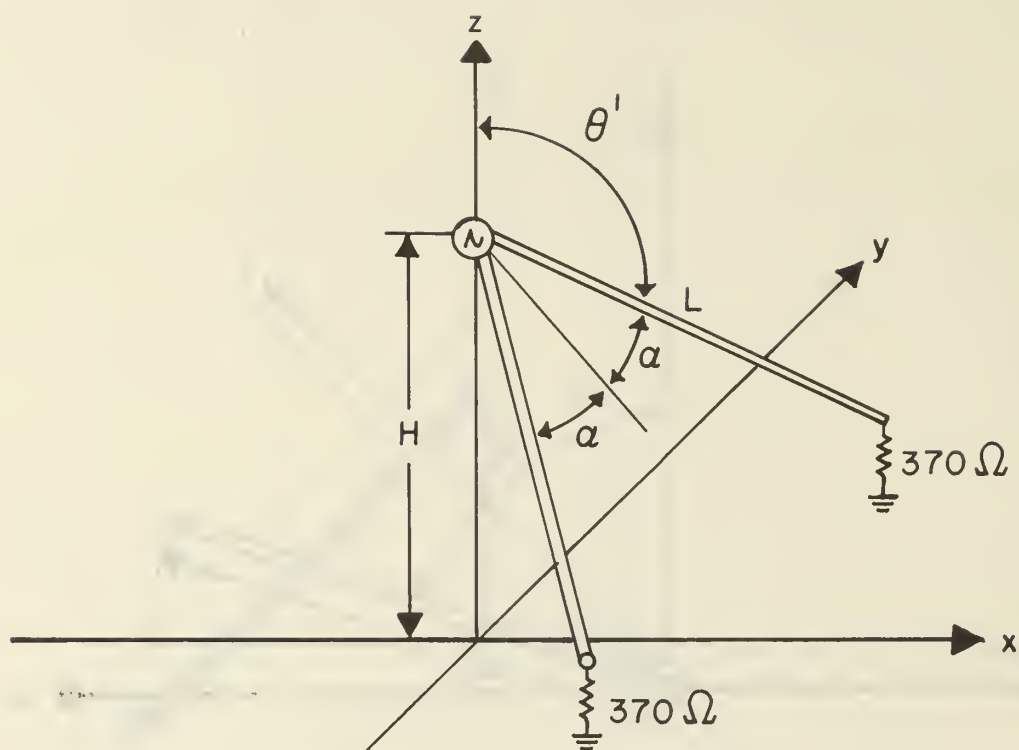


Figure C-6
Sloping Long-Wire Geometry



L - Length of sloping element

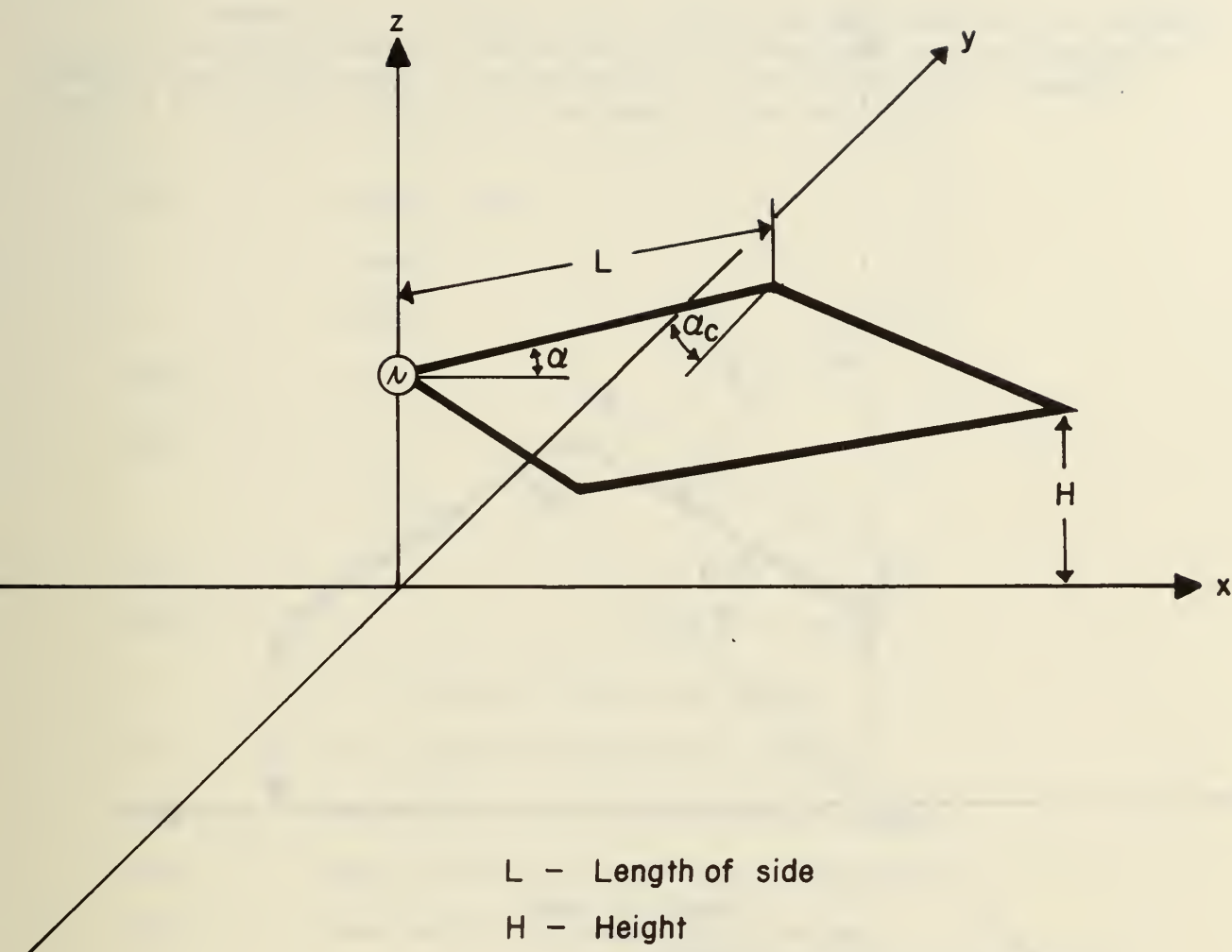
α - Half element separation

θ' - Slope angle (measured from zenith)

H - Height of load point

Figure C-7

Terminated Sloping Vee Geometry



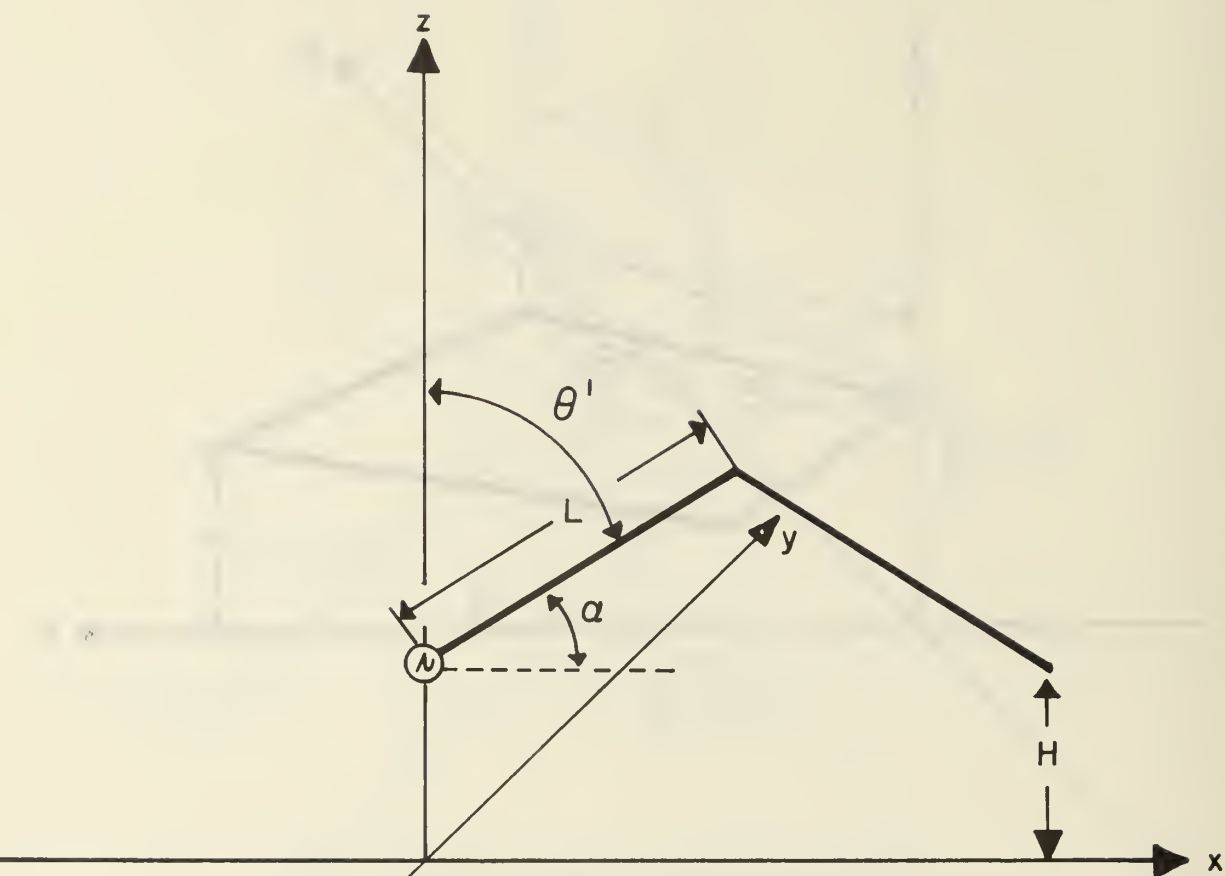
L - Length of side

H - Height

α - Half separation angle at load point

α_c - Complement of α

Figure C-8
Rhombic Geometry



L - Length of side

H - Height

θ' - Tilt angle of side (measured from zenith)

α - Half separation angle of rhombus at load point

Figure C-9
Vertical Half - Rhombic Geometry

APPENDIX D

PROGRAM LISTING

This appendix is a listing of the Fortran implementation of the antenna pattern graphics program. The listing is preceded by a partial listing of definitions of the computer variables used. Variables used in graphics display processors only were not included in the below list.

ANTN	Antenna Type
L	length
H	height
PHIPR	ϕ
THEPR	θ
F	f in mhz
EPSLN	ϵ_r
SIGMA	α
M	θ in degrees (observation zenith)
KAY	ϕ in degrees (observation azimuth)
PAR	Reinitialize and Log Gain option command
ISTRH	Save horizontal pattern option command
ISTRV	Save vertical pattern option command
ALPH	α
ALPCM	α_c
DLPRI	Δ
LMDA	λ
K	k
C2	k_2
RHPRI	R_h
RVPRI	R_v
S	S

SX	S_x
SY	S_y
SZ	S_z
YO	Y_o
ZO	Z_o
ROW	ρ
R	r
R1	r_1
R2	r_2
PI	Π
RIN	R_{in}
THETA	Θ
PHI	ϕ
KCOS	$\cos (\theta)$
RV	R_v
RH	R_h
SIGHV	ψ_v
SIGHH	ψ_h
DELTA	Δ
COSDL	$\cos (\Delta)$
SINDL	$\sin (\Delta)$
SINDP	$\sin (\Delta')$
COSDP	$\cos (\Delta')$
SINPI	$\sin (\phi)$
COSPI	$\cos (\phi)$

ETH1	E_{Θ_1}
ETH2	E_{Θ_2}
EPHI1	E_{ϕ_1}
EPHI2	E_{ϕ_2}
G	Gain (relative pattern)
GAIN	Max Gain
NORM	Max value of linear gain
EPHI	$ E_{\phi} ^2$
ETHET	$ E_{\phi} ^2$
KOS1	$\cos \psi_1$
KOS2	$\cos \psi_2$
KOS3	$\cos \psi_3$
KOS4	$\cos \psi_4$
KOS5	$\cos \psi_5$
KOS6	$\cos \psi_6$
KOS7	$\cos \psi_7$
KOS8	$\cos \psi_8$
COSU1	$\cos (\psi_1)$
COSU2	$\cos (\psi_2)$
COSU3	$\cos (\psi_3)$
COSU4	$\cos (\psi_4)$
SINU1	$\sin (\psi_1)$
SINU2	$\sin (\psi_2)$
SINU3	$\sin (\psi_3)$

SINU4	$\text{SIN } (U_4)$
SINAC	$\text{SIN } (\alpha_c)$
COSAC	$\text{COS } (\phi_c)$
IRCAL	Recall saved pattern option command
ISEA	Sea State
ICRS	Sea direction
SIGL	$\text{Log}_{10} (G (M, KAY))$
ADA	η
DPHIP	$\Delta \phi$
WAVE	wave
DLTI	$\Delta \theta_1$
DLT2	$\Delta \theta_2$
DLT3	$\Delta \theta_3$
SIND3	$\text{SIN } (\Delta \theta_3)$
SINA	$\text{SIN } (\Delta \phi)$
COSA	$\text{COS } (\Delta \phi)$
VAR	ωt (wave)
Z	Z
RGRAL	R_{ij}
XGRAL	X_{ij}
CEE	C
SRFAC	$(A_3 + jB_3)$
DLTZ1	ΔZ_1
DLTZ2	ΔZ_2
CV	$ R_v $
CH	$ R_h $

Computer variables that are identical to the terms in Appendix C they represent, are not listed here.

The following sub-programs are included in the program:

1. SUBROUTINE SINUS (X, SC)

$$\text{COMPUTER } Si(x) = - \int_x^{\infty} \frac{\sin t}{t} dt$$

2. SUBROUTINE KOSINUS (X, CC)

$$\text{COMPUTER } Ci(x) = - \int_x^{\infty} \frac{\cos t}{t} dt$$

3. FUNCTION CINC (x)

$$\text{COMPUTER } \frac{\cos x}{x}$$

4. FUNCTION SINC (x)

$$\text{COMPUTER } \frac{\sin x}{x}$$

5. FUNCTION AKEX (x)

$$\text{COMPUTER } Ei(\pm jx)$$

6. FUNCTION ADAE (x)

$$\text{COMPUTER } \eta_e$$

7. FUNCTION ZGRAL (x)

$$\text{COMPUTER integrand for } \Delta Z_2 = \int () dr$$

8. FUNCTION RESIST (s)

$$\text{COMPUTER INTEGRAND for } Ri_j = \int () ds$$

9. FUNCTION REACT (s)

$$\text{COMPUTER INTEGRAND for } Xi_j = \int () ds$$

ASSIGN 1 T0 IF0(1)
ASSIGN 2 T0 IF0(2)
ASSIGN 3 T0 IF0(3)
ASSIGN 4 T0 IF0(4)
ASSIGN 5 T0 IF0(5)
ASSIGN 6 T0 IF0(6)
ASSIGN 7 T0 IF0(7)
ASSIGN 8 T0 IF0(8)
ASSIGN 9 T0 IF0(9)
ASSIGN 10 T0 IF0(10)
ASSIGN 11 T0 IF0(11)
ASSIGN 12 T0 IF0(12)
ASSIGN 13 T0 IF0(13)
ASSIGN 14 T0 IF0(14)
ASSIGN 15 T0 IF0(15)
ASSIGN 16 T0 IF0(16)
ASSIGN 17 T0 IF0(17)
ASSIGN 18 T0 IF0(18)
ASSIGN 19 T0 IF0(19)
ASSIGN 20 T0 IF0(20)
ASSIGN 21 T0 IF0(21)
ASSIGN 22 T0 IF0(22)
ASSIGN 123 T0 IF0(23)
ASSIGN 125 T0 IF0(25)
ASSIGN 127 T0 IF0(27)
ASSIGN 129 T0 IF0(29)
ASSIGN 131 T0 IF0(31)
ASSIGN 133 T0 IF0(33)
ASSIGN 135 T0 IF0(35)
ASSIGN 137 T0 IF0(37)
ASSIGN 139 T0 IF0(39)
ASSIGN 141 T0 IF0(41)
ASSIGN 143 T0 IF0(43)
ASSIGN 145 T0 IF0(45)
ASSIGN 147 T0 IF0(47)


```

*00100 2 1 IF(23)
*00110 2 1 IF(20)
*00120 2 1 IF(32)
*00130 2 1 IF(24)
*00140 2 1 IF(26)
*00150 2 1 IF(28)
*00160 2 1 IF(42)
*00170 2 1 IF(42)
*00180 2 1 IF(42)
*00190 2 1 IF(44)
C- 50 I=1,44
C- CODE(C(I), IF9(I), IPAR(I))
170 CONTINUE
C- 114 I=1,50
114 I(I)=C
C- 111 I=1,50
111 Y1(I)=Y1(I)=0.0
C- 112 I=1,90
112 Y2(I)=Y2(I)=0.0
C- 113 I=1,360
113 Y3(I)=Y3(I)=0.0
C- CONTINUE
C- 51 I=1,43,2
CALL TEXTR(IDEV, IPAR(I), NW(I), LN(I), IP(I), 1, 3, IER)
11 IF(IER.NE.0) OUTPUT(101) IFR, 'TBLK', I
C- PARAFTEK AND OPTIONS COMMAND INPUT PROCESSOR
C- 52 I=2,44,2
CALL TEXTR(IDEV, IPAR(I), NW(I), LN(I), IP(I), 1, 3, IER)
11 IF(IER.NE.0) OUTPUT(101) IFR, 'TBLK', I
C- 22+1/2
C- IF(ND(I)IK(J), 2) .EQ.0) G3 TO 53
CALL TEXTR(IDEV, IPAR(I), NW(I), LN(I), IP(I), IER)
C- IF(IER.NE.0) OUTPUT(101) IFR, 'IPAR', I
1 FORMAT('A11')
2 FORMAT('LE, 9')
3 FORMAT('L5, 1')

```

```

7  FORMAT('PHIP')
9  FORMAT('THEP')
11 FORMAT('FREQ')
13 FORMAT('EPSL')
15 FORMAT('SGMA')
17 FORMAT('PHI ')
19 FORMAT('THET')
21 FORMAT('PARM')
2  FORMAT(' ')
4  FORMAT(' ')
6  FORMAT(' ')
8  FORMAT(' ')
10 FORMAT(' ')
12 FORMAT(' ')
14 FORMAT(' ')
16 FORMAT(' ')
18 FORMAT(' ')
20 FORMAT(' ')
22 FORMAT(' ')
123 FORMAT('ISTH')
125 FORMAT('ISTV')
127 FORMAT('IRCL')
129 FORMAT('HGTT')
131 FORMAT('ALPH')
133 FORMAT('GAIN')
135 FORMAT('ISEA')
137 FORMAT('ICRS')
139 FORMAT('SIGL')
141 FORMAT(' ')
143 FORMAT(' ')
95  FORMAT(F4.0)
96  FORMAT(F4.1)
97  FORMAT(F4.2)
98  FORMAT(F4.3)
101 FORMAT(I4)

```

```

DEC0DE(4,101,IPAR(2))ANTN
DEC0DE(4,96,IPAR(4))L
DEC0DE(4,96,IPAR(6))H
DEC0DE(4,101,IPAR(8))ITEM
PHIPR=ITEM
DEC0DE(4,101,IPAR(10))ITEM
THEPR=ITEM
DEC0DE(4,95,IPAR(12))F
DEC0DE(4,96,IPAR(14))EPSLN
DEC0DE(4,97,IPAR(16))SIGMA
DEC0DE(4,101,IPAR(18))M
DEC0DE(4,101,IPAR(20))KAY
DEC0DE(4,101,IPAR(22))PAR
DEC0DE(4,101,IPAR(24))ISTRH
DEC0DE(4,101,IPAR(26))ISTRV
DEC0DE(4,101,IPAR(28))IRCAL
DEC0DE(4,96,IPAR(30))HT
DEC0DE(4,101,IPAR(32))ITEM
ALPH=ITEM
DEC0DE(4,101,IPAR(36))ISEA
DEC0DE(4,101,IPAR(38))ICRS
WRITE(6,3300)ISEA,ICRS
3300 FORMAT(I4,5X,I4)
IF(HT.EQ.(1.))SIGMA=SIGMA*.1
IF(HT.EQ.(2.))SIGMA=SIGMA*.01
IF(PAR.EQ.1)G0 T0 170
IF(HT.GT.75.0)L=L+100.0
IF(HT.GT.85.0)L=L+200.0
CALL DGINIT(IDEV,IGDIR,ISIZE,IER)
WRITE(6,103)M,KAY,PAR,PHIPR,THEPR,L,H,F,EPSLN,SIGMA
103 FORMAT(3I5,7F12.8)
IF(PAR.EQ.1)G0 T0 170
C PATTERN MANUAL ENTRY PROCESSOR
D0 151 I=1,50
151 IMD(I)=IM(I)

```

```

ITRY(1)=IHEAD(0,10)
ITRY(2)=IPACK(0,6,0)
ITRY(3)=IPACK(0,4,1)
ITRY(4)=IPACK(-.1,.5,0)
ITRY(5)=IPACK(.1,.5,1)
ITRY(6)=IPACK(0,.6,0)
ITRY(7)=IPACK(0,.4,1)
ITRY(8)=IPACK(-.1,.5,0)
ITRY(9)=IPACK(.1,.5,1)
ITRY(10)=IPACK(0,0,0)
D0 153 I=11,50
J=I-1
153 ITRY(I)=IPACK(X1(I),Y1(I),IMD(I))
CALL GRAPHR(IDEV,ITRY,50,1,IER)
IF(IER.NE.0)OUTPUT(101) IER,'GBLK1'
154 IF(MOD(IGDIR(1),8).EQ.0)GO TO 154
CALL GRAPHI(IDEV,ITRY,1,IER)
IF(IER.NE.0)OUTPUT(101)IER,'IGBLK'
D0 155 I=1,50
CALL UNPACK(ITRY(I),X1(I),Y1(I),IMD(I))
155 IM(I)=IMD(I)
C ENVIRONMENTAL CONSTANTS PR0CESS0R
PHIPR=PHIPR*(3.14159265/180)
THEPR=THEPR*(3.14159265/180)
ALPH=(3.14159265/180)*ALPH
ALPCM=(3.14159265/2.0)*ALPH
DLPRI=(3.14159265/2.0)-THEPR
F=F*1.0E 06
ABMEG=2*3.14159265*F
ADA1=CMPLX(0,0,1.26E-06*ABMEG)
ADA2=CMPLX(SIGMA,ABMEG*EPSLN*8.854E-12)
ADA=(ADA1/ADA2)*0.5
TEMP1=REAL(ADA)
TEMP2=AIMAG(ADA)
WRITE(6,2400)TEMP1,TEMP2

```

```

2400 FORMAT('ADA=',2F12.6)
LMDA=3.0E08/F
K=6.28318530/LMDA
F=F*1.0E-06
C2=K*CMPLX(EP SLN,-1.8E04*SIGMA/F)**0.5
RHPRI=(K-C2)/(K+C2)
HTEMP=H
THTEM=THEPR
RVPRI=(C2-K)/(C2+K)
JJJ=0
II=0
DPHIP=0.0
IF(ISEA.GT.0)G0 T0 3000
3010 CONTINUE
C INPUT RESISTANCE PROCESSOR
IF(ANTN.EQ.1)G0 T0 1100
IF(ANTN.EQ.2)G0 T0 1200
IF(ANTN.EQ.3)G0 T0 1300
IF(ANTN.EQ.4)G0 T0 1400
IF(ANTN.EQ.5)G0 T0 1500
IF(ANTN.EQ.6)G0 T0 1600
IF(ANTN.EQ.7)G0 T0 1700
IF(ANTN.EQ.8)G0 T0 1800
2000 CONTINUE
IF(ISEA.GT.0)G0 T0 3013
WRITE(6,104)RIN
3013 CONTINUE
104 FORMAT(F12.4)
C OBSERVATION ANGLE CONSTANTS PROCESSOR
D0 42 N=1,2
IF(N.EQ.1) G0 T0 71
IF(N.EQ.2) G0 T0 72
71 D0 42 I=1,90
J=M
E=I

```

```

60 T0 73
72 D0 42 J=1,360
I=KAY
E=J
60 T0 73
73 CONTINUE
THETA=I*(3.14159265/180)
PHI=J*(3.14159265/180)
K0S=C0S(THETA)*C0S(THPR)+SIN(THETA)*SIN(THPR)*C0S(PHI*PHIPR)
SINSQ=1-(K0S**2)
WOSQ=(3.14159265/180)**2
IF(SINSQ.LT.WOSQ)SINSQ=WOSQ
FIFA=(1-((K/C2)*SIN(THETA))**2)**0.5
K0S=C0S(THETA)
RV=(K0S-(K/C2)*FIFA)/(K0S+(K/C2)*FIFA)
RH=(K0S-(C2/K)*FIFA)/(K0S+(C2/K)*FIFA)
VR=REAL(RV)
VI=AIMAG(RV)
SIGHV=ATAN2(VI,VR)
HR=REAL(RH)
HI=AIMAG(RH)
SIGHH=ATAN2(HI,HR)
GAIN PROCESSOR
IF(ANTN.EQ.1)G0 T0 100
IF(ANTN.EQ.2)G0 T0 200
IF(ANTN.EQ.3)G0 T0 300
IF(ANTN.EQ.4)G0 T0 400
IF(ANTN.EQ.5)G0 T0 500
IF(ANTN.EQ.6)G0 T0 600
IF(ANTN.EQ.7)G0 T0 700
IF(ANTN.EQ.8)G0 T0 800
42 CONTINUE
C NORMALIZE AND MAX GAIN PROCESSOR
N=1
D0 43 J=1,2

```



```

IF(J.EQ.1) GO TO 75
IF(J.EQ.2) GO TO 76
75 DO 43 I=1,90
   GO TO 77
76 DO 43 I=1,360
   GO TO 77
77 CONTINUE
FAC(N)=G(J,I)
43 N=N+1
   NORM=0.0
   DO 46 I=1,450
   NORM=AMAX1(NORM,FAC(I))
   GAIN=NORM/RIN
   ATEMP=ALOG10(GAIN)
   ENCODE(4,96,IPAR(34))ATEMP
   CALL TEXT0(IDEV,IPAR(34),1,34,1,1,3,IER)
   IF(IER.NE.0)OUTPUT(101)IER,'GAIN'
   IF(ISEA.GT.0)GO TO 3020
   GAIN=10.*ATEMP
   IF(PAR.EQ.2)GO TO 171
181 CONTINUE
   WRITE(6,106)NORM,GAIN
106 FORMAT(2F12.6)
3021 CONTINUE
C
PATTERN DISPLAY PROCESSOR
DO 44 I=1,360
PHI=I*(3.14159265/180)
G(2,I)=G(2,I)/(NORM*2.0)
X(I)=G(2,I)*COS(PHI)
44 Y(I)=G(2,I)*SIN(PHI)+0.5
   IMD(1)=0
DO 45 I=2,360
45 IMD(I)=1
   PATRN(1)=IHEAD(0,10)
DO 47 I=2,361

```

```

J=I-1
47 PATRN(I)=IPACK(X(J),Y(J),IMD(J))
   PATRN(362)=0
   IF(ISEA.GT.C)G0 T0 3030
   CALL GRAPHR(IDEV,PATRN,362,2,IER)
   IF(IER.NE.O)OUTPUT(101)IER,'GBLK'
60 IF(M0D(IGDIR(2),8).EQ.O)G0 T0 60
3011 CONTINUE
   IF(ISTRH.EQ.1) G0 T0 156
158 CONTINUE
C   DISPLAY VERT PATTERN AT REQUESTED PHI
   D0 49 I=1,90
   THETA=I*(3.14159265/180)
   G(1,I)=G(1,I)/(N0RM*2.O)
   X(I)=G(1,I)*SIN(THETA)
49 Y(I)=G(1,I)*COS(THETA)-0.5
   IMD(1)=0
   D0 61 I=2,90
61 IMD(I)=1
   VPAT(1)=IHEAD(0,10)
   D0 62 I=2,91
   J=I-1
62 VPAT(I)=IPACK(X(J),Y(J),IMD(J))
   VPAT(92)=0
   IF(ISEA.GT.O)G0 T0 3040
   CALL GRAPHR(IDEV,VPAT,92,3,IER)
   IF(IER.NE.O) OUTPUT(101)IER,'GBLK2'
64 IF(M0D(IGDIR(3),8).EQ.O)G0 T0 64
3012 CONTINUE
   IF(ISTRV.EQ.1)G0 T0 159
161 CONTINUE
   IF(IRCAL.EQ.1)G0 T0 162
   IF(ISEA.GT.O)G0 T0 3000
   G0 T0 33
C   PATTERN SAVE PR0CESS0R

```



```

156 CALL GRAPHI(IDEV,PATRN,2,IER)
   IF(IER.NE.0)OUTPUT(101)IER,'GBLK2'
D0 157 I=1,360
157 CALL UNPACK(PATRN(I+1),X3(I),Y3(I),IMD(I))
   G0 T0 158

159 CALL GRAPHI(IDEV,VPAT,3,IER)
   IF(IER.NE.0)OUTPUT(101)IER,'GBLK3'
D0 160 I=1,90
160 CALL UNPACK(VPAT(I+1),X2(I),Y2(I),IMD(I))
   G0 T0 161
C   DISPLAY SAVED PATTERNS PR0CESS0R
162 IMD(1)=0
   ISAVH(1)=IHEAD(0,10)
D0 163 I=2,360
163 IMD(I)=1
   D0 164 I=2,361
   J=I-1
164 ISAVH(I)=IPACK(X3(J),Y3(J),IMD(J))
   ISAVH(362)=0
   CALL GRAPHR(IDEV,ISAVH,362,4,IER)
   IF(IER.NE.0)OUTPUT(101)IER,'GBLK4'
166 IF(M0D(IGDIR(4),8).EQ.0)G0 T0 166
   ISAVV(1)=IHEAD(0,10)
D0 167 I=2,91
   J=I-1
167 ISAVV(I)=IPACK(X2(J),Y2(J),IMD(J))
   ISAVV(92)=0
   CALL GRAPHR(IDEV,ISAVV,92,5,IER)
   IF(IER.NE.0)OUTPUT(101)IER,'GBLK5'
169 IF(M0D(IGDIR(5),8).EQ.0)G0 T0 169
   G0 T0 33
C   LOG GAIN PR0CESS0R
171 BLIM=.001
D0 172 I=1,90
   TEMP=G(1,I)/N0RM

```

```

IF(TEMP.LT.BLIM)TEMP=BLIM
G(1,I)=ALOG10(TEMP)+3.0
172 CONTINUE
DO 173 I=1,360
TEMP=G(2,I)/NORM
IF(TEMP.LT.BLIM)TEMP=BLIM
G(2,I)=ALOG10(TEMP)+3.0
173 CONTINUE
190 FORMAT(2F12.8,I5)
NORM=3.0
GO TO 181
3000 II=II+1
II=MOD(II,36)
PI=3.14159265
D2R=PI/180.0
VAR=(PI/18.0)*II
WAVE=(ISEA*8*SIN(VAR))*D2R
VAR1=ICRS*D2R
DLT1=WAVE*SIN(VAR1)
DLT2=WAVE*COS(VAR1)*0.3
IF(ANTN.EQ.5)GO TO 3500
H=HTEMP*COS(DLT1)*COS(DLT2)
THEPR=THTEM-DLT1
DLPRI=PI/2.-THEPR
GO TO 3090
3500 AA=2*L*SIN(DLT1/2.)
BB=2*L*SIN(DLT2/2.)
CC=SQRT(AA**2+BB**2)
DD=SQRT(L**2-(CC/2.）**2)
DLT3=2.*ATAN2((CC/2.),DD)
SIND3=SIN(DLT3)
IF((SIND3.LT.WOSQ).AND.(SIND3.GE.0.0))SIND3=WOSQ
IF((SIND3.GT.-WOSQ).AND.(SIND3.LT.0.0))SIND3=-WOSQ
SINA=SIN(DLT1)/SIND3
SINA=ABS(SINA)

```

```

C9SA=SGRT(1.0-SINA**2)
DPHIP=ATAN2(SINA,C9SA)
IF((DLT1.LT.0.0).AND.(DLT2.GE.0.0))DPHIP=-DPHIP
IF((DLT1.LT.0.0).AND.(DLT2.LT.0.0))DPHIP=-(PI-DPHIP)
IF((DLT1.GT.0.0).AND.(DLT2.LT.0.0))DPHIP=PI-DPHIP
THEPR=THTEM+DLT3
DLPRI=PI/2.-THEPR
3090  JJJ=JJJ+1
      IF(JJJ,EQ.37)GO TO 33
      WRITE(6,3301)DLT1,DLT2,DLT3,WAVE,H,II
3301  FORMAT(5F12.6,I4)
      GO TO 3010
3020  TEMP=G(1,KAY)/RIN
      SIGL=ALOG10(TEMP)
      ENCODE(4,96,IPAR(40))SIGL
      CALL TEXT0(IDEV,IPAR(40),1,40,1,1,3,IER)
      IF(IER.NE.0)OUTPUT(101)IER,'SIGL'
      SIG=10*SIGL
      WRITE(6,3022)SIG,RIN
3022  FORMAT(F12.6,5X,F12.6)
      GO TO 3021
3030  CALL GRAPH0(IDEV,PATRN,362,2,IER)
      IF(IER.NE.0)OUTPUT(101)IER,'HPAT'
      GO TO 3011
3040  CALL GRAPH0(IDEV,VPAT,92,3,IER)
      IF(IER.NE.0)OUTPUT(101)IER,'VPAT'
      GO TO 3012
C      ARBITRARILY TILTED DIP0LE
      DELTA=(3.14159265/2.-THETA)
      S1=C9S(SIGHH-2*K*H*SIN(DELTA))
      S2=SIN(SIGHH-2*K*H*SIN(DELTA))
      S3=C9S(SIGHV-2*K*H*SIN(DELTA))
      S4=SIN(SIGHV-2*K*H*SIN(DELTA))
      CV=CABS(RV)
      CH=CABS(RH)

```

```

C0SDL=C0S(DELTA)
SINDL=SIN(DELTA)
SINDP=SIN(DLPRI)
C0SDP=C0S(DLPRI)
SINPI=SIN(PHI-PHI(PIR))
C0SPI=C0S(PHI-PHI(PIR))
FCT=1.0-(-SINDL*SINDP+C0SDL*C0SDP*SINPI)**2
FCTR=1.0-(-SINDP*SINDL+C0SDL*C0SDP*SINPI)**2
GI=(C0S(0.5*K*L)*(SINDL*SINDP+C0SDL*C0SDP*SINPI))-C0S(0.5*K*L))/FCT
DI=(C0S(0.5*K*L)*(C0SDL*C0SDP*SINPI-SINDL*SINDP))-C0S(0.5*K*L))/
1FCTR
ETHT1=(C0SDP*SINPI*SINDL-SINDP*C0SDL)*GI-(C0SDP*SINPI*SINDL+
1SINDP*C0SDL)*DI*CV*S3
EPHI1=C0SDP*C0SPI*(GI+DI*CH*S1)
ETHT2=(C0SDP*SINPI*SINDL+SINDP*C0SDL)*DI*CV*S4
EPHI2=C0SDP*C0SPI*DI*CH*S2
IF(FCTR.LT.W0SQ)ETHT1=EPHI1*0.0
IF(FCTR.LT.W0SQ)ETHT2=EPHI2*0.0
G(N,E)=120.0*(ETHT1**2+ETHT2**2+EPHI1**2+EPHI2**2)
GO TO 42

1100 SAVIT=DLPRI
DO 1110 I=1,2
YO=.00001
ZO=0.0
IF(I.EQ.1)DLPRI=0.0
IF(I.EQ.2)DLPRI=SAVIT
IF(I.EQ.2)YO=2.0*H*C0S(DLPRI)/LMDA
S=-0.5*L/LMDA
RGRAL=RESIST(S)/2
DS=L/(LMDA*100)
DO 1115 N=2,100
S=S+DS
1115 RGRAL=RESIST(S)+RGRAL
RGRAL=-30.0*(RGRAL+(RESIST(0.5*L/LMDA))/2.0)*DS
S=-0.5*L/LMDA

```

```

XGRAL=REACT(S)/2.
D9 1116 N=2,10C
S=S+DS
1116 XGRAL=XGRAL+REACT(S)
XGRAL=-30.0*(XGRAL+(REACT(0.5*L/LMDA))/2)*DS
1110 Z(I)=CMPLX(RGRAL,XGRAL)
AJ=CMPLX(0,1)
ENE=C9S(DLPRI)
TW9=-SIN(DLPRI)
CEE=(RHPRI*C9S(DLPRI)+AJ*RVPRI*SIN(DLPRI))*CMPLX(ONE,TW9)
RIN=REAL(Z(1))+REAL(Z(2)*CEE)
G9 T9 2000
1500 CALL SINUS((2*K*L),SC)
SI1=-SC
CALL K9SINUS((2*K*L),CC)
CI1=CC
CALL K9SINUS((4*K*L),CC)
CI2=CC
CALL SINUS((4*K*L),SC)
SI2=-SC
RIN=30.0*(0.5*(AL9G(K*L)+0.577-CI2)+.693+C9S(K*L))*(C9S(K*L)*
1 (AL9G(K*L)+.577-2*CI1+CI2)-SIN(K*L))*(SI2-2.*SI1))
G9 T9 2000
C VERTICAL MONOPOLE
C VERTICAL MONOPOLE GAIN
200 DELTA=3.14159264/2.-THETA
SINDL=SIN(DELTA)
C9SDL=C9S(DELTA)
CV=CABS(RV)
S3=C9S(SIGHV)
S4=SIN(SIGHV)
A=C9S(K*L*SINDL)-C9S(K*L)
B=SIN(K*L*SINDL)-SINDL*SIN(K*L)
G(N,E)=(30.0/C9SDL**2)*((A*(1.+CV*S3)+B*CV*S4)**2+
1 (B*(1.-CV*S3)+A*CV*S4)**2)

```

```

C 1200 G8 T8 42
      VERTICAL MONOPOLE
      CALL K0SINUS((4*K*L),CC)
      CIN2=ALOG(4*K*L)+.577-CC
      CALL K0SINUS((2*K*L),CC)
      CIN1=ALOG(2*K*L)+.577-CC
      CALL SINUS((4*K*L),SC)
      SIN2=1.57078633+SC
      CALL SINUS((2*K*L),SC)
      SIN1=1.57078633+SC
      RIN=15.*((2.+2*C0S(2*K*L))*CIN1-C0S(2*K*L)*CIN2-2*SIN(2*K*L)*SIN1+
      1SIN(2*K*L)*SIN2)
      G8 T8 2000
      VERTICAL MONOPOLE WITH GROUND SCREEN
      DELTA=3.14159265/2.-THETA
      IF((N.EQ.2).AND.(J.GT.1))G8 T8 310
      SINDL=SIN(DELTA)
      C0SDL=C0S(DELTA)
      CV=CABS(RV)
      S3=C0S(SIGHV)
      S4=SIN(SIGHV)
      A=C0S(K*L*SINDL)-C0S(K*L)
      B=SIN(K*L*SINDL)-SINDL*SIN(K*L)
      C1=SIN(K*L)
      IF((C1.LT.WOSQ).AND.(C1.GE.0.0))C1=WOSQ
      IF((C1.GT.-WOSQ).AND.(C1.LT.0.0))C1=-WOSQ
      C3=A
      IF((C3.LT.WOSQ).AND.(C3.GE.0.0))C3=WOSQ
      IF((C3.GT.-WOSQ).AND.(C3.LT.0.0))C3=-WOSQ
      XB=K*H
      DX=XB/100
      XX=C
      GRAL=PTGRL(XX)/2
      D8 315 II=2,100
      XX=XX+DX

```



```

315 GRAL=GRAL+PTGRL(XX)
   GRAL=(CRAL+PTGRL(XB)/2)*DX
   GRAL=1.0-(ADA*SIN(THETA)*GRAL)/120.*3.14159265*C1*C3
   SRFAC=(CABS(GRAL))*2
   WRITE(6,311)SRFAC
311 FORMAT('SRFAC=',F12.6)
310 CONTINUE
   G(N,E)=(30.0/C8SDL**2)*((A*(1.+CV*S3)+B*CV*S4)**2+
1(B*(1.-CV*S3)+A*CV*S4)**2)*SRFAC/C1**2
   GO TO 42
C   GROUND SCREEN
1300 C1=SIN(K*L)**2
   IF(C1.LT.WOSQ)C1=WOSQ
   RO=(H**2+L**2)**0.5
   R1=H+RO
   ARGP=CMPLX(0.0,K*L)
   ARCM=CMPLX(0.0,-K*L)
   ARGP2=CMPLX(0.0,2*K*L)
   ARGM2=CMPLX(0.0,-2*K*L)
   DLTZ1=(ADA/4*3.14159265*C1)*(CEXP(ARGP2)*AKEX(-2*K*(RO+L))+
1CEXP(ARGM2)*AKEX(-2*K*(RO-L))+2*C8S(K*L)**2*AKEX(-2*K*H)+
14*C8S(K*L)*AKEX(-K*R1)-4*C8S(K*L)*CEXP(ARGM)*AKEX(-K*(R1-L))-
14*C8S(K*L)*CEXP(ARGP)*AKEX(-K*(R1+L)))
   WRITE(6,1311)DLTZ1
1311 FORMAT('DLTZ1=',F12.6)
      NN=120
      WIRE=.01
      DX=(H-.01)/100
      DUM=.01
      DLTZ2=ZGRAL(DUM)/2.
      DB 1310 II=2,100
      DUM=DUM+DX
1310 DLTZ2=DLTZ2+ZGRAL(DUM)
      DLTZ2=(DLTZ2+ZGRAL(H)/2.)*DX
      DLTZ2=-DLTZ2

```

```

WRITE(6,1312)DLTZ2
1312 FORMAT('DLTZ2=',F12.6)
CALL K0SINUS((4*K*L),CC)
CIN2=AL0G(4*K*L)+.577-CC
CALL K0SINUS((2*K*L),CC)
CIN1=AL0G(2*K*L)+.577-CC
CALL SINUS((4*K*L),SC)
SIN2=1.57078633+SC
CALL SINUS((2*K*L),SC)
SIN1=1.57078633+SC
RIN=15.*((2.+2*C0S(2*K*L))*CIN1-C0S(2*K*L)*CIN2-2*SIN(2*K*L)*SIN1
1+SIN(2*K*L)*SIN2)
RIN=RIN+REAL(DLTZ1+DLTZ2)
GO TO 2000
C INVERTED L
400 DELTA=(3.14159265/2.)-THETA
SINDL=SIN(DELTA)
C0SDL=C0S(DELTA)
CV=CABS(RV)
CH=CABS(RH)
DENM1=1.0-C0SDL**2*SIN(PHI)**2
S1=C0S(SIGHH-2*K*H*SINDL)
S2=SIN(SIGHH-2*K*H*SINDL)
S3=C0S(SIGHV-2*K*H*SINDL)
S4=SIN(SIGHV-2*K*H*SINDL)
A=C0S(K*L)*C0S(K*H*SINDL)-SINDL*SIN(K*L)*SIN(K*H*SINDL)
1-C0S(K*(H+L))
B=SINDL*SIN(K*L)*C0S(K*H*SINDL)+C0S(K*L)*SIN(K*H*SINDL)
1-SINDL*SIN(K*(H+L))
GI=SIN(K*L*C0SDL*SIN(PHI))-C0SDL*C0S(PHI)*SIN(K*L)
GR=C0S(K*L*C0SDL*SIN(PHI))-C0S(K*L)
ETHET=((SIN(PHI)*SINDL*(GR*(1.0-CV*S3)+GI*CV*S4)/DENM1)
1-(A*(1.0+CV*C0S(SIGHV))+B*CV*SIN(SIGHV))/C0SDL)**2
1+((SIN(PHI)*SINDL*(GI*(1.0-CV*S3)-GR*CV*S4)/DENM1)
1-(B*(1.0-CV*C0S(SIGHV))+A*CV*SIN(SIGHV))/C0SDL)**2

```



```

EPHI=(COS(PHI)/DENM1)**2*((GR*(1.0+CH*S1)-GI*CH*S2)**2
1+(GI*(1.0+CH*S1)+GR*CH*S2)**2)
C(N,E)=30.0*(ETHET+EPHI)
GO T9 42
INVERTED L
1400 CALL KOSINUS((2*K*H),CC)
CI1=CC
CALL KOSINUS((4*K*H),CC)
CI2=CC
CALL SINUS((2*K*H),SC)
SI1=-SC
CALL SINUS((4*K*H),SC)
SI2=-SC
RIN=60.*(1.41+ALOG(2*L/LMDA))+SINC(2*K*L))+30.*(-0.5*COS(2*K*H)*
1(ALOG(2*K*H)+1.270+CI2)+(1.0+COS(2*K*H))*(ALOG(2*K*H)+0.577-CI1)-
1SIN(2*K*H)*(0.5*SI2-SI1))
GO T9 2000
SLAPING LONG WIRE
SLAPE LONGWIRE
500 DELTA=3.14159265/2.0-THETA
PHI=PHI-DPHIP
C9SDL=COS(DELTA)
SINDL=SIN(DELTA)
C8SDP=COS(DLPRI)
SINDP=SIN(DLPRI)
C8SPI=COS(PHI)
SINPI=SIN(PHI)
FCT1=1.0-(SINDL*SINDP+C8SDL*C8SDP*C8SPI)**2
FCT2=1.0-(C8SDL*C8SDP*C8SPI-SINDL*SINDP)**2
CIG=(C8S(K*L)*(SINDL*SINDP+C8SDL*C8SDP*C8SPI))-C8S(K*L))/FCT1
SIG=(SIN(K*L)*(SINDL*SINDP+C8SDL*C8SDP*C8SPI))-
1(SINDL*SINDP+C8SDL*C8SDP*C8SPI)*SIN(K*L))/FCT1
CIGP=(C8S(K*L)*(C8SDL*C8SDP*C8SPI-SINDL*SINDP))-C8S(K*L))/FCT2
SIGP=(SIN(K*L)*(C8SDL*C8SDP*C8SPI-SINDL*SINDP))+
1(SINDL*SINDP-C8SDL*C8SDP*C8SPI)*SIN(K*L))/FCT2

```

```

CH=CABS(RH)
CV=CABS(RV)
EPHI1=-C0SDP*SINPI*(CIG+CH*(CIGP*C0S(SIGHH)-SIGP*SIN(SIGHH)))
EPHI2=-C0SDP*SINPI*(SIG+CH*(CIGP*SIN(SIGHH)+SIGP*C0S(SIGHH)))
ETH1=CIG*(C0SDP*C0SPI*SINDL-SINDP*C0SDL)-CV*(C0SDP*C0SPI*SINDL+
1SINDP*C0SDL)*(CIGP*C0S(SIGHV)-SIGP*SIN(SIGHV))
ETH2=SIG*(C0SDP*C0SPI*SINDL-SINDP*C0SDL)-CV*(C0SDP*C0SPI*SINDL+
1SINDP*C0SDL)*(CIGP*SIN(SIGHV)+SIGP*C0S(SIGHV))
IF(FCT1*LT*WOSQ)ETH1=ETH2*EPHI1=EPHI2=0.0
G(N,E)=30.0*(EPHI1**2+EPHI2**2+ETH1**2+ETH2**2)
IF((FCT1*LT*WOSQ).AND.(FCT2*LT*WOSQ))G(N,E)=0.1
GO TO 42
C SLAPING VEE
600 DELTA=(3.14159265/2.0)-THETA
SINDL=SIN(DELTA)
SINDP=SIN(DLPRI)
C0SDL=C0S(DELTA)
C0SDP=C0S(DLPRI)
ADJ=C0S(ALPH)*C0SDP
0PP=SIN(ALPH)
ALPH=ATAN2(0PP,ADJ)
C0SP=C0S(PHI+ALPH)
C0SM=C0S(PHI-ALPH)
K0S1=SINDL*SINDP+C0SDL*C0SDP*C0SM
K0S2=SINDL*SINDP+C0SDL*C0SDP*C0SP
K0S3=-SINDL*SINDP+C0SDL*C0SDP*C0SM
K0S4=-SINDL*SINDP+C0SDL*C0SDP*C0SP
K0S5=C0SDL*SINDP+SINDL*C0SDP*C0SM
K0S6=C0SDL*SINDP+SINDL*C0SDP*C0SP
K0S7=-C0SDL*SINDP+SINDL*C0SDP*C0SM
K0S8=-C0SDL*SINDP+SINDL*C0SDP*C0SP
U1=K*L*(1.0-K0S1)
U2=K*L*(1.0-K0S2)
U3=K*L*(1.0-K0S3)
U4=K*L*(1.0-K0S4)

```

```

S1=COS(SIGHH-2*K*H*SINDL)
S2=SIN(SIGHH-2*K*H*SINDL)
S3=COS(SIGHV-2*K*H*SINDL)
S4=SIN(SIGHV-2*K*H*SINDL)
COSU1=COS(U1)
COSU2=COS(U2)
COSU3=COS(U3)
COSU4=COS(U4)
SINU1=SIN(U1)
SINU2=SIN(U2)
SINU3=SIN(U3)
SINU4=SIN(U4)
A=(K0S7*(COSU1-1.)/U1-K0S8*(COSU2-1.)/U2)+CABS(RV)*((K0S6*(
1(COSU4-1.)*S3+SINU4*S4)/U4)-K0S5*((COSU3-1.)*S3+SINU3*S4)/U3)
B=(K0S8*SINU2/U2-K0S7*SINU1/U1)+CABS(RV)*((K0S5*(SINU3*S3-(COSU3-1.
1)*S4)/U3+K0S6*((COSU4-1.)*S4-SINU4*S3)/U4)
C=SIN(PHI+ALPH)*((COSU2-1.)/U2-SIN(PHI-ALPH)*(COSU1-1.)/U1
1+CABS(RH)*((SIN(PHI+ALPH)*(COSU4-1.)/U4-SIN(PHI-ALPH)*(COSU3-1.
1/U3)*S1-(SIN(PHI-ALPH)*SINU3/U3-SIN(PHI+ALPH)*SINU4/U4)*S2)
D=SIN(PHI-ALPH)*SINU1/U1-SIN(PHI+ALPH)*SINU2/U2+CABS(RH)*((SIN(PHI
1-ALPH)*SINU3/U3-SIN(PHI+ALPH)*SINU4/U4)*S1+(SIN(PHI+ALPH)*
1(COSU4-1.)/U4-SIN(PHI-ALPH)*(COSU3-1.)/U3)*S2)
G(N,E)=0.05*(A**2+B**2+C0SDP**2*(C**2+D**2))
G0 T0 42
C TILTED INVERTED V
1600 RIN=1.0
G0 T0 2000
C HORIZONTAL RHOMBIC
700 DELTA=(3.14159265/2.0)-THETA
SINDL=SIN(DELTA)
C0SDL=COS(DELTA)
SINAC=SIN(ALPCM)
C0SAC=COS(ALPCM)
C0SPI=COS(PHI)
SINPI=SIN(PHI)

```

```

S1=COS(SIGHH-2*K*H*SINDL)
S3=COS(SIGHV-2*K*H*SINDL)
U1=1.0-COSDL*(SINAC*COSPI+COSAC*SINPI)
U2=1.0-COSDL*(SINAC*COSPI-COSAC*SINPI)
G(N,E)=2.16*((COSAC*SIN(K*0.5*L*U1))*SIN(K*0.5*L*U2)/(U1*U2))**2)*
1((COSP1-SINAC*COSDL)**2)*((CABS(RH))**2+1.0+2.0*(CABS(RH))*S1)+
1(SINDL**2)*(SINPI**2)*((CABS(RV))**2+1.0-2.0*(CABS(RV))*S3))
68 TO 42
C HORIZONTAL RHOMBIC
1700 RIN=1.0
68 TO 2000
C VERTICAL HALF RHOMBIC
800 DELTA=(3.14159265/2.0)-THETA
COSDL=COS(DELTA)
SINDL=SIN(DELTA)
COSAC=COS(ALPCM)
SINAC=SIN(ALPCM)
COSP1=COS(PHI)
SINPI=SIN(PHI)
FACK1=1.0-COSDL*COSAC*COSPI-SINDL*SINAC
FACK2=1.0-COSDL*COSAC*COSPI+SINDL*SINAC
UU1=COS(SIGHH-2*K*H*SINDL)
UU2=SIN(SIGHH-2*K*H*SINDL)
UU3=COS(SIGHV-2*K*H*SINDL)
UU4=SIN(SIGHV-2*K*H*SINDL)
S1=SIN(K*L*FACK1)
CE1=COS(K*L*FACK1)
S2=SIN(K*L*FACK2)
CE2=COS(K*L*FACK2)
R1=(1.0-CE1)/FACK1
A11=S1/FACK1
R2=(CE1*(1.0-CE2)+S1*S2)/FACK2
A12=(CE1*S2-S1*(1.0-CE2))/FACK2
R3=(1.0-CE1)*COS(2*K*L*SINAC*SINDL)-(1.0-CE1)*SIN(2*K*L*SINAC
1*SINDL)

```

```

F1=(AI3*CE1-R3*S1)/FACK1
F2=(R3*CE1+AI3*S1)/FACK1
F3=(1.0-CE2)/FACK2
F4=S2/FACK2
RB=R1+R2+CABS(RV)*((F2+F3)*UU3-(F1+F4)*UU4)
BI=AI1+AI2-CABS(RV)*((F2+F3)*UU4+(F1+F4)*UU3)
RC=R2-R1+CABS(RV)*((F2-F3)*UU3-(F1-F4)*UU4)
CC=AI2-AI1+CABS(RV)*((F2-F3)*UU4+(F1-F4)*UU3)
RA=R1+R2+CABS(RH)*((F2+F3)*UU1+(F1+F4)*UU2)
AI=AI1+AI2+CABS(RH)*((F2+F3)*UU2+(F1+F4)*UU1)
G(N,E)=0.1*((RB*C0SAC*C0SPI*SINDL+RC*SINAC*C0SDL)**2
1+(BI*C0SAC*C0SPI*SINDL+CC*SINAC*C0SDL)**2+(RA*C0SAC*SINPI)**2
1+(AI*C0SAC*SINPI)**2)
GO T0 42
C VERTICAL HALF RHOMBIC
1800 RIN=1.0
GO T0 2000
END

```

```

SUBROUTINE SINUS(X,SC)
IF(X.GE.10.0)G0 T0 10
DX=X/100
GRAL=0.5
XA=0.0
D0 100 I=2,100
XA=XA+DX
100 GRAL=GRAL+SINC(XA)
GRAL=(GRAL+SINC(X)/2.)*DX
SC=-3.14159265/2.+.GRAL
G0 T0 20
10 SC=-COS(X)/X
20 CONTINUE
RETURN
END

```



```

SUBROUTINE KOSINUS(X,CC)
IF(X.GE.10.0)GO TO 10
DX=X/100
GRAL=0.0
XA=0.0
DO 100 I=2,100
  XA=XA+DX
  100 GRAL=GRAL+(1.0-COS(XA))/XA
  GRAL=(GRAL+(1.0-COS(X))/2*X)*DX
  CC=ALOG(1.781072*X)-GRAL
  GO TO 20
  10 CC=SIN(X)/X
  20 CONTINUE
  RETURN
  END

```

```
FUNCTION CINC(X)  
CINC=COS(X)/X  
RETURN  
END
```



```
FUNCTION SINC(X)  
  SINC=SIN(X)/X  
  RETURN  
END
```

```

FUNCTION ZGRAL(X)
C REQUIRED FOR VERTICAL WHIP WITH GROUND SCREEN
COMPLEX ZGRAL
COMPLEX ARG1, ARG2
REAL L, LMDA, K
COMPLEX ADA, ADAE
COMMON /IMP/ ZO, YO, L, DLPRI, LMDA, NN, WIRE, K, ADA, COSDL
ARG1=CMPLX(0.0, -K*(X**2+L**2)**0.5)
ARG2=CMPLX(0.0, -K*X)
C1=SIN(K*L)**2
IF(C1.LT.0.01)C1=.01
XX=(240.*3.14159265**2*X/(NN*LMDA))*ALOG(X/(NN*WIRE))
ADAE=CMPLX(0.0, XX)
ZGRAL=(ADA*ADAE/(ADA+ADAE))*((CEXP(ARG1)-CEXP(ARG2)*COS(K*L))/
1(2*3.14159265*C1*X))
TEMP1=AIMAG(ADAE)
TEMP2=REAL(ZGRAL)
TEMP3=AIMAG(ZGRAL)
WRITE(6,10)NN,K,WIRE,TEMP1,TEMP2,TEMP3
10 FORMAT('ZGRAL',6F12.6)
RETURN
END

```

```

C
FUNCTION PTGRL(X)
  REQUIRED FOR VERTICAL WHIP WITH GROUND SCREEN
  COMPLEX PTGRL
  COMMON /IMP/ ZO,YO,L,DLPRI,LMDA,NN,WIRE,K,ADA,COSDL
  COMPLEX ADA
  REAL L,LMDA,K
  COMPLEX ARG1,ARG2
  S=X*COSDL
  PI=3.14159265
  IF(S.LE.1)GO TO 20
  P1=1+15/(2*(8*S)**2)-(225.*7*9)/(24*(8*S)**4)+(225.*49*81*143)/
    1(720*(8*S)**6)
  Q1=3/(8*S)-315/(6*(8*S)**3)+(9*35*35*99)/(120*(8*S)**6)
  AJ1=(2./(PI*S))**0.5*(P1*COS(S-3*PI/4)-Q1*SIN(S-3*PI/4))
  GO TO 30
20 CONTINUE
  AJ1=S/2-S**3/16+S**5/384-S**7/(128*144)+S**9/(512*24*120)
30 CONTINUE
  Z=(X**2+K**2*L**2)**0.5
  ARG1=CMPLX(0.0,-Z)
  ARG2=CMPLX(0.0,-X)
  PTGRL=(CEXP(ARG1)-CEXP(ARG2)*COS(K*L))*AJ1
  RETURN
END

```

```

FUNCTION AKEX(X)
C REQUIRED FOR VERTICAL WHIP WITH GROUND SCREEN
COMPLEX AKEX
XX=ABS(X)
CALL K0SINUS(XX,CC)
CALL SINUS(XX,SC)
IF(X.LT.0.0)AKEX=CMPLX(CC,-SC)
IF(X.GE.0.0)AKEX=CMPLX(CC,SC)
RETURN
END

```

```

C REQUIRED FOR DIPPLE
FUNCTION RESIST(S)
REAL L,LMDA,K
COMPLEX ADA
COMMON /IMP/ ZO,YO,L,DLPRI,LMDA,NN,WIRE,K,ADA,COSDL
PI=3.14159265
SZ=S*COS(2*DLPRI)
SY=-S*SIN(2*DLPRI)
TERM=YO+SY
ROW2=(YO+SY)**2
CA=ZO+SZ
CA1=CA+0.5*L/LMDA
CA2=CA-0.5*L/LMDA
R=SQRT(ROW2+CA**2)
R1=SQRT(ROW2+CA1**2)
R2=SQRT(ROW2+CA2**2)
SR=SIN(2*PI*R)/R
SR1=SIN(2*PI*R1)/R1
SR2=SIN(2*PI*R2)/R2
FACR=2*SR*COS(PI*L/LMDA)
RESIST=((SR1*CA1+SR2*CA2-FACR*CA)*SY)/TERM+(FACR-SR1-SR2)*SZ)*
1SIN(2*PI*(0.5*L/LMDA-ABS(S)))/S
RETURN
END

```

```

FUNCTION REACT(S)
C REQUIRED FOR DIPALE
COMPLEX ADA
REAL L,LMDA,K
COMMON /IMP/ ZO,YO,L,DLPRI,LMDA,NN,WIRE,K,ADA,COSDL
PI=3.14159265
SZ=S*COS(2*DLPRI)
SY=-S*SIN(2*DLPRI)
TERM=YO+SY
R0W2=(YO+SY)**2
CA=ZO+SZ
CA1=CA+0.5*L/LMDA
CA2=CA-0.5*L/LMDA
R=SQRT(R0W2+CA**2)
R1=SQRT(R0W2+CA1**2)
R2=SQRT(R0W2+CA2**2)
CR=COS(2*PI*R)/R
CR1=COS(2*PI*R1)/R1
CR2=COS(2*PI*R2)/R2
FACX=2*CR*COS(PI*L/LMDA)
REACT=((CR1*CA1+CA2*CR2-FACX*CA)*SY)/R0W2+(FACX-CR1-CR2)*SZ)*
1SIN(2*PI*(0.5*L/LMDA-ABS(S)))/S
RETURN
END

```

APPENDIX E

SHIPBOARD ANTENNA DYNAMIC SIMULATION EQUATIONS

This appendix presents the development of ship motion equations as functions of sea state and relative direction of the sea. Ship motion is resolved into parameter variation. The values for time varying parameters are used in the compute loop for the dynamic simulation.

The ship-ocean combination is modeled as follows:

1. The ship will roll sinusoidally 8 degrees per sea state if the sea is on the beam, ie. from 090° R or 270° R.
2. The ship will pitch sinusoidally 2.4 degrees per sea state if the sun is on the bow or stern, ie. from 000° R or 180° R. (This represents a small naval combatant ship)
3. Sea state and direction is resolved into ship motion:

Vertical Whip or Sloping Long Wire

$$AA = 2 \cdot L \cdot \sin (\Delta\theta/2)$$

$$BB = 2 \cdot L \cdot \sin (\Delta\theta/2)$$

$$CC = AA^2 + BB^2$$

$$DD = L^2 - (CC/2)^2$$

$$|\Delta_{\theta_3}| = 2 \cdot \tan^{-1} \frac{(CC/2)}{DD}$$

$$\Delta_{\theta_3} = |\Delta_{\theta_3}|$$

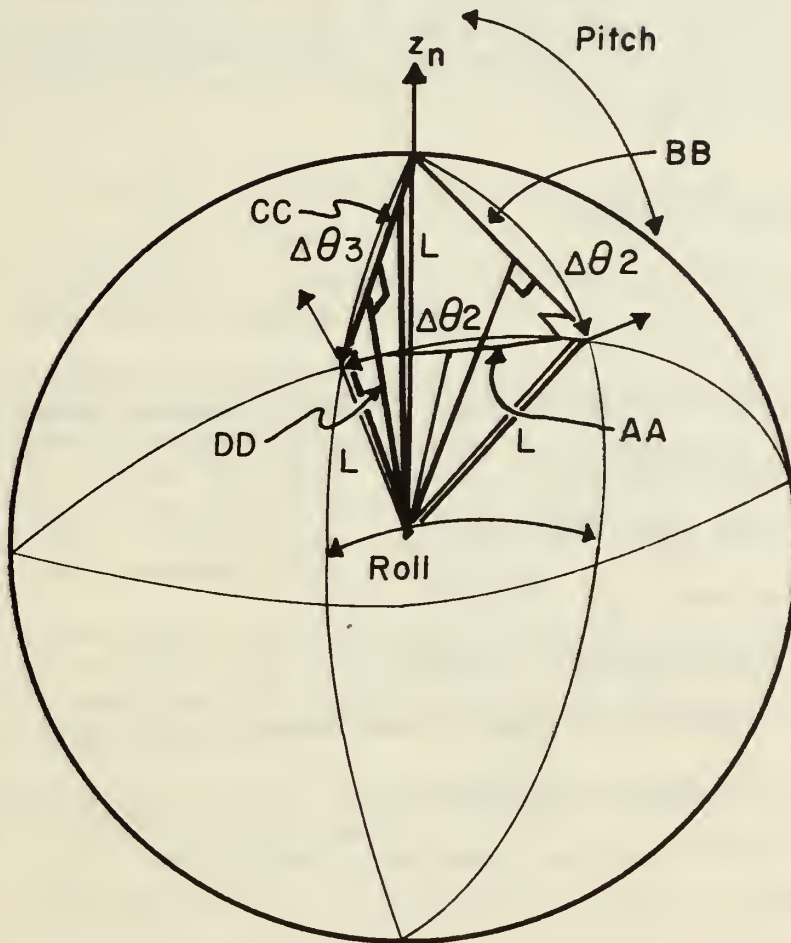
$$\theta'(t) = \theta'(0) - \Delta_{\theta_3} \quad (\text{TILT})$$

$$\sin \Delta_{\theta_3} = \sin \Delta\theta \sin \Delta_{\theta_3}$$

$$\cos \Delta\theta = (1 - \sin^2 \Delta\theta)^{1/2}$$

$$|\Delta\phi| = \tan^{-1} \frac{\sin \Delta\phi}{\cos \Delta\phi} \quad (\text{azimuth change})$$

Pitch $\Delta\theta_2 = 0.3 \cdot \text{wave} \cdot \cos(\text{CRS} \cdot \pi/180)$



Dipole : $\theta'(\tau) = \theta'(0) - \Delta\theta_2$ (filt)

$H(\tau) = H(0) \cdot \cos(\Delta\theta_1) \cdot \cos(\Delta\theta_1)$ (height)

if: $\Delta_{\theta_1} > 0$ and $\Delta_{\theta_2} > 0$ $\Delta \phi = |\Delta \phi|$
 $\Delta_{\theta_1} < 0$ and $\Delta_{\theta_2} > 0$ $\Delta \phi = -|\Delta \phi|$
 $\Delta_{\theta_1} > 0$ and $\Delta_{\theta_2} < 0$ $\Delta \phi = (\pi - |\Delta \phi|)$
 $\Delta_{\theta_1} < 0$ and $\Delta_{\theta_2} < 0$ $\Delta \phi = -(\pi - |\Delta \phi|)$

$$\dot{\phi}(t) = \dot{\phi}(0) - \Delta \phi$$

Definition of terms:

Δ_{θ_1} - ship roll

Δ_{θ_2} - ship pitch

Δ_{θ_3} - tilt of whip or long wire caused by ship motion (whip)

wave - sinusoidal wave

ω - wave radar frequency

$\theta(t)$ - antenna tilt (Dipole)

$h(t)$ - antenna height (Dipole)

$\Delta \phi$ - variation in antenna train caused by ship motion (whip)

$\dot{\phi}(t)$ - antenna train (whip)

CRS - Direction of sea relative to ship's bow

APPENDIX F

OPERATING INSTRUCTIONS FOR U.S. NAVAL POSTGRADUATE SCHOOL GRAPHICS COMPUTER LAB

This appendix gives step by step operating instructions required to use the antenna patterns graphics program at the Naval Postgraduate School. Use of the graphics library program "GATED" and computer light-off procedures are covered in the operators manual and laboratory memoranda and are not included in this appendix.

1. Light-off SDS digital computer in accordance with operating instructions.

2. Light off ADAGE graphics computer in accordance with operating instructions and load library program "Gated".

3. Load the program in the XDS-9300 computer. If an overlayed version of the program is used, the entire program may be loaded. If an overlayed version of the program is not used, computer memory limitations allow loading only two antennas at a time. The input resistance branches, gain branches, and required subroutines for the antennas desired should be loaded along with the main program. A missing labels warning will result but the program may be operated if only antennas loaded are called.

4. When the input light on the teletype is lighted type IDEV = 1* if ADAGE 1 is to be used or IDEV = 2* if ADAGE 2 is to be used. Pushing the carriage return will cause the data input format to be displayed at the graphics terminal.

5. Enter parameters and option commands using "Gated" text editing techniques. Inputs should be as follows:

a. Under ANTEN enter one of the following to specify antenna type:

0001	Tilted Dipole
0002	Vertical Whip
0003	Vertical Whip with Ground Screen
0004	Inverted L
0005	Sloping Longwire
0006	Sloping Vee
0007	Horizontal Rhombic
0008	Vertical Half Rhombic

b. Under LENG enter length in format F4.1

c. Under HGHT enter height in format F4.1

d. Under PHIP enter ϕ in format I4

e. Under THEP enter θ in format F4.0

f. Under FREQ enter f in format F4.0

g. Under EPSL enter ϵ_r in format F4.1

h. Under SGMA enter σ in format F4.2

i. Under PHI enter the observation azimuth angle for the vertical pattern using format I4.

j. Under THET enter the observation zenith angle for the horizontal pattern using format I4.

k. Under PARM enter 0000. If reinitialization is desired to erase a manually entered pattern, enter 0001. If Log Gain patterns are desired, enter 0002.

l. Under ISTH and ISTV enter 0000. If saving the pattern that will be computed in the current compute cycle is desired, enter 0001. If it is desired to keep the pattern in the save array, these option commands must be set to 0000 in the succeeding compute cycle.

m. Under IRCL enter 0000. If displaying saved patterns is desired, enter 0001.

n. Under HGTT enter 0000. Entering 01.0 will multiply the value of sigma by .1. Entering 02.0 will multiply the value of sigma by .01.

o. Under ALPH enter α in format I4.

There are two unused data blocks which no operation edits must be made to finish the data input processor.

6. Axes and a blank graphics data block will now be displayed on the terminal screen. A pattern desired for comparison purposes may be entered in this block using manual graphics editing techniques. To terminate this processor operation, push the end edit button on the function switch panel. This processor may be terminated without entry if desired.

7. The antenna patterns selected will be computed and the horizontal pattern displayed on the upper axis. Pushing the end edit button will cause the vertical pattern to be displayed on the lower axis.

8. If the display saved patterns option has been selected, pushing the end edit button two additional times will cause the vertical and horizontal patterns to be superimposed on the current vertical and horizontal patterns. The program will, then, return to the enter parameters and option commands processor. If recall has not been selected, the program will return to the enter parameters processor from terminating the vertical pattern display processor termination (end edit).

The compute cycle is now repeated. Ending the program must be done in accordance with laboratory operating instructions. Figures 4.1 thru 4.20 are the entries for the examples of section 4.

p. Under ISEA enter sea state in I4, if a dynamic display is desired for dipole, whip or longwire antennas. If dynamic display is not desired, enter 0000.

q. Under ICRS enter relative direction of seas if dynamic display is desired.

LIST OF REFERENCES

1. U. S. Department of Commerce/Environmental Science Service Administration Report ERL 110-ITS 78, Predicting Long Term Operational Parameters of High Frequency Sky Wave Telecommunication Systems, A. F. Barghausen, J. W. Finney, L. L. Proctor, L. D. Scholty, May 1969.
2. Jordan, E. C., and Balmain, K. G. Electromagnetic Waves and Radiating Systems, Prentice Hall, 1968.
3. U. S. Department of Commerce/Environmental Science Service Administration Report ERL 104-ITS 74, Power Gain for Antennas Over Lossy Plane Ground, M. T. Ma, L. C. Walters, April 1969.
4. Baker, H. C., Lagrone A. H., Digital Computation of the Mutual Impedance between Thin Dipoles, Proc. IRE Trans. AP-10, No. 2, P. 172-178.
5. Wait J. R., Pope W. A., Characteristics of a Vertical Antenna with a Radial Conductor Ground System, Appl. Sci. Res B, Vol. 4, P. 177-195.

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<p>An interactive computer graphics antenna gain pattern computation and display program for real-world antenna systems is presented. The use of the program as a teaching tool at the Naval Postgraduate School is discussed. Methods for applying the program for the synthesis and design of complex antenna systems are indicated. Research applications include techniques for rapid inspection of gain equations of newly developed antennas. A ship motion model is developed for studying the effects of electrical geometry variations caused by ship motion in heavy seas on maritime antenna systems and a dynamic presentation of pattern variations is made.</p>			

14 KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
Antennas						
Antenna Gain						
Antenna Patterns						
Computer Program						
Interactive Computer Solution						
Graphics Display						

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